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	DEHDASHT PETROCHEMICAL INDUSTRY COMPANY DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT	
	DOCUMENT TITLE: Mechanical Calculation for Economizer	POI: IFA
Contract No.: DPIC/98-12	DOCUMENT NUMBER: DPIC9812-000-VD-1002-ME-CLN-0030	Rev. No.: D0

DOCUMENT TITLE:

**Mechanical Calculation for Economizer
(E-PK6101-3)**

PURCHASER'S COMMENT/APPROVAL STATUS					Purchaser: NARGAN
1	AP: Approved (Released for Manufacturing)				Requisition No.: DPIC98-12-001-000-ME-MR-4150-0001-D1
2	AN: Approved With Minor Comments (Fabrication may Proceed)				
3	NF: Approved With Comments (Fabrication not Proceed)				Item No. (Tag No.): PK-6101
4	RJ: Rejected				
5	NR: Not be Returned				Vendor Doc. No.: DPIC9812-000-VD-1002-ME-CLN-030-D0
Date:		Signature:			
					
D0	23.Dec.21	A.VOSOUGH	DR.A.NEJATI	DR.A.NEJATI	
REV	DATE ISSUE	PREPARED	CHECKED	APPROVED	



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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT



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Page	Rev-D0	Rev-D1	Rev-D2	Rev-D3	Rev-D4
1	x				
2	x				
3	x				
4	x				
5	x				
6	x				
7	x				
8	x				
9	x				
10	x				
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26	x				
27	x				
28	x				
29	x				
30	x				
31	x				
32	x				
33	x				
34	x				
35	x				

Page	Rev-D0	Rev-D1	Rev-D2	Rev-D3	Rev-D4
36	x				
37	x				
38	x				
39	x				
40	x				
41	x				
42	x				
43	x				
44	x				
45	x				
46	x				
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Page	Rev-D0	Rev-D1	Rev-D2	Rev-D3	Rev-D4
71	x				
72	x				
73	x				
74	x				
75	x				
76	x				
77	x				
78	x				
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96	x				
97	x				
98	x				
99	x				
100	x				
101	x				
102	x				
103	x				
104	x				
105	x				

Page	Rev-D0	Rev-D1	Rev-D2	Rev-D3	Rev-D4
106	x				
107	x				
108	x				
109	x				
110	x				
111	x				
112	x				
113	x				
114	x				
115	x				
116	x				
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135	x				
136	x				
137	x				
138	x				
139	x				
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Page	Rev-D0	Rev-D1	Rev-D2	Rev-D3	Rev-D4
141	x				
142	x				
143	x				
144	x				
145	x				
146	x				
147	x				
148	x				
149	x				
150	x				
151	x				
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164	x				
165	x				
166	x				
167	x				
168	x				
169	x				
170					
171					
172					
173					
174					
175					

Page	Rev-D0	Rev-D1	Rev-D2	Rev-D3	Rev-D4
176					
177					
178					
179					
180					
181					
182					
183					
184					
185					
186					
187					
188					
189					
190					
191					
192					
193					
194					
195					
196					
197					
198					
199					
200					
201					
202					
203					
204					
205					
206					
207					
208					
209					
210					

Table of Contents

Warnings and Errors:	7
Input Echo:	8
XY Coordinate Calculations:	18
Flg Calc [Int P]: FLANGE	19
Flg Calc [Int P]: New Flange	24
Internal Pressure Calculations:	29
External Pressure Calculations:	35
Element and Detail Weights:	39
Nozzle Flange MAWP:	42
Wind Load Calculation:	43
Earthquake Load Calculation:	46
Center of Gravity Calculation:	48
Horizontal Vessel Analysis (Ope.):	49
Horizontal Vessel Analysis (Test):	66
Nozzle Summary:	80
Nozzle Calcs.: T4	81
Nozzle Calcs.: T1	86
Nozzle Calcs.: S2	97
Nozzle Calcs.: S1	108
Nozzle Calcs.: S3	119
Nozzle Calcs.: T2	130
Nozzle Calcs.: T3	141
Nozzle Schedule:	146
ASME TS Calc:	148
MDMT Summary:	164
Vessel Design Summary:	166

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
Tag no:E-PK6101-3 ECONOMIZER

DESIGN CALCULATION

In Accordance with ASME Section VIII Division 1

ASME Code Version : 2017

Analysis Performed by : SPLM Licensed User

Job File :

Date of Analysis : Dec 22,2021 11:46pm

PV Elite 2019 SP1, March 2019

Note:

PV Elite performs all calculations internally in Imperial Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulas. The finalized results are reflected to show the user's set of selected units.

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Tag no:E-PK6101-3 ECONOMIZER
PV Elite 2019 SP1 Licensee: SPLM Licensed User
FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
Warnings and Errors: Step: 0 11:46pm Dec 22,2021

Class From To : Basic Element Checks.
=====

Class From To: Check of Additional Element Data
=====

There were no geometry errors or warnings.

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Input Echo: Step: 1 11:46pm Dec 22,2021

PV Elite Vessel Analysis Program: Input Data

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 Tag no:E-PK6101-3 ECONOMIZER

Exchanger Design Pressures and Temperatures

Shell Side Design Pressure	23	bars
Channel Side Design Pressure	23	bars
Shell Side Design Temperature	120.0	°C
Channel Side Design Temperature	120.0	°C
Radiography, Shell Side	RT-1	
Radiography, Channel Side	RT-1	
Service Type, Shell Side	None	
Service Type, Channel Side	None	
MDMT (CET), Shell Side	-45.0	°C
MDMT (CET), Tube Side	-45.0	°C
User defined MAWP, Shell Side	0	bars
User defined MAWP, Channel Side	0	bars
User defined MAPnc, Shell Side	0	bars
User defined MAPnc, Channel Side	0	bars
User defined Test Pres., Shell Side	0	bars
User defined Test Pres., Channel Side	0	bars

Type of Hydrotest	UG-99(b) Note [36]	
Hydrotest Position	Horizontal	
Projection of Nozzle from Vessel Top	0	mm.
Projection of Nozzle from Vessel Bottom	0	mm.
Type of Construction	Welded	
Use Higher Longitudinal Stresses (Flag)	Y	
Select t for Internal Pressure (Flag)	N	
Select t for External Pressure (Flag)	N	
Select t for Axial Stress (Flag)	N	
Select Location for Stiff. Rings (Flag)	N	
Consider Vortex Shedding	N	
Perform a Corroded Hydrotest	Y	

Load Case 1	NP+EW+WI+FW+BW
Load Case 2	NP+EW+EE+FS+BS
Load Case 3	NP+OW+WI+FW+BW
Load Case 4	NP+OW+EQ+FS+BS
Load Case 5	NP+HW+HI
Load Case 6	NP+HW+HE
Load Case 7	IP+OW+WI+FW+BW
Load Case 8	IP+OW+EQ+FS+BS
Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS
Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+EW
Load Case 17	NP+VO+OW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

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Input Echo: Step: 1 11:46pm Dec 22,2021

Wind Design Code		ASCE-7 2010	
Wind Load Reduction Scale Factor		0.600	
Basic Wind Speed	[V]	195	Km/hr
Surface Roughness Category		C: Open Terrain	
Importance Factor		1.0	
Type of Surface		Moderately Smooth	
Base Elevation		123000	mm.
Percent Wind for Hydrotest		33.0	
Using User defined Wind Press. Vs Elev.		N	
Height of Hill or Escarpment	H or Hh	0	mm.
Distance Upwind of Crest	Lh	0	mm.
Distance from Crest to the Vessel	x	0	mm.
Type of Terrain (Hill, Escarpment)		Flat	
Damping Factor (Beta) for Wind (Ope)		0.0100	
Damping Factor (Beta) for Wind (Empty)		0.0000	
Damping Factor (Beta) for Wind (Filled)		0.0000	

Seismic Design Code		ASCE 7-2010	
Seismic Load Reduction Scale Factor		0.700	
Importance Factor		1.500	
Table Value Fa		1.000	
Table Value Fv		1.300	
Short Period Acceleration value Ss		1.163	
Long Period Acceleration Value Sl		0.600	
Moment Reduction Factor Tau		1.000	
Force Modification Factor R		2.000	
Site Class		C	
Component Elevation Ratio	z/h	0.000	
Amplification Factor	Ap	0.000	
Force Factor		0.000	
Consider Vertical Acceleration		No	
Minimum Acceleration Multiplier		0.000	
User Value of Sds (used if > 0)		0.000	
User Value of Sd1 (used if > 0)		0.000	

Design Pressure + Static Head		Y
Consider MAP New and Cold in Noz. Design		N
Consider External Loads for Nozzle Des.		Y
Use ASME VIII-1 Appendix 1-9		N

Material Database Year	Current w/Addenda or Code Year
------------------------	--------------------------------

Configuration Directives:

Do not use Nozzle MDMT Interpretation VIII-1 01-37	No
Use Table G instead of exact equation for "A"	Yes
Shell Head Joints are Tapered	Yes
Compute "K" in corroded condition	Yes
Use Code Case 2286	No
Use the MAWP to compute the MDMT	Yes
For thickness ratios <= 0.35, MDMT will be -155F (-104C)	Yes
For PWHT & P1 Materials the MDMT can be < -55F (-48C)	No
Using Metric Material Databases, ASME II D	No
Calculate B31.3 type stress for Nozzles with Loads	Yes
Reduce the MDMT due to lower membrane stress	Yes
Consider Longitudinal Stress in MDMT calcs. (Div. 1)	No

Complete Listing of Vessel Elements and Details:

Element From Node	10
-------------------	----

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Input Echo: Step: 1 11:46pm Dec 22,2021

Element To Node	20
Element Type	Elliptical
Description	HEAD 1
Distance "FROM" to "TO"	50 mm.
Inside Diameter	581 mm.
Element Thickness	10 mm.
Internal Corrosion Allowance	3 mm.
Nominal Thickness	12 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	23 bars
Design Temperature Internal Pressure	120 °C
Design External Pressure	1.1 bars
Design Temperature External Pressure	120 °C
Effective Diameter Multiplier	1.2
Material Name	SA-516 70 [Normalized]
Allowable Stress, Ambient	137.9 N./mm ²
Allowable Stress, Operating	137.9 N./mm ²
Allowable Stress, Hydrotest	235.8 N./mm ²
Material Density	0.00775 kg./cm ³
P Number Thickness	30.988 mm.
Yield Stress, Operating	236 N./mm ²
UCS-66 Chart Curve Designation	D
External Pressure Chart Name	CS-2
UNS Number	K02700
Product Form	Plate
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Elliptical Head Factor	2.0
Weld is pre-Heated	No

Element From Node	10
Detail Type	Liquid
Detail ID	1
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Liquid	581 mm.
Liquid Density	0.0005998 kg./cm ³
Element From Node	10
Detail Type	Insulation
Detail ID	Ins: 10
Dist. from "FROM" Node / Offset dist	-145.25 mm.
Height/Length of Insulation	195.25 mm.
Thickness of Insulation	80 mm.
Density	0.00012 kg./cm ³

Element From Node	20
Element To Node	30
Element Type	Cylinder
Description	CHANNEL 01
Distance "FROM" to "TO"	407 mm.
Inside Diameter	581 mm.
Element Thickness	10 mm.
Internal Corrosion Allowance	3 mm.
Nominal Thickness	10 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	23 bars
Design Temperature Internal Pressure	120 °C
Design External Pressure	1.1 bars
Design Temperature External Pressure	120 °C

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Input Echo: Step: 1 11:46pm Dec 22,2021

Effective Diameter Multiplier	1.2
Material Name	SA-516 70 [Normalized]
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No
Element From Node	20
Detail Type	Liquid
Detail ID	2
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Liquid	581 mm.
Liquid Density	0.0005998 kg./cm ³
Element From Node	20
Detail Type	Insulation
Detail ID	Ins: 20
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Insulation	407 mm.
Thickness of Insulation	80 mm.
Density	0.00012 kg./cm ³
Element From Node	20
Detail Type	Nozzle
Detail ID	T4
Dist. from "FROM" Node / Offset dist	203 mm.
Nozzle Diameter	0.75 in.
Nozzle Schedule	None
Nozzle Class	300
Layout Angle	90.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.05845 kN
Grade of Attached Flange	GR 1.1
Nozzle Matl	SA-350 LF2 [Impact Tested]
Element From Node	20
Detail Type	Nozzle
Detail ID	T1
Dist. from "FROM" Node / Offset dist	203 mm.
Nozzle Diameter	4 in.
Nozzle Schedule	120
Nozzle Class	300
Layout Angle	270.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.2988 kN
Grade of Attached Flange	GR 1.1
Nozzle Matl	SA-333 6 [Impact Tested]

Element From Node	30
Element To Node	40
Element Type	Flange
Description	BODY FLANGE 01
Distance "FROM" to "TO"	88 mm.
Flange Inside Diameter	581 mm.
Element Thickness	62 mm.
Internal Corrosion Allowance	3 mm.
Nominal Thickness	79 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	23 bars
Design Temperature Internal Pressure	120 °C

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Design External Pressure	1.1	bars
Design Temperature External Pressure	120	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-350 LF2	[Impact Tested]
Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	137.9	N./mm ²
Allowable Stress, Hydrotest	223.4	N./mm ²
Material Density	0.00775	kg./cm ³
P Number Thickness	31.75	mm.
Yield Stress, Operating	223.57	N./mm ²
UCS-66 Chart Curve Designation	Impact Tested	
External Pressure Chart Name	CS-2	
UNS Number	K03011	
Class / Thickness / Grade	1::	
Product Form	Forgings	
Perform Flange Stress Calculation (Y/N)	Y	
Weight of ANSI B16.5/B16.47 Flange	0	kN
Class of ANSI B16.5/B16.47 Flange		
Grade of ANSI B16.5/B16.47 Flange		
Weld is pre-Heated	No	
Element From Node	30	
Detail Type	Liquid	
Detail ID	3	
Dist. from "FROM" Node / Offset dist	0	mm.
Height/Length of Liquid	540	mm.
Liquid Density	0.0006998	kg./cm ³
Element From Node	30	
Detail Type	Insulation	
Detail ID	Ins: 30	
Dist. from "FROM" Node / Offset dist	0	mm.
Height/Length of Insulation	88	mm.
Thickness of Insulation	80	mm.
Density	0.00012	kg./cm ³

Element From Node	40	
Element To Node	50	
Element Type	Cylinder	
Description	SHELL	
Distance "FROM" to "TO"	5880	mm.
Inside Diameter	581	mm.
Element Thickness	10	mm.
Internal Corrosion Allowance	3	mm.
Nominal Thickness	10	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	23	bars
Design Temperature Internal Pressure	120	°C
Design External Pressure	1.1	bars
Design Temperature External Pressure	120	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-516 70	[Normalized]
Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	137.9	N./mm ²
Allowable Stress, Hydrotest	235.8	N./mm ²
Material Density	0.00775	kg./cm ³
P Number Thickness	30.988	mm.
Yield Stress, Operating	236	N./mm ²
UCS-66 Chart Curve Designation	D	

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External Pressure Chart Name	CS-2
UNS Number	K02700
Product Form	Plate
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No
Element From Node	40
Detail Type	Saddle
Detail ID	Fixed Saddle
Dist. from "FROM" Node / Offset dist	940 mm.
Width of Saddle	150 mm.
Height of Saddle at Bottom	600 mm.
Saddle Contact Angle	120.0
Height of Composite Ring Stiffener	0 mm.
Width of Wear Plate	225 mm.
Thickness of Wear Plate	10 mm.
Contact Angle, Wear Plate (degrees)	132.0
Friction coefficient	0.0
Moment Factor	3.0
Dimension E at base (optional)	0 mm.
Circumferential Eff. over Saddle	1.0
Circumferential Eff. at Midspan	1.0
Tangent to Tangent dist. (optional)	0 mm.
Element From Node	40
Detail Type	Saddle
Detail ID	Sliding Saddle
Dist. from "FROM" Node / Offset dist	4940 mm.
Width of Saddle	150 mm.
Height of Saddle at Bottom	600 mm.
Saddle Contact Angle	120.0
Height of Composite Ring Stiffener	0 mm.
Width of Wear Plate	225 mm.
Thickness of Wear Plate	10 mm.
Contact Angle, Wear Plate (degrees)	132.0
Friction coefficient	0.40000001
Moment Factor	3.0
Dimension E at base (optional)	0 mm.
Circumferential Eff. over Saddle	1.0
Circumferential Eff. at Midspan	1.0
Tangent to Tangent dist. (optional)	0 mm.
Element From Node	40
Detail Type	Liquid
Detail ID	4
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Liquid	581 mm.
Liquid Density	0.0005998 kg./cm ³
Element From Node	40
Detail Type	Insulation
Detail ID	Ins: 40
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Insulation	5880 mm.
Thickness of Insulation	80 mm.
Density	0.00012 kg./cm ³
Element From Node	40
Detail Type	Nozzle
Detail ID	S2

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Dist. from "FROM" Node / Offset dist	196	mm.
Nozzle Diameter	6	in.
Nozzle Schedule	80	
Nozzle Class	300	
Layout Angle	90.0	
Blind Flange (Y/N)	N	
Weight of Nozzle (Used if > 0)	0.5568	kN
Grade of Attached Flange	GR 1.1	
Nozzle Matl	SA-333 6	[Impact Tested]

Element From Node	40	
Detail Type	Nozzle	
Detail ID	S1	
Dist. from "FROM" Node / Offset dist	5680	mm.
Nozzle Diameter	6	in.
Nozzle Schedule	80	
Nozzle Class	300	
Layout Angle	90.0	
Blind Flange (Y/N)	N	
Weight of Nozzle (Used if > 0)	0.5568	kN
Grade of Attached Flange	GR 1.1	
Nozzle Matl	SA-333 6	[Impact Tested]

Element From Node	40	
Detail Type	Nozzle	
Detail ID	S3	
Dist. from "FROM" Node / Offset dist	146	mm.
Nozzle Diameter	2	in.
Nozzle Schedule	160	
Nozzle Class	300	
Layout Angle	270.0	
Blind Flange (Y/N)	N	
Weight of Nozzle (Used if > 0)	0.1184	kN
Grade of Attached Flange	GR 1.1	
Nozzle Matl	SA-333 6	[Impact Tested]

Element From Node	40	
Detail Type	Weight	
Detail ID	WEIGHT BAFFLE	
Dist. from "FROM" Node / Offset dist	2940	mm.
Miscellaneous Weight	1.4709	kN
Offset from Element Centerline	0	mm.

Element From Node	50	
Element To Node	60	
Element Type	Flange	
Description	BODY FLANGE 002	
Distance "FROM" to "TO"	88	mm.
Flange Inside Diameter	581	mm.
Element Thickness	62	mm.
Internal Corrosion Allowance	3	mm.
Nominal Thickness	79	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	23	bars
Design Temperature Internal Pressure	120	°C
Design External Pressure	1.1	bars
Design Temperature External Pressure	120	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-350 LF2	[Impact Tested]

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Input Echo: Step: 1 11:46pm Dec 22,2021

Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	137.9	N./mm ²
Allowable Stress, Hydrotest	223.4	N./mm ²
Material Density	0.00775	kg./cm ³
P Number Thickness	31.75	mm.
Yield Stress, Operating	223.57	N./mm ²
UCS-66 Chart Curve Designation	Impact Tested	
External Pressure Chart Name	CS-2	
UNS Number	K03011	
Class / Thickness / Grade	1::	
Product Form	Forgings	
Perform Flange Stress Calculation (Y/N)	Y	
Weight of ANSI B16.5/B16.47 Flange	0	kN
Class of ANSI B16.5/B16.47 Flange		
Grade of ANSI B16.5/B16.47 Flange		
Weld is pre-Heated	No	

Element From Node	50	
Detail Type	Liquid	
Detail ID	5	
Dist. from "FROM" Node / Offset dist	0	mm.
Height/Length of Liquid	581	mm.
Liquid Density	0.0005998	kg./cm ³

Element From Node	50	
Detail Type	Insulation	
Detail ID	Ins: 50	
Dist. from "FROM" Node / Offset dist	0	mm.
Height/Length of Insulation	88	mm.
Thickness of Insulation	80	mm.
Density	0.00012	kg./cm ³

Element From Node	60	
Element To Node	70	
Element Type	Cylinder	
Description	CHANNEL 002	
Distance "FROM" to "TO"	407	mm.
Inside Diameter	581	mm.
Element Thickness	10	mm.
Internal Corrosion Allowance	3	mm.
Nominal Thickness	10	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	23	bars
Design Temperature Internal Pressure	120	°C
Design External Pressure	1.1	bars
Design Temperature External Pressure	120	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-516 70	[Normalized]
Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	137.9	N./mm ²
Allowable Stress, Hydrotest	235.8	N./mm ²
Material Density	0.00775	kg./cm ³
P Number Thickness	30.988	mm.
Yield Stress, Operating	236	N./mm ²
UCS-66 Chart Curve Designation	D	
External Pressure Chart Name	CS-2	
UNS Number	K02700	
Product Form	Plate	
Efficiency, Longitudinal Seam	1.0	

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Input Echo: Step: 1 11:46pm Dec 22,2021

Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No
Element From Node	60
Detail Type	Liquid
Detail ID	5
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Liquid	581 mm.
Liquid Density	0.0005998 kg./cm ³
Element From Node	60
Detail Type	Insulation
Detail ID	Ins: 60
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Insulation	407 mm.
Thickness of Insulation	80 mm.
Density	0.00012 kg./cm ³
Element From Node	60
Detail Type	Nozzle
Detail ID	T2
Dist. from "FROM" Node / Offset dist	200 mm.
Nozzle Diameter	6 in.
Nozzle Schedule	80
Nozzle Class	300
Layout Angle	90.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.5568 kN
Grade of Attached Flange	GR 1.1
Nozzle Matl	SA-333 6 [Impact Tested]
Element From Node	60
Detail Type	Nozzle
Detail ID	T3
Dist. from "FROM" Node / Offset dist	203 mm.
Nozzle Diameter	1 in.
Nozzle Schedule	None
Nozzle Class	300
Layout Angle	270.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.05845 kN
Grade of Attached Flange	GR 1.1
Nozzle Matl	SA-350 LF2 [Impact Tested]

Element From Node	70
Element To Node	80
Element Type	Elliptical
Description	HEAD 002
Distance "FROM" to "TO"	50 mm.
Inside Diameter	581 mm.
Element Thickness	10 mm.
Internal Corrosion Allowance	3 mm.
Nominal Thickness	12 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	23 bars
Design Temperature Internal Pressure	120 °C
Design External Pressure	1.1 bars
Design Temperature External Pressure	120 °C
Effective Diameter Multiplier	1.2

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Input Echo: Step: 1 11:46pm Dec 22,2021

Material Name	SA-516 70 [Normalized]
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Elliptical Head Factor	2.0
Weld is pre-Heated	No
Element From Node	70
Detail Type	Liquid
Detail ID	6
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Liquid	581 mm.
Liquid Density	0.0005998 kg./cm ³
Element From Node	70
Detail Type	Insulation
Detail ID	Ins: 70
Dist. from "FROM" Node / Offset dist	0 mm.
Height/Length of Insulation	195.25 mm.
Thickness of Insulation	80 mm.
Density	0.00012 kg./cm ³

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 XY Coordinate Calculations: Step: 2 11:46pm Dec 22,2021

XY Coordinate Calculations:

From	To	X (Horiz.) mm.	Y (Vert.) mm.	DX (Horiz.) mm.	DY (Vert.) mm.
HEAD 1		50	...	50	...
CHANNEL 01		457	...	407	...
BODY FLANGE 01		545	...	88	...
SHELL		6493.18	...	5880	...
BODY FLANGE 002		6587.35	...	88	...
CHANNEL 002		7056.35	...	407	...
HEAD 002		7106.35	...	50	...

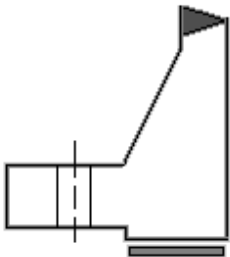
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Flg Calc [Int P]: FLANGE Flng: 3 11:46pm Dec 22,2021

Flange Input Data Values Description: FLANGE :

BODY FLANGE 01

Description of Flange Geometry (Type)		Integral Weld Neck	
Design Pressure	P	23.04	bars
Design Temperature		120	°C
Internal Corrosion Allowance	ci	3.0000	mm.
External Corrosion Allowance	ce	0.0000	mm.
Use Corrosion Allowance in Thickness Calcs.		Yes	
Flange Inside Diameter	B	581.000	mm.
Flange Outside Diameter	A	715.000	mm.
Flange Thickness	t	62.0000	mm.
Thickness of Hub at Small End	go	10.0000	mm.
Thickness of Hub at Large End	gl	17.0000	mm.
Length of Hub	h	26.0000	mm.
Flange Material		SA-350 LF2	
Flange Material UNS number		K03011	
Flange Allowable Stress At Temperature	Sfo	137.90	N./mm ²
Flange Allowable Stress At Ambient	Sfa	137.90	N./mm ²
Bolt Material		SA-320 L7	
Bolt Allowable Stress At Temperature	Sb	172.38	N./mm ²
Bolt Allowable Stress At Ambient	Sa	172.38	N./mm ²
Diameter of Bolt Circle	C	673.000	mm.
Nominal Bolt Diameter	a	19.0500	mm.
Type of Threads		UNC Thread Series	
Number of Bolts		32	
Flange Face Outside Diameter	Fod	644.000	mm.
Flange Face Inside Diameter	Fid	581.000	mm.
Flange Facing Sketch		1, Code Sketch 1a	
Gasket Outside Diameter	Go	641.000	mm.
Gasket Inside Diameter	Gi	611.000	mm.
Gasket Factor	m	3.7800	
Gasket Design Seating Stress	y	62.05	N./mm ²
Column for Gasket Seating		2, Code Column II	
Gasket Thickness	tg	3.0000	mm.
Length of Partition Gasket	lp	1078.0000	mm.
Width of Partition Gasket	tp	6.0000	mm.
Partition Gasket Factor	mPart	3.7500	
Partition Gasket Design Seating Stress	yPart	62.05	N./mm ²



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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Flg Calc [Int P]: FLANGE Flng: 3 11:46pm Dec 22,2021

ASME Code, Section VIII Division 1, 2017

Hub Small End Required Thickness due to Internal Pressure:

$$= (P*(D/2+Ca))/(S*E-0.6*P) \text{ per UG-27 (c)(1)}$$

$$= (23.04*(581.0/2+3.0))/(137.9*1.0-0.6*23.04)+Ca$$

$$= 7.9530 \text{ mm.}$$

Hub Small End Hub MAWP:

$$= (S*E*t)/(R+0.6*t) \text{ per UG-27 (c)(1)}$$

$$= (137.9 * 1.0 * 7.0)/(293.5 + 0.6 * 7.0)$$

$$= 32.423 \text{ bars}$$

Corroded Flange Thickness, $t_c = T - c_i$	59.000	mm.
Corroded Flange ID, $B_{cor} = B + 2 * F_{cor}$	587.000	mm.
Corroded Large Hub, $g_{lCor} = g_l - c_i$	14.000	mm.
Corroded Small Hub, $g_{oCor} = g_o - c_i$	7.000	mm.
Code R Dimension, $R = ((C - B_{cor})/2) - g_{lCor}$	29.000	mm.
Gasket Contact Width, $N = (G_o - G_i) / 2$	15.000	mm.
Basic Gasket Width, $b_o = N / 2$	7.500	mm.
Effective Gasket Width, $b = C_b \text{ sqrt}(b_o)$	6.901	mm.
Gasket Reaction Diameter, $G = G_o - 2 * b$	627.198	mm.

Basic Flange and Bolt Loads:

Hydrostatic End Load due to Pressure [H]:

$$= 0.785 * G^2 * P_{eq}$$

$$= 0.785 * 627.1979^2 * 23.037$$

$$= 711.728 \text{ kN}$$

Contact Load on Gasket Surfaces [Hp]:

$$= 2 * b * P_i * G * m * P + 2 * l_p * b_{Part} * m_{Part} * P$$

$$= 2 * 6.9011 * 3.1416 * 627.1979 * 3.78 * 23.04$$

$$+ 2.0 * 1078.0 * 3.0 * 3.75 * 23.0371$$

$$= 292.690 \text{ kN}$$

Hydrostatic End Load at Flange ID [Hd]:

$$= P_i * B_{cor}^2 * P / 4$$

$$= 3.1416 * 587.0^2 * 23.0371 / 4$$

$$= 623.421 \text{ kN}$$

Pressure Force on Flange Face [Ht]:

$$= H - H_d$$

$$= 712 - 623$$

$$= 88.307 \text{ kN}$$

Operating Bolt Load [Wm1]:

$$= \max(H + H_p + H'p, 0)$$

$$= \max(712 + 293 + 0, 0)$$

$$= 1004.418 \text{ kN}$$

Gasket Seating Bolt Load [Wm2]:

$$= y * b * P_i * G + y_{Part} * b_{Part} * l_p$$

$$= 62.05 * 6.9011 * 3.141 * 627.198 + 62.05 * 3.0 * 1078.0$$

$$= 1044.378 \text{ kN}$$

Required Bolt Area [Am]:

$$= \text{Maximum of } W_{m1}/S_b, W_{m2}/S_a$$

$$= \text{Maximum of } 1004/172, 1044/172$$

$$= 60.593 \text{ cm}^2$$

ASME Maximum Circumferential Spacing between Bolts per App. 2 eq. (3) [Bsmax]:

$$= 2a + 6t/(m + 0.5)$$

$$= 2 * 19.05 + 6 * 59.0/(3.78 + 0.5)$$

$$= 120.810 \text{ mm.}$$

Actual Circumferential Bolt Spacing [Bs]:

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Flg Calc [Int P]: FLANGE Flng: 3 11:46pm Dec 22,2021

$$= C * \sin(\pi / n)$$

$$= 673.0 * \sin(3.142/32)$$

$$= 65.966 \text{ mm.}$$

ASME Moment Multiplier for Bolt Spacing per App. 2 eq. (7) [Bsc]:

$$= \max(\text{sqrt}(Bs / (2a + t)), 1)$$

$$= \max(\text{sqrt}(65.966 / (2 * 19.05 + 59.0)), 1)$$

$$= 1.0000$$

Bolting Information for UNC Thread Series (Non Mandatory):

	Minimum	Actual	Maximum
Bolt Area, cm ²	60.593	62.348	
Radial Distance between Hub and Bolts:	28.575	29.000	
Radial Distance between Bolts and Edge:	20.637	21.000	
Circ. Spacing between the Bolts:	44.450	65.966	120.810

Min. Gasket Contact Width (Brownell Young) [Not an ASME Calc] [Nmin]:

$$= Ab * Sa / (y * \pi * (Go + Gi))$$

$$= 62.348 * 172.38 / (62.05 * 3.14 * (641.0 + 611.0))$$

$$= 4.403 \text{ mm.}$$

Flange Design Bolt Load, Gasket Seating [W]:

$$= Sa * (Am + Ab) / 2$$

$$= 172.38 * (60.5927 + 62.3483) / 2$$

$$= 1059.51 \text{ kN}$$

Gasket Load for the Operating Condition [HG]:

$$= Wm1 - H$$

$$= 1004 - 712$$

$$= 292.69 \text{ kN}$$

Moment Arm Calculations:

Distance to Gasket Load Reaction [hg]:

$$= (C - G) / 2$$

$$= (673.0 - 627.1979) / 2$$

$$= 22.9011 \text{ mm.}$$

Distance to Face Pressure Reaction [ht]:

$$= (R + g1 + hg) / 2$$

$$= (29.0 + 14.0 + 22.9011) / 2$$

$$= 32.9505 \text{ mm.}$$

Distance to End Pressure Reaction [hd]:

$$= R + (g1 / 2)$$

$$= 29.0 + (14.0 / 2.0)$$

$$= 36.0000 \text{ mm.}$$

Summary of Moments for Internal Pressure: (N-m)

Loading	Force	Distance	Bolt Corr	Moment
End Pressure, Md	623.	36.0000	1.0000	22452.
Face Pressure, Mt	88.	32.9505	1.0000	2911.
Gasket Load, Mg	293.	22.9011	1.0000	6706.
Gasket Seating, Matm	1060.	22.9011	1.0000	24274.
Total Moment for Operation, Mop				32069. N-m
Total Moment for Gasket seating, Matm				24274. N-m
Effective Hub Length, ho = sqrt(Bcor*goCor)			64.101 mm.	
Hub Ratio, h/h0 = HL / H0			0.406	

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Flg Calc [Int P]: FLANGE Flng: 3 11:46pm Dec 22,2021

Thickness Ratio, $g1/g0 = (g1Cor/goCor)$ 2.000

Flange Factors for Integral Flange:

Factor F 0.847
 Factor V 0.257
 Factor f 1.586
 Factors from Figure 2-7.1 K = 1.218
 T = 1.832 U = 10.940
 Y = 9.955 Z = 5.135
 d = 0.13358E+06 mm.³ e = 0.0132 mm.⁻¹
 Stress Factors ALPHA = 1.779
 BETA = 2.039 GAMMA = 0.972
 DELTA = 1.537 Lamda = 2.509

Longitudinal Hub Stress, Operating [SHo]:

$= (f * Mop / Bcor) / (L * g1^2)$
 $= (1.5858 * 32069 / 587.0) / (2.5089 * 14.0^2)$
 $= 176.12 \text{ N./mm}^2$

Longitudinal Hub Stress, Seating [SHa]:

$= (f * Matm / Bcor) / (L * g1^2)$
 $= (1.5858 * 24274 / 587.0) / (2.5089 * 14.0^2)$
 $= 133.31 \text{ N./mm}^2$

Radial Flange Stress, Operating [SRo]:

$= (Beta * Mop / Bcor) / (L * t^2)$
 $= (2.0393 * 32069 / 587.0) / (2.5089 * 59.0^2)$
 $= 12.75 \text{ N./mm}^2$

Radial Flange Stress, Seating [SRa]:

$= (Beta * Matm / Bcor) / (L * t^2)$
 $= (2.0393 * 24274 / 587.0) / (2.5089 * 59.0^2)$
 $= 9.65 \text{ N./mm}^2$

Tangential Flange Stress, Operating [STo]:

$= (Y * Mo / (t^2 * Bcor)) - Z * SRO$
 $= (9.9551 * 32069 / (59.0^2 * 587.0)) - 5.1351 * 13$
 $= 90.71 \text{ N./mm}^2$

Tangential Flange Stress, Seating [STa]:

$= (y * Matm / (t^2 * Bcor)) - Z * SRA$
 $= (9.9551 * 24274 / (59.0^2 * 587.0)) - 5.1351 * 10$
 $= 68.66 \text{ N./mm}^2$

Average Flange Stress, Operating [SAo]:

$= (SHo + \max(SRO, STo)) / 2$
 $= (176 + \max(13, 91)) / 2$
 $= 133.41 \text{ N./mm}^2$

Average Flange Stress, Seating [SAa]:

$= (SHa + \max(SRA, STa)) / 2$
 $= (133 + \max(10, 69)) / 2$
 $= 100.98 \text{ N./mm}^2$

Bolt Stress, Operating [BSo]:

$= Wm1 / Ab$
 $= 1004 / 62.3483$
 $= 161.11 \text{ N./mm}^2$

Bolt Stress, Seating [BSa]:

$= (Wm2 / Ab)$
 $= (1044 / 62.3483)$
 $= 167.52 \text{ N./mm}^2$

Flange Stress Analysis Results: N./mm²

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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Flg Calc [Int P]: FLANGE Flng: 3 11:46pm Dec 22,2021

	Actual	Operating Allowed	Gasket Actual	Seating Allowed
Longitudinal Hub	176.	207.	133.	207.
Radial Flange	13.	138.	10.	138.
Tangential Flange	91.	138.	69.	138.
Maximum Average	133.	138.	101.	138.
Bolting	161.	172.	168.	172.

Minimum Required Flange Thickness 60.833 mm.
 Estimated M.A.W.P. (Operating) 23.812 bars
 Estimated Finished Weight of Flange at given Thk. 70.6 kg.
 Estimated Unfinished Weight of Forging at given Thk 93.0 kg.

Flange Rigidity Based on Required Thickness [ASME]:

Flange Rigidity Index, Seating (rotation check) per APP. 2 [Js]:

$$= 52.14 * Ma / Bsc * Cnv_fac * V / (Lambda * Eamb * go^2 * ho * Ki)$$

$$= 52.14 * 24273.7/1.0 * 999.68 * 0.257/(2.409 * 202713 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.707 \text{ (should be } \leq 1)$$

Flange Rigidity Index Operating (rotation check) per APP. 2 [J]:

$$= 52.14 * Mo / Bsc * Cnv_fac * V / (Lambda * Eop * goc^2 * ho * Ki)$$

$$= 52.14 * 32068.8/1.0 * 999.68 * 0.257/(2.409 * 196922 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.962 \text{ (should be } \leq 1)$$

Flange Rigidity Based on Given Thickness [ASME]:

Flange Rigidity Index, Seating (rotation check) per APP. 2 [Js]:

$$= 52.14 * Ma / Bsc * Cnv_fac * V / (Lambda * Eamb * go^2 * ho * Ki)$$

$$= 52.14 * 24273.7/1.0 * 999.68 * 0.257/(2.509 * 202713 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.679 \text{ (should be } \leq 1)$$

Flange Rigidity Index Operating (rotation check) per APP. 2 [J]:

$$= 52.14 * Mo / Bsc * Cnv_fac * V / (Lambda * Eop * goc^2 * ho * Ki)$$

$$= 52.14 * 32068.8/1.0 * 999.68 * 0.257/(2.509 * 196922 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.924 \text{ (should be } \leq 1)$$

Minimum Design Metal Temperature Results:

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification -46 °C

Note: UCS-66(b)-(c) was considered in the flange MDMT calculation.

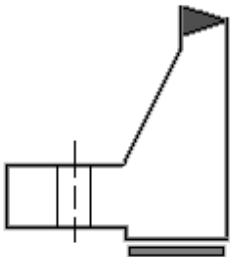
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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Flg Calc [Int P]: New Flange Flng: 4 11:46pm Dec 22,2021

Flange Input Data Values Description: New Flange :

BODY FLANGE 002

Description of Flange Geometry (Type)		Integral Weld Neck	
Design Pressure	P	23.03	bars
Design Temperature		120	°C
Internal Corrosion Allowance	ci	3.0000	mm.
External Corrosion Allowance	ce	0.0000	mm.
Use Corrosion Allowance in Thickness Calcs.		Yes	
Flange Inside Diameter	B	581.000	mm.
Flange Outside Diameter	A	715.000	mm.
Flange Thickness	t	62.0000	mm.
Thickness of Hub at Small End	go	10.0000	mm.
Thickness of Hub at Large End	gl	17.0000	mm.
Length of Hub	h	26.0000	mm.
Flange Material		SA-350 LF2	
Flange Material UNS number		K03011	
Flange Allowable Stress At Temperature	Sfo	137.90	N./mm ²
Flange Allowable Stress At Ambient	Sfa	137.90	N./mm ²
Bolt Material		SA-320 L7	
Bolt Allowable Stress At Temperature	Sb	172.38	N./mm ²
Bolt Allowable Stress At Ambient	Sa	172.38	N./mm ²
Diameter of Bolt Circle	C	673.000	mm.
Nominal Bolt Diameter	a	19.0500	mm.
Type of Threads		UNC Thread Series	
Number of Bolts		32	
Flange Face Outside Diameter	Fod	644.000	mm.
Flange Face Inside Diameter	Fid	581.000	mm.
Flange Facing Sketch		1, Code Sketch 1a	
Gasket Outside Diameter	Go	641.000	mm.
Gasket Inside Diameter	Gi	611.000	mm.
Gasket Factor	m	3.7800	
Gasket Design Seating Stress	y	62.05	N./mm ²
Column for Gasket Seating		2, Code Column II	
Gasket Thickness	tg	3.0000	mm.
Length of Partition Gasket	lp	1078.0000	mm.
Width of Partition Gasket	tp	6.0000	mm.
Partition Gasket Factor	mPart	3.7500	
Partition Gasket Design Seating Stress	yPart	62.05	N./mm ²



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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Flg Calc [Int P]: New Flange Flng: 4 11:46pm Dec 22,2021

ASME Code, Section VIII Division 1, 2017

Hub Small End Required Thickness due to Internal Pressure:

$$= (P*(D/2+Ca))/(S*E-0.6*P) \text{ per UG-27 (c)(1)}$$

$$= (23.03*(581.0/2+3.0))/(137.9*1.0-0.6*23.03)+Ca$$

$$= 7.9524 \text{ mm.}$$

Hub Small End Hub MAWP:

$$= (S*E*t)/(R+0.6*t) \text{ per UG-27 (c)(1)}$$

$$= (137.9 * 1.0 * 7.0)/(293.5 + 0.6 * 7.0)$$

$$= 32.423 \text{ bars}$$

Corroded Flange Thickness, $t_c = T - c_i$	59.000	mm.
Corroded Flange ID, $B_{cor} = B + 2 * F_{cor}$	587.000	mm.
Corroded Large Hub, $g_{lCor} = g_l - c_i$	14.000	mm.
Corroded Small Hub, $g_{oCor} = g_o - c_i$	7.000	mm.
Code R Dimension, $R = ((C - B_{cor})/2) - g_{lCor}$	29.000	mm.
Gasket Contact Width, $N = (G_o - G_i) / 2$	15.000	mm.
Basic Gasket Width, $b_o = N / 2$	7.500	mm.
Effective Gasket Width, $b = C_b \text{ sqrt}(b_o)$	6.901	mm.
Gasket Reaction Diameter, $G = G_o - 2 * b$	627.198	mm.

Basic Flange and Bolt Loads:

Hydrostatic End Load due to Pressure [H]:

$$= 0.785 * G^2 * P_{eq}$$

$$= 0.785 * 627.1979^2 * 23.034$$

$$= 711.639 \text{ kN}$$

Contact Load on Gasket Surfaces [Hp]:

$$= 2 * b * P_i * G * m * P + 2 * l_p * b_{Part} * m_{Part} * P$$

$$= 2 * 6.9011 * 3.1416 * 627.1979 * 3.78 * 23.03$$

$$+ 2.0 * 1078.0 * 3.0 * 3.75 * 23.0342$$

$$= 292.653 \text{ kN}$$

Hydrostatic End Load at Flange ID [Hd]:

$$= P_i * B_{cor}^2 * P / 4$$

$$= 3.1416 * 587.0^2 * 23.0342 / 4$$

$$= 623.343 \text{ kN}$$

Pressure Force on Flange Face [Ht]:

$$= H - H_d$$

$$= 712 - 623$$

$$= 88.296 \text{ kN}$$

Operating Bolt Load [Wm1]:

$$= \max(H + H_p + H'p, 0)$$

$$= \max(712 + 293 + 0, 0)$$

$$= 1004.292 \text{ kN}$$

Gasket Seating Bolt Load [Wm2]:

$$= y * b * P_i * G + y_{Part} * b_{Part} * l_p$$

$$= 62.05 * 6.9011 * 3.141 * 627.198 + 62.05 * 3.0 * 1078.0$$

$$= 1044.378 \text{ kN}$$

Required Bolt Area [Am]:

$$= \text{Maximum of } W_{m1}/S_b, W_{m2}/S_a$$

$$= \text{Maximum of } 1004/172, 1044/172$$

$$= 60.593 \text{ cm}^2$$

ASME Maximum Circumferential Spacing between Bolts per App. 2 eq. (3) [Bsmax]:

$$= 2a + 6t/(m + 0.5)$$

$$= 2 * 19.05 + 6 * 59.0/(3.78 + 0.5)$$

$$= 120.810 \text{ mm.}$$

Actual Circumferential Bolt Spacing [Bs]:

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Flg Calc [Int P]: New Flange Flng: 4 11:46pm Dec 22,2021

$$\begin{aligned}
 &= C * \sin(\pi / n) \\
 &= 673.0 * \sin(3.142/32) \\
 &= 65.966 \text{ mm.}
 \end{aligned}$$

ASME Moment Multiplier for Bolt Spacing per App. 2 eq. (7) [Bsc]:

$$\begin{aligned}
 &= \max(\text{sqrt}(Bs / (2a + t)), 1) \\
 &= \max(\text{sqrt}(65.966 / (2 * 19.05 + 59.0)), 1) \\
 &= 1.0000
 \end{aligned}$$

Bolting Information for UNC Thread Series (Non Mandatory):

	Minimum	Actual	Maximum
Bolt Area, cm ²	60.593	62.348	
Radial Distance between Hub and Bolts:	28.575	29.000	
Radial Distance between Bolts and Edge:	20.637	21.000	
Circ. Spacing between the Bolts:	44.450	65.966	120.810

Min. Gasket Contact Width (Brownell Young) [Not an ASME Calc] [Nmin]:

$$\begin{aligned}
 &= Ab * Sa / (y * \pi * (Go + Gi)) \\
 &= 62.348 * 172.38 / (62.05 * 3.14 * (641.0 + 611.0)) \\
 &= 4.403 \text{ mm.}
 \end{aligned}$$

Flange Design Bolt Load, Gasket Seating [W]:

$$\begin{aligned}
 &= Sa * (Am + Ab) / 2 \\
 &= 172.38 * (60.5927 + 62.3483) / 2 \\
 &= 1059.51 \text{ kN}
 \end{aligned}$$

Gasket Load for the Operating Condition [HG]:

$$\begin{aligned}
 &= Wm1 - H \\
 &= 1004 - 712 \\
 &= 292.65 \text{ kN}
 \end{aligned}$$

Moment Arm Calculations:

Distance to Gasket Load Reaction [hg]:

$$\begin{aligned}
 &= (C - G) / 2 \\
 &= (673.0 - 627.1979) / 2 \\
 &= 22.9011 \text{ mm.}
 \end{aligned}$$

Distance to Face Pressure Reaction [ht]:

$$\begin{aligned}
 &= (R + g1 + hg) / 2 \\
 &= (29.0 + 14.0 + 22.9011) / 2 \\
 &= 32.9505 \text{ mm.}
 \end{aligned}$$

Distance to End Pressure Reaction [hd]:

$$\begin{aligned}
 &= R + (g1 / 2) \\
 &= 29.0 + (14.0 / 2.0) \\
 &= 36.0000 \text{ mm.}
 \end{aligned}$$

Summary of Moments for Internal Pressure: (N-m)

Loading	Force	Distance	Bolt Corr	Moment
End Pressure, Md	623.	36.0000	1.0000	22449.
Face Pressure, Mt	88.	32.9505	1.0000	2911.
Gasket Load, Mg	293.	22.9011	1.0000	6705.
Gasket Seating, Matm	1060.	22.9011	1.0000	24274.
Total Moment for Operation, Mop				32065. N-m
Total Moment for Gasket seating, Matm				24274. N-m
Effective Hub Length, ho = sqrt(Bcor*goCor)			64.101 mm.	
Hub Ratio, h/h0 = HL / H0			0.406	

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Flg Calc [Int P]: New Flange Flng: 4 11:46pm Dec 22,2021

Thickness Ratio, $g1/g0 = (g1Cor/goCor)$ 2.000

Flange Factors for Integral Flange:

Factor F 0.847
 Factor V 0.257
 Factor f 1.586
 Factors from Figure 2-7.1 K = 1.218
 T = 1.832 U = 10.940
 Y = 9.955 Z = 5.135
 d = 0.13358E+06 mm.³ e = 0.0132 mm.⁻¹
 Stress Factors ALPHA = 1.779
 BETA = 2.039 GAMMA = 0.972
 DELTA = 1.537 Lamda = 2.509

Longitudinal Hub Stress, Operating [SHo]:

$$= (f * Mop / Bcor) / (L * g1^2)$$

$$= (1.5858 * 32065 / 587.0) / (2.5089 * 14.0^2)$$

$$= 176.10 \text{ N./mm}^2$$

Longitudinal Hub Stress, Seating [SHa]:

$$= (f * Matm / Bcor) / (L * g1^2)$$

$$= (1.5858 * 24274 / 587.0) / (2.5089 * 14.0^2)$$

$$= 133.31 \text{ N./mm}^2$$

Radial Flange Stress, Operating [SRo]:

$$= (Beta * Mop / Bcor) / (L * t^2)$$

$$= (2.0393 * 32065 / 587.0) / (2.5089 * 59.0^2)$$

$$= 12.75 \text{ N./mm}^2$$

Radial Flange Stress, Seating [SRa]:

$$= (Beta * Matm / Bcor) / (L * t^2)$$

$$= (2.0393 * 24274 / 587.0) / (2.5089 * 59.0^2)$$

$$= 9.65 \text{ N./mm}^2$$

Tangential Flange Stress, Operating [STo]:

$$= (Y * Mo / (t^2 * Bcor)) - Z * SRO$$

$$= (9.9551 * 32065 / (59.0^2 * 587.0)) - 5.1351 * 13$$

$$= 90.69 \text{ N./mm}^2$$

Tangential Flange Stress, Seating [STa]:

$$= (y * Matm / (t^2 * Bcor)) - Z * SRA$$

$$= (9.9551 * 24274 / (59.0^2 * 587.0)) - 5.1351 * 10$$

$$= 68.66 \text{ N./mm}^2$$

Average Flange Stress, Operating [SAo]:

$$= (SHo + \max(SRO, STo)) / 2$$

$$= (176 + \max(13, 91)) / 2$$

$$= 133.40 \text{ N./mm}^2$$

Average Flange Stress, Seating [SAa]:

$$= (SHa + \max(SRA, STa)) / 2$$

$$= (133 + \max(10, 69)) / 2$$

$$= 100.98 \text{ N./mm}^2$$

Bolt Stress, Operating [BSo]:

$$= Wm1 / Ab$$

$$= 1004 / 62.3483$$

$$= 161.09 \text{ N./mm}^2$$

Bolt Stress, Seating [BSa]:

$$= (Wm2 / Ab)$$

$$= (1044 / 62.3483)$$

$$= 167.52 \text{ N./mm}^2$$

Flange Stress Analysis Results: N./mm²

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Flg Calc [Int P]: New Flange Flng: 4 11:46pm Dec 22,2021

	Actual	Operating Allowed	Gasket Seating Actual	Gasket Seating Allowed
Longitudinal Hub	176.	207.	133.	207.
Radial Flange	13.	138.	10.	138.
Tangential Flange	91.	138.	69.	138.
Maximum Average	133.	138.	101.	138.
Bolting	161.	172.	168.	172.

Minimum Required Flange Thickness 60.808 mm.
 Estimated M.A.W.P. (Operating) 23.812 bars
 Estimated Finished Weight of Flange at given Thk. 70.6 kg.
 Estimated Unfinished Weight of Forging at given Thk 93.0 kg.

Flange Rigidity Based on Required Thickness [ASME]:

Flange Rigidity Index, Seating (rotation check) per APP. 2 [Js]:

$$= 52.14 * Ma / Bsc * Cnv_fac * V / (Lambda * Eamb * go^2 * ho * Ki)$$

$$= 52.14 * 24273.7/1.0 * 999.68 * 0.257/(2.409 * 202713 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.707 \text{ (should be } \leq 1)$$

Flange Rigidity Index Operating (rotation check) per APP. 2 [J]:

$$= 52.14 * Mo / Bsc * Cnv_fac * V / (Lambda * Eop * goc^2 * ho * Ki)$$

$$= 52.14 * 32064.8/1.0 * 999.68 * 0.257/(2.409 * 196922 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.962 \text{ (should be } \leq 1)$$

Flange Rigidity Based on Given Thickness [ASME]:

Flange Rigidity Index, Seating (rotation check) per APP. 2 [Js]:

$$= 52.14 * Ma / Bsc * Cnv_fac * V / (Lambda * Eamb * go^2 * ho * Ki)$$

$$= 52.14 * 24273.7/1.0 * 999.68 * 0.257/(2.509 * 202713 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.679 \text{ (should be } \leq 1)$$

Flange Rigidity Index Operating (rotation check) per APP. 2 [J]:

$$= 52.14 * Mo / Bsc * Cnv_fac * V / (Lambda * Eop * goc^2 * ho * Ki)$$

$$= 52.14 * 32064.8/1.0 * 999.68 * 0.257/(2.509 * 196922 * 7.0^2 * 64.101 * 0.3)$$

$$= 0.923 \text{ (should be } \leq 1)$$

Minimum Design Metal Temperature Results:*Note:**This Material was specified as being an Impact Tested (Low Temperature) Material.*

Impact Test Temperature provided per Specification -46 °C

*Note: UCS-66(b)-(c) was considered in the flange MDMT calculation.***PV Elite is a trademark of Intergraph CADWorx & Analysis Solutions, Inc. 2019**

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Internal Pressure Calculations: Step: 5 11:46pm Dec 22,2021

Element Thickness, Pressure, Diameter and Allowable Stress :

From	To	Int. Press + Liq. Hd bars	Nominal Thickness mm.	Total Corr Allowance mm.	Element Diameter mm.	Allowable Stress(SE) N./mm ²
HEAD 1		23.034	12	3	581	137.9
CHANNEL 01		23.034	10	3	581	137.9
BODY FLANGE 01		23.037	79	3	581	137.9
SHELL		23.034	10	3	581	137.9
BODY FLANGE 002		23.034	79	3	581	137.9
CHANNEL 002		23.034	10	3	581	137.9
HEAD 002		23.034	12	3	581	137.9

Element Required Thickness and MAWP :

From	To	Design Pressure bars	M.A.W.P. Corroded bars	M.A.P. New & Cold bars	Minimum Thickness mm.	Required Thickness mm.
HEAD 1		23	No Calc	No Calc	10	7.8451
CHANNEL 01		23	No Calc	No Calc	10	7.95245
BODY FLANGE 01		23	No Calc	No Calc	62	60.833
SHELL		23	No Calc	No Calc	10	7.95245
BODY FLANGE 002		23	No Calc	No Calc	62	60.8076
CHANNEL 002		23	No Calc	No Calc	10	7.95245
HEAD 002		23	No Calc	No Calc	10	7.8451

Internal Pressure Calculation Results :

ASME Code, Section VIII Division 1, 2017

Elliptical Head From 10 To 20 SA-516 70 , UCS-66 Crv. D at 120 °C

HEAD 1

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$= (P \cdot D \cdot K_{cor}) / (2 \cdot S \cdot E - 0.2 \cdot P) \text{ Appendix 1-4(c)}$$

$$= (23.034 \cdot 587.0 \cdot 0.987) / (2 \cdot 137.9 \cdot 1.0 - 0.2 \cdot 23.034)$$

$$= 4.8451 + 3.0000 = 7.8451 \text{ mm.}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P \cdot (K_{cor} \cdot D + 0.2 \cdot t)) / (2 \cdot E \cdot t)$$

$$= (23.034 \cdot (0.987 \cdot 587.0 + 0.2 \cdot 7.0)) / (2 \cdot 1.0 \cdot 7.0)$$

$$= 95.519 \text{ N./mm}^2$$

Straight Flange Required Thickness:

$$= (P \cdot R) / (S \cdot E - 0.6 \cdot P) + c \text{ per UG-27 (c)(1)}$$

$$= (23.034 \cdot 293.5) / (137.9 \cdot 1.0 - 0.6 \cdot 23.034) + 3.0$$

$$= 7.952 \text{ mm.}$$

Straight Flange Maximum Allowable Working Pressure:

Less Operating Hydrostatic Head Pressure of 0.034 bars

$$= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c)(1)}$$

$$= (137.9 \cdot 1.0 \cdot 9.0) / (293.5 + 0.6 \cdot 9.0)$$

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Internal Pressure Calculations: Step: 5 11:46pm Dec 22,2021

$$= 41.520 - 0.034 = 41.485 \text{ bars}$$

Factor K, corroded condition [Kcor]:

$$= (2 + (\text{Inside Diameter}/(2 * \text{Inside Head Depth}))^2)/6$$

$$= (2 + (587.0/(2 * 148.25))^2)/6$$

$$= 0.986578$$

Percent Elong. per UCS-79, VIII-1-01-57 $(75 * t_{nom}/R_f) * (1 - R_f/R_o)$ 8.590 %

Note: Please Check Requirements of UCS-79 as Elongation is > 5%.

MDMT Calculations in the Knuckle Portion:

Govrn. thk, tg = 10.0, tr = 4.845, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.692$, Temp. Reduction = 17 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

MDMT Calculations in the Head Straight Flange:

Govrn. thk, tg = 12.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.55$, Temp. Reduction = 27 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

Cylindrical Shell From 20 To 30 SA-516 70 , UCS-66 Crv. D at 120 °C

CHANNEL 01

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$= (P * R) / (S * E - 0.6 * P) \text{ per UG-27 (c)(1)}$$

$$= (23.034 * 293.5) / (137.9 * 1.0 - 0.6 * 23.034)$$

$$= 4.9524 + 3.0000 = 7.9524 \text{ mm.}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P * (R + 0.6 * t)) / (E * t)$$

$$= (23.034 * (293.5 + 0.6 * 7.0)) / (1.0 * 7.0)$$

$$= 97.967 \text{ N./mm}^2$$

% Elongation per Table UG-79-1 $(50 * t_{nom}/R_f) * (1 - R_f/R_o)$ 1.692 %

Minimum Design Metal Temperature Results:

Govrn. thk, tg = 10.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.707$, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

Cylindrical Shell From 40 To 50 SA-516 70 , UCS-66 Crv. D at 120 °C

SHELL

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$= (P * R) / (S * E - 0.6 * P) \text{ per UG-27 (c)(1)}$$

$$= (23.034 * 293.5) / (137.9 * 1.0 - 0.6 * 23.034)$$

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Internal Pressure Calculations: Step: 5 11:46pm Dec 22,2021

$$= 4.9524 + 3.0000 = 7.9524 \text{ mm.}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P*(R+0.6*t))/(E*t)$$

$$= (23.034*(293.5+0.6*7.0))/(1.0*7.0)$$

$$= 97.967 \text{ N./mm}^2$$

% Elongation per Table UG-79-1 (50*tnom/Rf*(1-Rf/Ro)) 1.692 %

Minimum Design Metal Temperature Results:

Govern. thk, tg = 10.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.707, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

Cylindrical Shell From 60 To 70 SA-516 70 , UCS-66 Crv. D at 120 °C

CHANNEL 002

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$= (P*R)/(S*E-0.6*P) \text{ per UG-27 (c)(1)}$$

$$= (23.034*293.5)/(137.9*1.0-0.6*23.034)$$

$$= 4.9524 + 3.0000 = 7.9524 \text{ mm.}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P*(R+0.6*t))/(E*t)$$

$$= (23.034*(293.5+0.6*7.0))/(1.0*7.0)$$

$$= 97.967 \text{ N./mm}^2$$

% Elongation per Table UG-79-1 (50*tnom/Rf*(1-Rf/Ro)) 1.692 %

Minimum Design Metal Temperature Results:

Govern. thk, tg = 10.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.707, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

Elliptical Head From 70 To 80 SA-516 70 , UCS-66 Crv. D at 120 °C

HEAD 002

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$= (P*D*Kcor)/(2*S*E-0.2*P) \text{ Appendix 1-4(c)}$$

$$= (23.034*587.0*0.987)/(2*137.9*1.0-0.2*23.034)$$

$$= 4.8451 + 3.0000 = 7.8451 \text{ mm.}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P*(Kcor*D+0.2*t))/(2*E*t)$$

$$= (23.034*(0.987*587.0+0.2*7.0))/(2*1.0*7.0)$$

$$= 95.519 \text{ N./mm}^2$$

Straight Flange Required Thickness:

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Internal Pressure Calculations: Step: 5 11:46pm Dec 22,2021

$$= (P \cdot R) / (S \cdot E - 0.6 \cdot P) + c \quad \text{per UG-27 (c)(1)}$$

$$= (23.034 \cdot 293.5) / (137.9 \cdot 1.0 - 0.6 \cdot 23.034) + 3.0$$

$$= 7.952 \text{ mm.}$$

Straight Flange Maximum Allowable Working Pressure:

Less Operating Hydrostatic Head Pressure of 0.034 bars

$$= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \quad \text{per UG-27 (c)(1)}$$

$$= (137.9 \cdot 1.0 \cdot 9.0) / (293.5 + 0.6 \cdot 9.0)$$

$$= 41.520 - 0.034 = 41.485 \text{ bars}$$

Factor K, corroded condition [Kcor]:

$$= (2 + (\text{Inside Diameter} / (2 \cdot \text{Inside Head Depth}))^2) / 6$$

$$= (2 + (587.0 / (2 \cdot 148.25))^2) / 6$$

$$= 0.986578$$

Percent Elong. per UCS-79, VIII-1-01-57 $(75 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 8.590 %

Note: Please Check Requirements of UCS-79 as Elongation is > 5%.

MDMT Calculations in the Knuckle Portion:

Govrn. thk, tg = 10.0, tr = 4.845, c = 3.0 mm., E* = 1.0

Thickness Ratio = $tr \cdot (E^*) / (tg - c) = 0.692$, Temp. Reduction = 17 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

MDMT Calculations in the Head Straight Flange:

Govrn. thk, tg = 12.0, tr = 4.952, c = 3.0 mm., E* = 1.0

Thickness Ratio = $tr \cdot (E^*) / (tg - c) = 0.55$, Temp. Reduction = 27 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C

Note: Heads and Shells Exempted to -20F (-29C) by paragraph UG-20F

Hydrostatic Test Pressure Results:**Exchanger Shell Side Hydrostatic Test Pressures:**

Pressure per UG99b[36]	= 1.30 * Design Pres * Sa/S	29.900 bars
Pressure per PED	= max(1.43*DP, 1.25*DP*ratio)	32.890 bars
Pressure per App 27-4	= M.A.W.P.	43.964 bars

Exchanger Channel Side Hydrostatic Test Pressures:

Pressure per UG99b	= 1.30 * M.A.W.P. * Sa/S	32.043 bars
Pressure per UG99b[36]	= 1.30 * Design Pres * Sa/S	29.900 bars
Pressure per UG99c	= 1.30 * M.A.P. - Head(Hyd)	31.986 bars
Pressure per UG100	= 1.10 * M.A.W.P. * Sa/S	27.113 bars
Pressure per PED	= max(1.43*DP, 1.25*DP*ratio)	32.890 bars
Pressure per App 27-4	= M.A.W.P.	24.648 bars

UG-99(b) Note 36, Test Pressure Calculation [Shell Side]:

$$= \text{Test Factor} \cdot \text{Design Pressure} \cdot \text{Stress Ratio}$$

$$= 1.3 \cdot 23.0 \cdot 1.0$$

$$= 29.900 \text{ bars}$$

UG-99(b) Note 36, Test Pressure Calculation [Channel Side]:

$$= \text{Test Factor} \cdot \text{Design Pressure} \cdot \text{Stress Ratio}$$

$$= 1.3 \cdot 23.0 \cdot 1.0$$

$$= 29.900 \text{ bars}$$

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Internal Pressure Calculations: Step: 5 11:46pm Dec 22,2021

Horizontal Test performed per: UG-99b (Note 36)

Please note that Nozzle, Shell, Head, Flange, etc MAWPs are all considered when determining the hydrotest pressure for those test types that are based on the MAWP of the vessel.

Stresses on Elements due to Test Pressure (N./mm² & bars):

From To	Stress	Allowable	Ratio	Pressure
HEAD 1	124.2	235.8	0.527	29.96
CHANNEL 01	127.4	235.8	0.540	29.96
SHELL	127.4	235.8	0.540	29.96
CHANNEL 002	127.4	235.8	0.540	29.96
HEAD 002	124.2	235.8	0.527	29.96

Stress ratios for Nozzle and Pad Materials (N./mm²):

Description	Pad/Nozzle	Ambient	Operating	Ratio
T4	Nozzle	137.90	137.90	1.000
T1	Nozzle	117.90	117.90	1.000
T1	Pad	137.90	137.90	1.000
S2	Nozzle	117.90	117.90	1.000
S2	Pad	137.90	137.90	1.000
S1	Nozzle	117.90	117.90	1.000
S1	Pad	137.90	137.90	1.000
S3	Nozzle	117.90	117.90	1.000
S3	Pad	137.90	137.90	1.000
T2	Nozzle	117.90	117.90	1.000
T2	Pad	137.90	137.90	1.000
T3	Nozzle	137.90	137.90	1.000
Minimum				1.000

Stress ratios for Pressurized Vessel Elements (N./mm²):

Description	Ambient	Operating	Ratio
HEAD 1	137.90	137.90	1.000
CHANNEL 01	137.90	137.90	1.000
BODY FLANGE 01	137.90	137.90	1.000
SHELL	137.90	137.90	1.000
BODY FLANGE 002	137.90	137.90	1.000
CHANNEL 002	137.90	137.90	1.000
HEAD 002	137.90	137.90	1.000
Minimum			1.000

Stress ratios for Exchanger Materials (N./mm²):

Description	Ambient	Operating	Ratio
Tube Material	117.90	117.90	1.000
Tubesheet Material	137.90	137.90	1.000
Minimum			1.000

Hoop Stress in Nozzle Wall during Pressure Test (N./mm²):

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 Internal Pressure Calculations: Step: 5 11:46pm Dec 22,2021

Description	Ambient	Operating	Ratio
T4	5.13	223.40	0.023
T1	24.22	217.19	0.112
S2	36.99	217.19	0.170
S1	36.99	217.19	0.170
S3	18.25	217.19	0.084
T2	36.99	217.19	0.170
T3	5.96	223.40	0.027

Elements Suitable for Internal Pressure.

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 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 External Pressure Calculations: Step: 6 11:46pm Dec 22,2021

External Pressure Calculation Results :

External Pressure Calculations:

From	To	Section Length mm.	Outside Diameter mm.	Corroded Thickness mm.	Factor A	Factor B N./mm ²
10	20	No Calc	601	7	0.0016177	98.7135
20	30	505.417	601	7	0.0020537	104.349
30	40	No Calc	...	59	No Calc	No Calc
40	50	5880	601	7	0.00016747	16.7432
50	60	No Calc	...	59	No Calc	No Calc
60	70	505.417	601	7	0.0020537	104.349
70	80	No Calc	601	7	0.0016177	98.7135

External Pressure Calculations:

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	20	10	4.60484	1.1	12.7742
20	30	10	4.83369	1.1	16.2041
30	40	62	50.8	1.1	No Calc
40	50	10	8.13291	1.1	2.60002
50	60	62	50.8	1.1	No Calc
60	70	10	4.83369	1.1	16.2041
70	80	10	4.60484	1.1	12.7742

Minimum 2.600

External Pressure Calculations:

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	20	No Calc	No Calc	No Calc	No Calc
20	30	505.417	12031.9	No Calc	No Calc
30	40	No Calc	No Calc	No Calc	No Calc
40	50	5880	131539	No Calc	No Calc
50	60	No Calc	No Calc	No Calc	No Calc
60	70	505.417	12031.9	No Calc	No Calc
70	80	No Calc	No Calc	No Calc	No Calc

Elements Suitable for External Pressure.

ASME Code, Section VIII Division 1, 2017

Elliptical Head From 10 to 20 Ext. Chart: CS-2 at 120 °C

HEAD 1

Elastic Modulus from Chart: CS-2 at 120 °C : 0.200E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca OD D/t Factor A B

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External Pressure Calculations: Step: 6 11:46pm Dec 22,2021

$$\begin{array}{r} 7.000 \quad 601.00 \quad 85.86 \quad 0.0016177 \quad 98.71 \\ \text{EMAP} = B / (K0 * D / t) = 98.7135 / (0.9 * 85.8571) = 12.7742 \text{ bars} \end{array}$$

Results for Required Thickness (Tca):

Tca	OD	D/t	Factor A	B
1.605	601.00	374.49	0.0003709	37.08

$$\text{EMAP} = B / (K0 * D / t) = 37.0789 / (0.9 * 374.4927) = 1.1001 \text{ bars}$$

*Check the requirements of UG-33(a)(1) using $P = 1.67 * \text{External Design pressure}$ for this head.*

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P * D * K_{cor}) / (2 * S * E - 0.2 * P) \text{ Appendix 1-4(c)} \\ &= (1.837 * 587.0 * 0.987) / (2 * 137.9 * 1.0 - 0.2 * 1.837) \\ &= 0.3858 + 3.0000 = 3.3858 \text{ mm.} \end{aligned}$$

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= ((2 * S * E * t) / (K_{cor} * D + 0.2 * t)) / 1.67 \text{ per Appendix 1-4 (c)} \\ &= ((2 * 137.9 * 1.0 * 7.0) / (0.987 * 587.0 + 0.2 * 7.0)) / 1.67 \\ &= 19.913 \text{ bars} \end{aligned}$$

Maximum Allowable External Pressure [MAEP]:

$$\begin{aligned} &= \min(\text{MAEP}, \text{MAWP}) \\ &= \min(12.77, 19.9128) \\ &= 12.774 \text{ bars} \end{aligned}$$

Thickness requirements per UG-33(a)(1) do not govern the required thickness of this head.

Cylindrical Shell From 20 to 30 Ext. Chart: CS-2 at 120 °C

CHANNEL 01

Elastic Modulus from Chart: CS-2 at 120 °C : 0.200E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
7.000	601.00	505.42	85.86	0.8410	0.0020537	104.35

$$\text{EMAP} = (4 * B) / (3 * (D / t)) = (4 * 104.3486) / (3 * 85.8571) = 16.2041 \text{ bars}$$

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.834	601.00	505.42	327.75	0.8410	0.0002705	27.04

$$\text{EMAP} = (4 * B) / (3 * (D / t)) = (4 * 27.043) / (3 * 327.7541) = 1.1001 \text{ bars}$$

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
7.000	601.00	12031.87	85.86	20.0198	0.0001535	15.35

$$\text{EMAP} = (4 * B) / (3 * (D / t)) = (4 * 15.3457) / (3 * 85.8571) = 2.383 \text{ bars}$$

Cylindrical Shell From 40 to 50 Ext. Chart: CS-2 at 120 °C

SHELL

Elastic Modulus from Chart: CS-2 at 120 °C : 0.200E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
-----	----	------	-----	-----	----------	---

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

External Pressure Calculations: Step: 6 11:46pm Dec 22,2021

7.000 601.00 5880.00 85.86 9.7837 0.0001675 16.74
 $EMAP = (4*B)/(3*(D/t)) = (4*16.7432)/(3*85.8571) = 2.6 \text{ bars}$

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
5.133	601.00	5880.00	117.09	9.7837	0.0000966	9.66

$EMAP = (4*B)/(3*(D/t)) = (4*9.6607)/(3*117.0875) = 1.1 \text{ bars}$

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
7.000	601.00	131538.89	85.86	50.0000	0.0001516	15.15

$EMAP = (4*B)/(3*(D/t)) = (4*15.1537)/(3*85.8571) = 2.3532 \text{ bars}$

Cylindrical Shell From 60 to 70 Ext. Chart: CS-2 at 120 °C

CHANNEL 002

Elastic Modulus from Chart: CS-2 at 120 °C : 0.200E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
7.000	601.00	505.42	85.86	0.8410	0.0020537	104.35

$EMAP = (4*B)/(3*(D/t)) = (4*104.3486)/(3*85.8571) = 16.2041 \text{ bars}$

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.834	601.00	505.42	327.75	0.8410	0.0002705	27.04

$EMAP = (4*B)/(3*(D/t)) = (4*27.043)/(3*327.7541) = 1.1001 \text{ bars}$

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
7.000	601.00	12031.87	85.86	20.0198	0.0001535	15.35

$EMAP = (4*B)/(3*(D/t)) = (4*15.3457)/(3*85.8571) = 2.383 \text{ bars}$

Elliptical Head From 70 to 80 Ext. Chart: CS-2 at 120 °C

HEAD 002

Elastic Modulus from Chart: CS-2 at 120 °C : 0.200E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	D/t	Factor A	B
7.000	601.00	85.86	0.0016177	98.71

$EMAP = B/(K0*D/t) = 98.7135/(0.9 * 85.8571) = 12.7742 \text{ bars}$

Results for Required Thickness (Tca):

Tca	OD	D/t	Factor A	B
1.605	601.00	374.49	0.0003709	37.08

$EMAP = B/(K0*D/t) = 37.0789/(0.9 * 374.4927) = 1.1001 \text{ bars}$

*Check the requirements of UG-33(a)(1) using $P = 1.67 * \text{External Design pressure for this head}$.*

Material UNS Number: K02700

Required Thickness due to Internal Pressure [tr]:

$$= (P*D*K_{cor})/(2*S*E-0.2*P) \text{ Appendix 1-4(c)}$$

$$= (1.837*587.0*0.987)/(2*137.9*1.0-0.2*1.837)$$

$$= 0.3858 + 3.0000 = 3.3858 \text{ mm.}$$

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External Pressure Calculations: Step: 6 11:46pm Dec 22,2021

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:
= $((2*S*E*t)/(Kcor*D+0.2*t))/1.67$ per Appendix 1-4 (c)
= $((2*137.9*1.0*7.0)/(0.987*587.0+0.2*7.0))/1.67$
= 19.913 bars

Maximum Allowable External Pressure [MAEP]:
= min(MAEP, MAWP)
= min(12.77, 19.9128)
= 12.774 bars

Thickness requirements per UG-33(a)(1) do not govern the required thickness of this head.

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 Element and Detail Weights: Step: 7 11:46pm Dec 22,2021

Element and Detail Weights:

From	To	Element Metal Wgt. kg.	Element ID Volume Cm3	Corroded Metal Wgt. kg.	Corroded ID Volume Cm3	Extra due Misc % kg.
10	20	47.807	38935.4	35.8552	40014.4	2.39035
20	30	58.5682	107923	41.2058	110164	2.92841
30	40	88.5944	28277.5	84.1151	28420.6	4.42972
40	50	846.145	841010	595.307	873380	42.3073
50	60	88.5944	28277.5	84.1151	28420.6	4.42972
60	70	58.5682	107923	41.2058	110164	2.92841
70	80	47.807	38935.4	35.8552	40014.4	2.39035
Total		1236	1191281.88	917	1230577.00	61

For elements specified as shell side elements, the volume(s) shown above for those elements, reflects the displacement of the tubes.

Weight of Details:

From	Type	Weight of Detail kg.	X Offset, Dtl. Cent. mm.	Y Offset, Dtl. Cent. mm.	Description
10	Liqd	23.347	-48.4167	0.18167E-04	1
10	Insl	6.82273	-47.625	...	Ins: 10
20	Liqd	64.7144	203.5	...	2
20	Insl	8.3585	203.5	...	Ins: 20
20	Nozl	6.25877	203	300.025	T4
20	Nozl	31.9986	203	347.65	T1
30	Liqd	19.1661	44	20.5	3
30	Insl	2.26866	44	...	Ins: 30
40	Sadl	60.6827	940	435.25	Fixed Saddle
40	Sadl	60.6827	4940	435.25	Sliding Saddle
40	Liqd	504.298	2940	...	4
40	Insl	120.757	2940	...	Ins: 40
40	Nozl	59.6222	196	374.637	S2
40	Nozl	59.6222	5680	374.637	S1
40	Nozl	12.683	146	320.663	S3
40	Wght	150	2940	...	WEIGHT BAFFLE
50	Liqd	16.9561	44	...	5
50	Insl	2.26866	44	...	Ins: 50
60	Liqd	64.7144	203.5	...	5
60	Insl	8.3585	203.5	...	Ins: 60
60	Nozl	59.6222	200	374.637	T2
60	Nozl	6.25877	203	303.2	T3
70	Liqd	23.347	98.4167	0.18167E-04	6
70	Insl	6.82273	97.625	...	Ins: 70
30	FTsh	140.973	125	...	TUBE SHEET
30	Tube	1815.16	3094	...	
30	RTsh	140.973	6063	...	

Total Weight of Each Detail Type:

Saddles	121.4
Liquid	716.5
Insulation	155.7
Nozzles	236.1

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 Tag no:E-PK6101-3 ECONOMIZER
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 Element and Detail Weights: Step: 7 11:46pm Dec 22,2021

Weights 150.0
 Exchanger Components 2097.1
 Liquid in Tubes 356.6

 Sum of the Detail Weights 3833.3 kg.

Weight Summation Results: (kg.)

	Fabricated	Shop Test	Shipping	Erected	Empty	Operating
Main Elements	1297.9	1297.9	1297.9	1297.9	1297.9	1297.9
Saddles	121.4	121.4	121.4	121.4	121.4	121.4
Nozzles	236.1	236.1	236.1	236.1	236.1	236.1
Wld Weights	150.0	150.0	150.0	150.0	150.0	150.0
Exchanger	2097.1	2097.1	2097.1	2097.1	2097.1	2097.1
Insulation	155.7	155.7
Ope. Liquid	716.5
Tube Ope Lqd	356.6
Test Liquid	...	1190.6
Tube Tst Lqd	...	509.4
Totals	3902.4	5602.4	3902.4	3902.4	4058.1	5131.2

Field Installation Options:

* Insulation installed after lifting.

Miscellaneous Weight Percent: 5.0 %

Note that the above value for the miscellaneous weight percent has been applied to the shells/heads/flange/tubesheets/tubes etc. in the weight calculations for metallic components.

Weight Summary:

Fabricated Wt. - Bare Weight without Removable Internals 3902.4 kg.
 Shop Test Wt. - Fabricated Weight + Water (Full) 5602.4 kg.
 Shipping Wt. - Fab. Weight + removable Intls.+ Shipping App. 3902.4 kg.
 Erected Wt. - Fab. Wt + or - loose items (trays,platforms etc.) 3902.4 kg.
 Ope. Wt. no Liq - Fab. Weight + Internals. + Details + Weights 4058.1 kg.
 Operating Wt. - Empty Weight + Operating Liq. Uncorroded 5131.2 kg.
 Oper. Wt. + CA - Corr Wt. + Operating Liquid 4796.9 kg.
 Field Test Wt. - Empty Weight + Water (Full) 5463.0 kg.

Exchanger Tube Data

Volume of Exchanger tubes : 509743.7 Cm3
 Weight of Ope Liq in tubes : 356.6 kg.
 Weight of Water in tubes : 509.4 kg.

Note:

The Corroded Weight and thickness are used in the Horizontal Vessel Analysis (Ope Case) and Earthquake Load Calculations.

Note: The Field Test weight as computed in the corroded condition.

Outside Surface Areas of Elements:

From	To	Surface Area
		cm^2
10	20	4917.99

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Tag no:E-PK6101-3 ECONOMIZER
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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
Element and Detail Weights: Step: 7 11:46pm Dec 22,2021

20		30		7684.56	
30		40		3640.82	
40		50		111020	
50		60		3640.82	
60		70		7684.56	
70		80		4917.99	

Total 143506.844 cm^2

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Tag no:E-PK6101-3 ECONOMIZER
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 Nozzle Flange MAWP: Step: 8 11:46pm Dec 22,2021

Nozzle Flange MAWP Results:

Nozzle Description	Flange Rating		Design Temp °C	Class	Grade/Group	Equiv. Press	Max Pressure		
	Ope. bars	Ambient bars					PVP	50%	DNV bars
T4	46.00	51.10	120	300	GR 1.1
T1	46.00	51.10	120	300	GR 1.1
S2	46.00	51.10	120	300	GR 1.1
S1	46.00	51.10	120	300	GR 1.1
S3	46.00	51.10	120	300	GR 1.1
T2	46.00	51.10	120	300	GR 1.1
T3	46.00	51.10	120	300	GR 1.1

Shellside Flange Rating

Lowest Flange Pressure Rating was (Ope)[ShellSide]: 46.000 bars
 Lowest Flange Pressure Rating was (Amb)[ShellSide]: 51.100 bars

Channelside Flange Rating

Lowest Flange Pressure Rating was (Ope)[TubeSide]: 46.000 bars
 Lowest Flange Pressure Rating was (Amb)[TubeSide]: 51.100 bars

Selected Method for Derating ANSI B16.5 Flange MAWP: None Selected

ANSI Ratings are per ANSI/ASME B16.5 2013 Metric Edition

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 Tag no:E-PK6101-3 ECONOMIZER
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 Wind Load Calculation: Step: 9 11:46pm Dec 22,2021

Input Values:

Wind Design Code	ASCE-7 2010
Wind Load Reduction Scale Factor	0.600
Basic Wind Speed [V]	195 Km/hr
Surface Roughness Category	C: Open Terrain
Importance Factor	1.0
Type of Surface	Moderately Smooth
Base Elevation	123000 mm.
Percent Wind for Hydrotest	33.0
Using User defined Wind Press. Vs Elev.	N
Height of Hill or Escarpment H or Hh	0 mm.
Distance Upwind of Crest Lh	0 mm.
Distance from Crest to the Vessel x	0 mm.
Type of Terrain (Hill, Escarpment)	Flat
Damping Factor (Beta) for Wind (Ope)	0.0100
Damping Factor (Beta) for Wind (Empty)	0.0000
Damping Factor (Beta) for Wind (Filled)	0.0000

Wind Analysis Results

Static Gust-Effect Factor, Operating Case [G]:

$$\begin{aligned}
 &= \min(0.85, 0.925((1 + 1.7 * gQ * Izbar * Q)/(1 + 1.7 * gV * Izbar))) \\
 &= \min(0.85, 0.925((1+1.7*3.4*0.143*0.836)/(1+1.7*3.4*0.143))) \\
 &= \min(0.85, 0.856) \\
 &= 0.850
 \end{aligned}$$

Natural Frequency of Vessel (Operating)	33.000 Hz
Natural Frequency of Vessel (Empty)	33.000 Hz
Natural Frequency of Vessel (Test)	33.000 Hz

Force Coefficient [Cf]	0.616
Structure Height to Diameter ratio	9.906

This is classified as a rigid structure. Static analysis performed.

Sample Calculation for the First Element

The ASCE code performs all calculations in Imperial Units only. The wind pressure is therefore computed in these units.

Value of [Alpha] and [Zg]:

Exposure Category: C from Table 26.9.1
 Alpha = 9.5: Zg = 274320. mm.

Effective Height [z]:

$$\begin{aligned}
 &= \text{Centroid Height} + \text{Vessel Base Elevation} \\
 &= 600.0 + 123000. = 123600.008 \text{ mm.} \\
 &= 405.512 \text{ ft. Imperial Units}
 \end{aligned}$$

Velocity Pressure coefficient evaluated at height z [Kz]:

$$\begin{aligned}
 &\text{Because } z (405.512 \text{ ft.}) > 15 \text{ ft.} \\
 &= 2.01 * (z / Zg)^{2 / \text{Alpha}} \\
 &= 2.01 * (405.512/900.0)^{2/9.5} \\
 &= 1.699
 \end{aligned}$$

Type of Hill: No Hill

Wind Directionality Factor [Kd]:

$$= 0.95 \text{ per Table 26.6-1}$$

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Wind Load Calculation: Step: 9 11:46pm Dec 22,2021

As there is No Hill Present: [Kzt]:
 K1 = 0, K2 = 0, K3 = 0

Topographical Factor [Kzt]:
 = (1 + K1 * K2 * K3)²
 = (1 + 0.0* 0.0* 0.0)²
 = 1.0

Velocity Pressure evaluated at height z, Imperial Units [qz]:
 = max(16, 0.00256 * Kz * Kzt * Kd * V(mph)²)
 = max(16, 0.00256 * 1.699 * 1.0 * 0.95 * 121.171²)
 = 60.7 psf [296.282] Kgs/m²

Force on the first element [F]:
 = qz * G * Cf * WindArea
 = 60.682 * 0.85 * 0.616 * 2.005
 = 63.7 lbs. [0.3] kN

Element	Hgt (z) mm.	K1	K2	K3	Kz	Kzt	qz Kgs/m ²
HEAD 1	*****	0.000	0.000	0.000	1.699	1.000	296.282
CHANNEL 01	*****	0.000	0.000	0.000	1.699	1.000	296.282
BODY FLANGE 01	*****	0.000	0.000	0.000	1.699	1.000	296.282
SHELL	*****	0.000	0.000	0.000	1.699	1.000	296.282
BODY FLANGE 002	*****	0.000	0.000	0.000	1.699	1.000	296.282
CHANNEL 002	*****	0.000	0.000	0.000	1.699	1.000	296.282
HEAD 002	*****	0.000	0.000	0.000	1.699	1.000	296.282

Wind Loads on Masses/Equipment/Piping

ID	Wind Area cm ²	Elevation mm.	Pressure Kgs/m ²	Force kN
WEIGHT BAFFLE	0.00	123600.01	296.28	0.00

Wind Load Calculation:

From	To	Wind Height mm.	Wind Diameter mm.	Wind Area cm ²	Wind Pressure Kgs/m ²	Element Wind Load kN
10	20	123600	913.2	1862.66	296.282	0.17005
20	30	123600	913.2	3716.72	296.282	0.33932
30	40	123600	889.2	782.496	296.282	0.071438
40	50	123600	913.2	53696.2	296.282	4.90221
50	60	123600	889.2	782.496	296.282	0.071438
60	70	123600	913.2	3716.72	296.282	0.33932
70	80	123600	913.2	1862.66	296.282	0.17005

Note:
 The Wind Loads calculated and printed in the Wind Load calculation report have been factored by the input scalar/load reduction factor of: 0.600.
 Be sure the wind speed is in accordance with the specified wind design code.

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Tag no:E-PK6101-3 ECONOMIZER
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Wind Load Calculation: Step: 9 11:46pm Dec 22,2021

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 Earthquake Load Calculation: Step: 10 11:46pm Dec 22,2021

Earthquake Load Calculation:

Input Values:

Seismic Design Code		ASCE 7-2010
Seismic Load Reduction Scale Factor		0.700
Importance Factor		1.500
Table Value Fa		1.000
Table Value Fv		1.300
Short Period Acceleration value Ss		1.163
Long Period Acceleration Value S1		0.600
Moment Reduction Factor Tau		1.000
Force Modification Factor R		2.000
Site Class		C
Component Elevation Ratio	z/h	0.000
Amplification Factor	Ap	0.000
Force Factor		0.000
Consider Vertical Acceleration		No
Minimum Acceleration Multiplier		0.000
User Value of Sds (used if > 0)		0.000
User Value of Sd1 (used if > 0)		0.000

Seismic Analysis Results:

Sms = Fa * Ss = 1.0 * 1.163 = 1.163
 Sml = Fv * S1 = 1.3 * 0.6 = 0.78
 Sds = 2/3 * Sms = 2/3 * 1.163 = 0.775
 Sd1 = 2/3 * Sml = 2/3 * 0.78 = 0.52

Check Approximate Fundamental Period from 12.8-7 [Ta]:

= Ct * hn^x where Ct = 0.020, x = 0.75 and hn = Structural Height (ft.)
 = 0.020 * (2.9216^{0.75})
 = 0.045 seconds

The Coefficient Cu from Table 12.8-1 is : 1.400

Fundamental Period (1/Frequency) [T]:

= (1/Natural Frequency) = (1/33.0)
 = 0.030

Check the Value of T which is the smaller of Cu*Ta and T:

= Minimum Value of (1.4 * 0.045, 0.03) per 12.8.2
 = 0.030

As the time period is < 0.06 second, use section 15.4.2.

Compute the Base Shear per equation 15.4-5, [V]:

= 0.3 * Sds * W * I
 = 0.3 * 0.775 * 47 * 1.5
 = 16.412 kN

Final Base Shear, V = 11.49 kN

Earthquake Load Calculation:

From	To	Earthquake Height mm.	Earthquake Weight kN	Element Ope Load kN
10	20	290.5	5.22647	1.27646

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Earthquake Load Calculation: Step: 10 11:46pm Dec 22,2021

20	30	290.5	5.22647	1.27646
30	40	290.5	5.22647	1.27646
40	Sad1	290.5	5.22647	1.27646
Sad1	50	290.5	5.22647	1.27646
40	50	290.5	5.22647	1.27646
50	60	290.5	5.22647	1.27646
60	70	290.5	5.22647	1.27646
70	80	290.5	5.22647	1.27646

Note:

The Earthquake Loads calculated and printed in the Earthquake Load calculation report have been factored by the input scalar/load reduction factor of: 0.700.

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Center of Gravity Calculation: Step: 11 11:46pm Dec 22,2021

Shop/Field Installation Options :

Insulation is installed in the Field after being lifted.

Note : The CG is computed from the first Element From Node

Center of Gravity of Saddles	3553.176 mm.
Center of Gravity of Liquid	3542.295 mm.
Center of Gravity of Insulation	3552.272 mm.
Center of Gravity of Nozzles	3787.192 mm.
Center of Gravity of Added Weights (Operating)	3553.176 mm.
Center of Gravity of Added Weights (Empty)	3553.176 mm.
Center of Gravity of Tubesheet(s)	3551.000 mm.
Center of Gravity of Tubes	3551.000 mm.
Center of Gravity of Bare Shell New and Cold	3548.732 mm.
Center of Gravity of Bare Shell Corroded	3547.492 mm.
Vessel CG in the Operating Condition	3560.783 mm.
Vessel CG in the Fabricated (Shop/Empty) Condition	3564.685 mm.
Vessel CG in the Test Condition	3560.682 mm.

Warning: CG of Vessel is too near or Outside the Lift Points!

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

ASME Horizontal Vessel Analysis: Stresses for the Left Saddle
 (per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Operating Case

Input and Calculated Values:

Vessel Mean Radius	Rm	297.00	mm.
Stiffened Vessel Length per 4.15.6	L	5880.00	mm.
Distance from Saddle to Vessel tangent	a	992.00	mm.
Saddle Width	b	150.00	mm.
Saddle Bearing Angle	theta	120.00	degrees
Wear Plate Width	b1	225.00	mm.
Wear Plate Bearing Angle	thetal	132.00	degrees
Wear Plate Thickness	tr	10.0	mm.
Wear Plate Allowable Stress	Sr	137.90	N./mm^2
Shell Allowable Stress used in Calculation		137.90	N./mm^2
Head Allowable Stress used in Calculation		137.90	N./mm^2
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		42.06	kN
Horizontal Vessel Analysis Results:	Actual	Allowable	
	N./mm^2	N./mm^2	

Long. Stress at Top of Midspan	38.30	137.90	
Long. Stress at Bottom of Midspan	59.36	137.90	
Long. Stress at Top of Saddles	81.35	137.90	
Long. Stress at Bottom of Saddles	30.81	137.90	

Tangential Shear in Shell	15.69	110.32	
Circ. Stress at Horn of Saddle	14.34	172.37	
Circ. Compressive Stress in Shell	2.07	137.90	

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E$$

$$= 3.0 * (6.1/2 + 0) * 600.0/537.8018$$

$$= 10.1 \text{ kN}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= \max(F_l, \text{Friction Load, Sum of X Forces}) * B / L_s$$

$$= \max(0.59, 0.0, 0) * 600.0/4000.0$$

$$= 0.1 \text{ kN}$$

Saddle Reaction Force due to Earthquake Fl or Friction [Fsl]:

$$= \max(F_l, \text{Friction Force, Sum of X Forces}) * B / L_s$$

$$= \max(11.49, 0.0, 0) * 600.0/4000.0$$

$$= 1.7 \text{ kN}$$

Saddle Reaction Force due to Earthquake Ft [Fst]:

$$= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E$$

$$= 3.0 * (11/2 + 0) * 600.0/537.8018$$

$$= 19.2 \text{ kN}$$

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Load Combination Results for Q + Wind or Seismic [Q]:
 = Saddle Load + Max(Fwl, Fwt, Fsl, Fst)
 = 23 + Max(0.1, 10, 2, 19)
 = 42.1 kN

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	42.65	kN
Transverse Shear Load Saddle	5.74	kN
Longitudinal Shear Load Saddle	11.49	kN

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0529	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0434
K7p = 0.0434			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right]$$

$$= -42 \cdot 992.0 \left[1 - \left(1 - \frac{992.0}{5880.0} + \frac{(297.0^2 - 0.0^2)}{(2 \cdot 992.0 \cdot 5880.0)} \right) / \left(1 + \frac{(4 \cdot 0.0)}{3 \cdot 5880.0} \right) \right]$$

$$= -6726.3 \text{ N-m}$$

Moment per Equation 4.15.4 [M2]:

$$= \frac{Q \cdot L}{4} \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) - 4 \frac{a}{L}$$

$$= \frac{42 \cdot 5880}{4} \left(1 + 2 \frac{(297^2 - 0^2)}{(5880^2)} \right) / \left(1 + \frac{(4 \cdot 0)}{3 \cdot 5880} \right) - 4 \cdot \frac{992}{5880}$$

$$= 20428.3 \text{ N-m}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 \cdot t)$$

$$= 23.017 \cdot 297.0 / (2 \cdot 7.0) - 20428.3 / (\pi \cdot 297.0^2 \cdot 7.0)$$

$$= 38.30 \text{ N./mm}^2$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t)$$

$$= 23.017 \cdot 297.0 / (2 \cdot 7.0) + 20428.3 / (\pi \cdot 297.0^2 \cdot 7.0)$$

$$= 59.36 \text{ N./mm}^2$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t)$$

$$= 23.017 \cdot 297.0 / (2 \cdot 7.0) - 6726.3 / (0.1066 \cdot \pi \cdot 297.0^2 \cdot 7.0)$$

$$= 81.35 \text{ N./mm}^2$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t)$$

$$= 23.017 \cdot 297.0 / (2 \cdot 7.0) + 6726.3 / (0.1923 \cdot \pi \cdot 297.0^2 \cdot 7.0)$$

$$= 30.81 \text{ N./mm}^2$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$= \frac{Q(L - 2a)}{(L + (4 \cdot h^2 / 3))}$$

$$= \frac{42(5880.0 - 2 \cdot 992.0)}{(5880.0 + (4 \cdot 0.0 / 3))}$$

$$= 27.9 \text{ kN}$$

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 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Shear Stress in the shell no rings, not stiffened (4.15.14) [τ_{u2}]:

$$= K2 * T / (Rm * t)$$

$$= 1.1707 * 27.87 / (297.0 * 7.0)$$

$$= 15.69 \text{ N./mm}^2$$

Decay Length (4.15.22) [$x1, x2$]:

$$= 0.78 * \text{sqrt}(Rm * t)$$

$$= 0.78 * \text{sqrt}(297.0 * 7.0)$$

$$= 35.565 \text{ mm.}$$

Circumferential Stress in shell, no rings (4.15.23) [σ_{a6}]:

$$= -K5 * Q * k / (t * (b + X1 + X2))$$

$$= -0.7603 * 42 * 0.1 / (7.0 * (150.0 + 35.56 + 35.56))$$

$$= -2.07 \text{ N./mm}^2$$

Effective reinforcing plate width (4.15.1) [B1]:

$$= \text{min}(b + 1.56 * \text{sqrt}(Rm * t), 2a)$$

$$= \text{min}(150.0 + 1.56 * \text{sqrt}(297.0 * 7.0), 2 * 992.0)$$

$$= 221.13 \text{ mm.}$$

Wear Plate/Shell Stress ratio (4.15.29) [η]:

$$= \text{min}(Sr/S, 1)$$

$$= \text{min}(137.9/137.9, 1)$$

$$= 1.0000$$

Circumferential Stress at Saddle Base with Wear Plate (4.15.26) [$\sigma_{a6,r}$]:

$$= -K5 * Q * k / (B1(t + \eta * tr))$$

$$= -0.7603 * 42 * 0.1 / (221.13(7.0 + 1.0 * 10.0))$$

$$= -0.85 \text{ N./mm}^2$$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8Rm$ (4.15.27) [$\sigma_{a7,r}$]:

$$= -Q / (4(t + \eta * tr) b1) - 3 * K7 * Q / (2(t + \eta * tr)^2)$$

$$= -42 / (4(7.0 + 1.0 * 10.0) 221.13) -$$

$$3 * 0.053 * 42 / (2(7.0 + 1.0 * 10.0)^2)$$

$$= -14.34 \text{ N./mm}^2$$

Free Un-Restrained Thermal Expansion between the Saddles [Exp]:

$$= \text{Alpha} * Ls * (\text{Design Temperature} - \text{Ambient Temperature})$$

$$= 0.000012 * 4000.0 * (120.0 - 21.1)$$

$$= 4.838 \text{ mm.}$$

Results for Vessel Ribs, Web and Base:

Baseplate Length	Bplen	545.0000	mm.
Baseplate Thickness	Bpthk	16.0000	mm.
Baseplate Width	Bpwid	180.0000	mm.
Number of Ribs (inc. outside ribs)	Nribs	4	
Rib Thickness	Ribtk	10.0000	mm.
Web Thickness	Webtk	10.0000	mm.
Web Location	Webloc	Center	
Saddle Yield Stress	Sy	206.9	N./
Height of Web at Center	Hw,c	273.0	mm.
Friction Coefficient	mu	0.000	

Note: In the tables below I_o is I for the rectangle + Area * Centroid Distance²

Moment of Inertia of Saddle - Transverse Direction (90 degrees to long axis)

	B	D	Y	A	AY	I_o
Shell	295.7	7.0	3.5	20.7	7244.9	0.343E+04

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Wearplate	225.0	10.0	12.0	22.5	27000.0	0.326E+04
Web	10.0	273.5	153.7	27.3	420506.1	0.183E+04
BasePlate	180.0	16.0	298.5	28.8	859679.9	0.796E+04
Totals	99.3	1314430.9	0.165E+05

Distance to Centroid [C1]:

$$= AY / A$$

$$= 517.493/99.35$$

$$= 132.304 \text{ mm.}$$

Angle [beta]:

$$= 180 - \text{Saddle Angle}/2$$

$$= 180 - 120.0/2$$

$$= 120.0$$

Saddle Splitting Coefficient [K1]:

$$= (1 + \cos(\text{beta}) - 0.5 \cdot \sin(\text{beta})^2) / (\pi - \text{beta} + \sin(\text{beta}) \cos(\text{beta}))$$

$$= (1 + \cos(120.0) - 0.5 \cdot \sin(120.0)^2) / (\pi - 2.094 + \sin(120.0) \cos(120.0))$$

$$= 0.2035$$

Saddle Splitting Force [Fh]:

$$= K1 * Q$$

$$= 0.204 * 42.06$$

$$= 8.5601 \text{ kN}$$

$$\text{Tension Stress, } St = (Fh/As) = 1.0885 \text{ N./mm}^2$$

$$\text{Allowed Stress, } Sa = 0.6 * \text{Yield Str} = 124.1100 \text{ N./mm}^2$$

Saddle Splitting Dimension [d]:

$$= B - R * \sin(\text{theta}) / \text{theta}$$

$$= 600.0 - 293.5 * \sin(1.0472) / 1.0472$$

$$= 357.277 \text{ mm.}$$

$$\text{Bending Moment, } M = Fh * d = 3059.5596 \text{ N-m}$$

$$\text{Bending Stress, } Sb = (M * C1 / I) = 2.4547 \text{ N./mm}^2$$

$$\text{Allowed Stress, } Sa = 2/3 * \text{Yield Str} = 137.9000 \text{ N./mm}^2$$

Minimum Thickness of Baseplate per Moss:

$$= (3(Q + \text{Saddle_Wt}) \text{BasePlateWidth} / (4 * \text{BasePlateLength} * \text{AllStress}))^{1/2}$$

$$= (3(42 + 0.6)180.0 / (4 * 545.0 * 137.9))^{1/2}$$

$$= 8.754 \text{ mm.}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress:

Web Length Dimension [Web Length]:

$$= 2 * \cos(90 - \text{Saddle Angle}/2) (\text{Inside Radius} + \text{Shell Thk} + \text{Wear Plate Thk})$$

$$= 2 * \cos(90 - 120.0/2) (290.5 + 10.0 + 10.0)$$

$$= 537.802 \text{ mm.}$$

Distance between Ribs [e]:

$$= \text{Web Length} / (\text{Nr ribs} - 1)$$

$$= 537.8018 / (4 - 1)$$

$$= 179.267 \text{ mm.}$$

Baseplate Pressure Area [Ap]:

$$= e * \text{Bpwid} / 2$$

$$= 179.2673 * 180.0/2$$

$$= 161.341 \text{ cm}^2$$

Axial Load [P]:

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

$$\begin{aligned}
 &= A_p * B_p \\
 &= 161.3 * 0.04 \\
 &= 6.917 \text{ kN}
 \end{aligned}$$

Area of the Rib and Web [Ar]:

$$\begin{aligned}
 &= \text{Rib Area} + \text{Web Area} \\
 &= 14.0 + 8.963 \\
 &= 22.963 \text{ cm}^2
 \end{aligned}$$

Compressive Stress [Sc]:

$$\begin{aligned}
 &= P/Ar \\
 &= 6.9/22.9634 \\
 &= 3.013 \text{ N./mm}^2
 \end{aligned}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	B	D	Y	A	AY	Io
Rib+Web	10.0	150.0	...	15.0	...	281.

Rib dimension [D]:

$$\begin{aligned}
 &= \text{Saddle Width} - \text{Web Thickness} \\
 &= 150.0 - 10.0 \\
 &= 140.000 \text{ mm.}
 \end{aligned}$$

Distance to Centroid from Datum [ytot]:

$$\begin{aligned}
 &= AY / A \\
 &= 0.0/22.963 \\
 &= 0.000 \text{ mm.}
 \end{aligned}$$

Distance to Centroid [C1]:

$$\begin{aligned}
 &= \text{Saddle Width} / 2 \\
 &= 150.0/2 \\
 &= 75.000 \text{ mm.}
 \end{aligned}$$

Radius of Gyration [r]:

$$\begin{aligned}
 &= \sqrt{\text{Total Inertia} / \text{Total Area}} \\
 &= \sqrt{281.2/22.963} \\
 &= 34.997 \text{ mm.}
 \end{aligned}$$

Length of Outer Rib [L]:

$$\begin{aligned}
 &= \text{Saddle Height} - \cos(\text{theta}/2) (\text{radius} + \text{shlthk} + \text{wpdthk}) - \text{bpthk} \\
 &= 600.0 - \cos(120.0/2) (290.5 + 10.0 + 10.0) - 16.0 \\
 &= 428.750 \text{ mm.}
 \end{aligned}$$

Intermediate Term [Cc]:

$$\begin{aligned}
 &= \sqrt{2 * \pi^2 * \text{Elastic Modulus} / \text{Yield Stress}} \\
 &= \sqrt{2 * \pi^2 * 0.19994\text{E}+09/206.9} \\
 &= 138.135
 \end{aligned}$$

Slenderness ratio [KL/r]:

$$\begin{aligned}
 &= KL/r \\
 &= 1 * 428.75/34.997 \\
 &= 12.251
 \end{aligned}$$

Bending Moment [Rm]:

$$\begin{aligned}
 &= F_l / (2 * B_{plen}) * e * L / 2 \\
 &= 11.5 / (2 * 545.0) * 179.267 * 428.75/2 \\
 &= 405.204 \text{ N-m}
 \end{aligned}$$

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Compressive Allowable, $KL/r < Cc$ (12.2511 < 138.1347) per AISC E2-1 [Sca]:
 $= (1 - (Klr)^2 / (2 * Cc^2)) Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$
 $= (1 - (12.25)^2 / (2 * 138.13^2)) 207 /$
 $(5/3 + 3 * (12.25) / (8 * 138.13) - (12.25^3) / (8 * 138.13^3))$
 $= 121.2 \text{ N./mm}^2$

AISC Unity Check of Outside Ribs (must be <= 1)

$= Sc/Sca + (Rm * C1 / I) / Sba$
 $= 3.01/121.21 + (405.2 * 75.0/2812500) / 137.9$
 $= 0.103$

Check of Inside Ribs:

Inertia of Saddle, Inner Ribs - Axial Direction

	B	D	Y	A	AY	Io
Rib	10.0	140.0	0.0	14.0	0.0	281.
Web	179.3	10.0	0.0	17.9	0.0	1.49
Totals	31.9	...	283.

Distance to Centroid from Datum [ytot]:

$= AY / A$
 $= 0.0/31.927$
 $= 0.000 \text{ mm.}$

Distance to Centroid [C1]:

$= \text{Saddle Width} / 2$
 $= 150.0/2$
 $= 75.000 \text{ mm.}$

Length of Inner Rib [L]:

$= \text{Saddle Height} - \sqrt{ (Ro + Wpdthk)^2 - (Pitch/2)^2 } - Bpthk$
 $= 600.0 - \sqrt{ (310.5 + 10.0)^2 - (179.267/2)^2 } - 16.0$
 $= 286.719 \text{ mm.}$

Radius of Gyration [r]:

$= \sqrt{ \text{Total Inertia} / \text{Total Area} }$
 $= \sqrt{ 282.7/31.927 }$
 $= 29.755 \text{ mm.}$

Slenderness ratio [KL/r]:

$= KL/r$
 $= 1 * 286.719/29.755$
 $= 9.636$

Unit Force [Force,u]:

$= F1 / (2 * \text{Baseplate Length})$
 $= 11.488 / (2 * 545.0)$
 $= 0.011 \text{ kN/mm.}$

Moment at base of inner Rib [Mbase,c]:

$= \text{Unit Force} * e * L$
 $= 0.011 * 179.267 * 286.719$
 $= 541.946 \text{ N-m}$

Bending Stress due to Transverse Force and Weight Load [SigmaB,base,c]:

$= \text{Bending Moment} / \text{Section Modulus}$
 $= 541.946/37688.074$
 $= 14.375 \text{ N./mm}^2$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Compressive Allowable, $KL/r < Cc$ (9.6361 < 138.1347) per AISC E2-1 [Sca]:
 $= (1 - (Klr)^2 / (2 * Cc^2)) Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$
 $= (1 - (9.64)^2 / (2 * 138.13^2)) 207 /$
 $(5/3 + 3 * (9.64) / (8 * 138.13) - (9.64^3) / (8 * 138.13^3))$
 $= 121.9 \text{ N./mm}^2$

AISC Unity Check of Inside Ribs (must be ≤ 1)

$= Sc/Sca + (Mbase,c * C1/I) / Sba$
 $= 4.27/121.9 + (541.95 * 75.0/282.66) / 137.9$
 $= 0.139$

Input Data for Base Plate Bolting Calculations:

Total Number of Bolts per BasePlate	Nbolts	4	
Total Number of Bolts in Tension/Baseplate	Nbt	2	
Bolt Material Specification		SA-193 B7	
Bolt Allowable Stress	Stba	172.38	N./mm ²
Bolt Corrosion Allowance	Bca	0.0	mm.
Distance from Bolts to Edge	Edgedis	72.0	mm.
Nominal Bolt Diameter	Bnd	24.0000	mm.
Thread Series	Series	TEMA Metric	
BasePlate Allowable Stress	S	108.25	N./mm ²
Area Available in a Single Bolt	BltArea	3.1275	cm ²
Saddle Load QO (Weight)	QO	23.4	kN
Saddle Load QL (Wind/Seismic contribution)	QL	1.7	kN
Maximum Transverse Force	Ft	5.7	kN
Maximum Longitudinal Force	F1	11.5	kN
Saddle Bolted to Steel Foundation		Yes	

Shear Stress in a Single Bolt [taub]:

$= \text{Shear Force} / (2 * \text{Bolt Area} * \text{Number of Bolts})$
 $= 11 / (2 * 3.13 * 4)$
 $= 4.6 \text{ N./mm}^2$. Must be less than 103.4 N./mm².

Bolt Area Calculation per Dennis R. Moss

Bolt Area Requirement Due to Longitudinal Load [Bltarearl]:

$= 0.0$ (QO > QL --> No Uplift in Longitudinal direction)

Bolt Area due to Shear Load [Bltarears]:

$= F1 / (Stba * Nbolts)$
 $= 11.49 / (172.38 * 4.0)$
 $= 0.1666 \text{ cm}^2$

Bolt Area due to Transverse Load:

Moment on Baseplate Due to Transverse Load [Rmom]:

$= B * Ft + \text{Sum of X Moments}$
 $= 600.0 * 5.74 + 0.0$
 $= 3447.84 \text{ N-m}$

Eccentricity (e):

$= Rmom / QO$
 $= 3447.84 / 23.43$
 $= 147.10 \text{ mm.} > Bplen/6$ --> Uplift in Transverse direction

$f = Bplen / 2 - Edgedis$
 $= 545.0/2 - 72.0$
 $= 200.50 \text{ mm.}$

$K1 = 3 (e - 0.5 * Bplen)$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

$$= 3 (147.1 - 0.5 * 545.0)$$

$$= -376.21 \text{ mm.}$$

$$K2 = 6 * n1 * At / Bpwid * (f + e)$$

$$= 6 * 1.0 * 6.25 / 180.0 * (200.5 + 147.1)$$

$$= 7247.36 \text{ mm.}^2$$

$$K3 = -K2 * (0.5 * Bplen + f)$$

$$= -7247.36 * (0.5 * 545.0 + 200.5)$$

$$= -3427999.84 \text{ mm.}^3$$

Iteratively Solving for the Effective Bearing Length:

$$Y^3 + K1 * Y^2 + K2 * Y + K3 = 0$$

$$Y^3 + -376.21 * Y^2 + 7247.36 * Y + -0.3E+07 = 0$$

$$Y = 380.81 \text{ mm.}$$

$$\text{Num} = (Bplen / 2 - Y / 3 - e)$$

$$= (545.0 / 2 - 380.81 / 3 - 147.1)$$

$$= -1.54$$

$$\text{Denom} = (Bplen / 2 - Y / 3 + f)$$

$$= (545.0 / 2 - 380.81 / 3 + 200.5)$$

$$= 346.06$$

Total Bolt Tension Force [Tforce]:

$$= -QO * \text{Num} / \text{Denom}$$

$$= -23.43 * -1.54 / 346.06$$

$$= 0.10 \text{ kN}$$

Bolt Area Required due to Transverse Load [Bltareart]:

$$= \text{Tforce} / (\text{Stba} * \text{Nbt})$$

$$= 0.1 / (172.38 * 2.0)$$

$$= 0.0030 \text{ cm}^2$$

Required Area of a Single Bolt [Bltarear]:

$$= \max[\text{Bltarearl}, \text{Bltarears}, \text{Bltareart}]$$

$$= \max[0.0, 0.1666, 0.003]$$

$$= 0.1666 \text{ cm}^2$$

Baseplate Thickness Calculation per D. Moss:

Bearing Pressure (fc)

$$= 2(QO + \text{Tforce}) / (Y * \text{Bpwid})$$

$$= 2(23.43 + 0.1) / (380.81 * 180.0)$$

$$= 6.87 \text{ bars}$$

Distance from Baseplate Edge to the Web [ADIST]:

$$= (Bplen - \text{Weblngth}) / 2$$

$$= (545.0 - 494.2) / 2$$

$$= 25.4000 \text{ mm.}$$

Overturing Moment due To Bolt Tension [Mt]:

$$= \text{Tforce} * \text{Adist}$$

$$= 0.1 * 25.4$$

$$= 2.64 \text{ N-m}$$

Equivalent Bearing Pressure (f1):

$$= fc * (Y - \text{Adist}) / Y$$

$$= 6.87 * (380.81 - 25.4) / 380.81$$

$$= 6.41 \text{ bars}$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Overturing Moment due to Bearing Pressure [Mc]:

$$= (Adist^2 * Bpwid / 6) * (f1 + 2 * fc)$$

$$= (25.4^2 * 180.0/6) * (6.41 + 2 * 6.87)$$

$$= 39.00 \text{ N-m}$$

Baseplate Required Thickness [Treq]:

$$= (6 * \max(Mt, Mc) / (Bpwid * Sba))^{1/2}$$

$$= (6 * \max(2.64, 39.0/(180.0 * 162.38))^{1/2}$$

$$= 2.8290 \text{ mm.}$$

ASME Horizontal Vessel Analysis: Stresses for the Right Saddle (per ASME Sec. VIII Div. 2 based on the Zick method.)

Input and Calculated Values:

Vessel Mean Radius	Rm	297.00	mm.
Stiffened Vessel Length per 4.15.6	L	5880.00	mm.
Distance from Saddle to Vessel tangent	a	992.00	mm.
Saddle Width	b	150.00	mm.
Saddle Bearing Angle	theta	120.00	degrees
Wear Plate Width	b1	225.00	mm.
Wear Plate Bearing Angle	thetal	132.00	degrees
Wear Plate Thickness	tr	10.0	mm.
Wear Plate Allowable Stress	Sr	137.90	N./mm ²
Shell Allowable Stress used in Calculation		137.90	N./mm ²
Head Allowable Stress used in Calculation		137.90	N./mm ²
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		42.24	kN
Horizontal Vessel Analysis Results:	Actual	Allowable	
	N./mm ²	N./mm ²	

Long. Stress at Top of Midspan	38.26	137.90	
Long. Stress at Bottom of Midspan	59.40	137.90	
Long. Stress at Top of Saddles	81.48	137.90	
Long. Stress at Bottom of Saddles	30.73	137.90	

Tangential Shear in Shell	15.76	110.32	
Circ. Stress at Horn of Saddle	14.40	172.37	
Circ. Compressive Stress in Shell	2.07	137.90	

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= Ftr * (Ft/Num of Saddles + Z Force Load) * B / E$$

$$= 3.0 * (6.1/2 + 0) * 600.0/537.8018$$

$$= 10.1 \text{ kN}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= \max(Fl, Friction Load, Sum of X Forces) * B / Ls$$

$$= \max(0.59, 9.44, 0) * 600.0/4000.0$$

$$= 1.4 \text{ kN}$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Saddle Reaction Force due to Earthquake Fl or Friction [Fsl]:
 = max(Fl, Friction Force, Sum of X Forces) * B / Ls
 = max(11.49, 9.44, 0) * 600.0/4000.0
 = 1.7 kN

Saddle Reaction Force due to Earthquake Ft [Fst]:
 = Ftr * (Ft/Num of Saddles + Z Force Load) * B / E
 = 3.0 * (11/2 + 0) * 600.0/537.8018
 = 19.2 kN

Load Combination Results for Q + Wind or Seismic [Q]:
 = Saddle Load + Max(Fwl, Fwt, Fsl, Fst)
 = 23 + Max(1, 10, 2, 19)
 = 42.2 kN

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	42.83	kN
Transverse Shear Load Saddle	5.74	kN
Longitudinal Shear Load Saddle	11.49	kN

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, k = 0.1

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0529	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0434
K7p = 0.0434			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to Rm / 2.

Moment per Equation 4.15.3 [M1]:
 = $-Q*a [1 - (1 - a/L + (R^2 - h^2)/(2a*L))/(1 + (4h^2)/3L)]$
 = $-42*992.0 [1 - (1 - 992.0/5880.0 + (297.0^2 - 0.0^2)/(2*992.0*5880.0))/(1 + (4*0.0)/(3*5880.0))]$
 = -6754.9 N-m

Moment per Equation 4.15.4 [M2]:
 = $Q*L/4(1 + 2(R^2 - h^2)/(L^2))/(1 + (4h^2)/(3L)) - 4a/L$
 = $42*5880/4(1 + 2(297^2 - 0^2)/(5880^2))/(1 + (4*0)/(3*5880)) - 4*992/5880$
 = 20515.2 N-m

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:
 = $P * Rm/(2t) - M2/(pi*Rm^2*t)$
 = $23.017 * 297.0/(2*7.0) - 20515.2/(pi*297.0^2*7.0)$
 = 38.26 N./mm²

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:
 = $P * Rm/(2t) + M2/(pi * Rm^2 * t)$
 = $23.017 * 297.0/(2 * 7.0) + 20515.2/(pi * 297.0^2 * 7.0)$
 = 59.40 N./mm²

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:
 = $P * Rm/(2t) - M1/(K1*pi*Rm^2*t)$
 = $23.017*297.0/(2*7.0) - 6754.9/(0.1066*pi*297.0^2*7.0)$
 = 81.48 N./mm²

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [σ^*4]:
 $= P * Rm / (2t) + M1 / (K1 * \pi * Rm^2 * t)$
 $= 23.017 * 297.0 / (2 * 7.0) + -6754.9 / (0.1923 * \pi * 297.0^2 * 7.0)$
 $= 30.73 \text{ N./mm}^2$

Maximum Shear Force in the Saddle (4.15.5) [T]:
 $= Q(L-2a) / (L + (4 * h^2 / 3))$
 $= 42(5880.0 - 2 * 992.0) / (5880.0 + (4 * 0.0 / 3))$
 $= 28.0 \text{ kN}$

Shear Stress in the shell no rings, not stiffened (4.15.14) [τ^*2]:
 $= K2 * T / (Rm * t)$
 $= 1.1707 * 27.99 / (297.0 * 7.0)$
 $= 15.76 \text{ N./mm}^2$

Decay Length (4.15.22) [$x1, x2$]:
 $= 0.78 * \text{sqrt}(Rm * t)$
 $= 0.78 * \text{sqrt}(297.0 * 7.0)$
 $= 35.565 \text{ mm.}$

Circumferential Stress in shell, no rings (4.15.23) [σ^*6]:
 $= -K5 * Q * k / (t * (b + X1 + X2))$
 $= -0.7603 * 42 * 0.1 / (7.0 * (150.0 + 35.56 + 35.56))$
 $= -2.07 \text{ N./mm}^2$

Effective reinforcing plate width (4.15.1) [B1]:
 $= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a)$
 $= \min(150.0 + 1.56 * \text{sqrt}(297.0 * 7.0), 2 * 992.0)$
 $= 221.13 \text{ mm.}$

Wear Plate/Shell Stress ratio (4.15.29) [η]:
 $= \min(Sr/S, 1)$
 $= \min(137.9/137.9, 1)$
 $= 1.0000$

Circumferential Stress at Saddle Base with Wear Plate (4.15.26) [σ^*6, r]:
 $= -K5 * Q * k / (B1(t + \eta * tr))$
 $= -0.7603 * 42 * 0.1 / (221.13(7.0 + 1.0 * 10.0))$
 $= -0.85 \text{ N./mm}^2$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8Rm$ (4.15.27) [σ^*7, r]:
 $= -Q / (4(t + \eta * tr)b1) - 3 * K7 * Q / (2(t + \eta * tr)^2)$
 $= -42 / (4(7.0 + 1.0 * 10.0)221.13) -$
 $3 * 0.053 * 42 / (2(7.0 + 1.0 * 10.0)^2)$
 $= -14.40 \text{ N./mm}^2$

Results for Vessel Ribs, Web and Base:

Baseplate Length	Bplen	545.0000	mm.
Baseplate Thickness	Bpthk	16.0000	mm.
Baseplate Width	Bpwid	180.0000	mm.
Number of Ribs (inc. outside ribs)	Nribs	4	
Rib Thickness	Ribtk	10.0000	mm.
Web Thickness	Webtk	10.0000	mm.
Web Location	Webloc	Center	
Saddle Yield Stress	Sy	206.9	N. /
Height of Web at Center	Hw,c	273.0	mm.
Friction Coefficient	mu	0.400	

Note: In the tables below I_o is I for the rectangle + Area * Centroid Distance²

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Moment of Inertia of Saddle - Transverse Direction (90 degrees to long axis)

	B	D	Y	A	AY	Io
Shell	295.7	7.0	3.5	20.7	7244.9	0.343E+04
Wearplate	225.0	10.0	12.0	22.5	27000.0	0.326E+04
Web	10.0	273.5	153.7	27.3	420506.1	0.183E+04
BasePlate	180.0	16.0	298.5	28.8	859679.9	0.796E+04
Totals	99.3	1314430.9	0.165E+05

Distance to Centroid [C1]:

$$= AY / A$$

$$= 517.493/99.35$$

$$= 132.304 \text{ mm.}$$

Angle [beta]:

$$= 180 - \text{Saddle Angle}/2$$

$$= 180 - 120.0/2$$

$$= 120.0$$

Saddle Splitting Coefficient [K1]:

$$= (1 + \cos(\beta) - 0.5 \cdot \sin(\beta)^2) / (\pi - \beta + \sin(\beta)\cos(\beta))$$

$$= (1 + \cos(120.0) - 0.5 \cdot \sin(120.0)^2) / (\pi - 2.094 + \sin(120.0)\cos(120.0))$$

$$= 0.2035$$

Saddle Splitting Force [Fh]:

$$= K1 \cdot Q$$

$$= 0.204 \cdot 42.239$$

$$= 8.5965 \text{ kN}$$

$$\text{Tension Stress, } St = (Fh/As) = 1.0931 \text{ N./mm}^2$$

$$\text{Allowed Stress, } Sa = 0.6 \cdot \text{Yield Str} = 124.1100 \text{ N./mm}^2$$

Saddle Splitting Dimension [d]:

$$= B - R \cdot \sin(\theta) / \theta$$

$$= 600.0 - 293.5 \cdot \sin(1.0472) / 1.0472$$

$$= 357.277 \text{ mm.}$$

$$\text{Bending Moment, } M = Fh \cdot d = 3072.5747 \text{ N-m}$$

$$\text{Bending Stress, } Sb = (M \cdot C1 / I) = 2.4652 \text{ N./mm}^2$$

$$\text{Allowed Stress, } Sa = 2/3 \cdot \text{Yield Str} = 137.9000 \text{ N./mm}^2$$

Minimum Thickness of Baseplate per Moss:

$$= (3(Q + \text{Saddle_Wt}) \text{BasePlateWidth} / (4 \cdot \text{BasePlateLength} \cdot \text{AllStress}))^{1/2}$$

$$= (3(42 + 0.6)180.0 / (4 \cdot 545.0 \cdot 137.9))^{1/2}$$

$$= 8.772 \text{ mm.}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress:

Web Length Dimension [Web Length]:

$$= 2 \cdot \cos(90 - \text{Saddle Angle}/2) (\text{Inside Radius} + \text{Shell Thk} + \text{Wear Plate Thk})$$

$$= 2 \cdot \cos(90 - 120.0/2) (290.5 + 10.0 + 10.0)$$

$$= 537.802 \text{ mm.}$$

Distance between Ribs [e]:

$$= \text{Web Length} / (\text{Nr ribs} - 1)$$

$$= 537.8018 / (4 - 1)$$

$$= 179.267 \text{ mm.}$$

Baseplate Pressure Area [Ap]:

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

$$\begin{aligned}
 &= e * Bpwid / 2 \\
 &= 179.2673 * 180.0 / 2 \\
 &= 161.341 \text{ cm}^2
 \end{aligned}$$

Axial Load [P]:

$$\begin{aligned}
 &= Ap * Bp \\
 &= 161.3 * 0.04 \\
 &= 6.947 \text{ kN}
 \end{aligned}$$

Area of the Rib and Web [Ar]:

$$\begin{aligned}
 &= \text{Rib Area} + \text{Web Area} \\
 &= 14.0 + 8.963 \\
 &= 22.963 \text{ cm}^2
 \end{aligned}$$

Compressive Stress [Sc]:

$$\begin{aligned}
 &= P / Ar \\
 &= 6.9 / 22.9634 \\
 &= 3.025 \text{ N./mm}^2
 \end{aligned}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	B	D	Y	A	AY	Io
Rib+Web	10.0	150.0	...	15.0	...	281.

Rib dimension [D]:

$$\begin{aligned}
 &= \text{Saddle Width} - \text{Web Thickness} \\
 &= 150.0 - 10.0 \\
 &= 140.000 \text{ mm.}
 \end{aligned}$$

Distance to Centroid from Datum [ytot]:

$$\begin{aligned}
 &= AY / A \\
 &= 0.0 / 22.963 \\
 &= 0.000 \text{ mm.}
 \end{aligned}$$

Distance to Centroid [C1]:

$$\begin{aligned}
 &= \text{Saddle Width} / 2 \\
 &= 150.0 / 2 \\
 &= 75.000 \text{ mm.}
 \end{aligned}$$

Radius of Gyration [r]:

$$\begin{aligned}
 &= \sqrt{\text{Total Inertia} / \text{Total Area}} \\
 &= \sqrt{281.2 / 22.963} \\
 &= 34.997 \text{ mm.}
 \end{aligned}$$

Length of Outer Rib [L]:

$$\begin{aligned}
 &= \text{Saddle Height} - \cos(\theta/2) (\text{radius} + \text{shlthk} + \text{wpdthk}) - \text{bpthk} \\
 &= 600.0 - \cos(120.0/2) (290.5 + 10.0 + 10.0) - 16.0 \\
 &= 428.750 \text{ mm.}
 \end{aligned}$$

Intermediate Term [Cc]:

$$\begin{aligned}
 &= \sqrt{2 * \pi^2 * \text{Elastic Modulus} / \text{Yield Stress}} \\
 &= \sqrt{2 * \pi^2 * 0.19994\text{E}+09 / 206.9} \\
 &= 138.135
 \end{aligned}$$

Slenderness ratio [KL/r]:

$$\begin{aligned}
 &= KL / r \\
 &= 1 * 428.75 / 34.997 \\
 &= 12.251
 \end{aligned}$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Bending Moment [Rm]:

$$= F1 / (2 * Bplen) * e * L / 2$$

$$= 11.5 / (2 * 545.0) * 179.267 * 428.75 / 2$$

$$= 405.204 \text{ N-m}$$

Compressive Allowable, $KL/r < Cc$ (12.2511 < 138.1347) per AISC E2-1 [Sca]:

$$= (1 - (KL/r)^2 / (2 * Cc^2)) Fy / (5/3 + 3 * (KL/r) / (8 * Cc) - (KL/r)^3 / (8 * Cc^3))$$

$$= (1 - (12.25)^2 / (2 * 138.13^2)) 207 / (5/3 + 3 * (12.25) / (8 * 138.13) - (12.25^3) / (8 * 138.13^3))$$

$$= 121.2 \text{ N./mm}^2$$

AISC Unity Check of Outside Ribs (must be ≤ 1)

$$= Sc / Sca + (Rm * C1 / I) / Sba$$

$$= 3.03 / 121.21 + (405.2 * 75.0 / 2812500) / 137.9$$

$$= 0.103$$

Check of Inside Ribs:

Inertia of Saddle, Inner Ribs - Axial Direction

	B	D	Y	A	AY	Io
Rib	10.0	140.0	0.0	14.0	0.0	281.
Web	179.3	10.0	0.0	17.9	0.0	1.49
Totals	31.9	...	283.

Distance to Centroid from Datum [ytot]:

$$= AY / A$$

$$= 0.0 / 31.927$$

$$= 0.000 \text{ mm.}$$

Distance to Centroid [C1]:

$$= \text{Saddle Width} / 2$$

$$= 150.0 / 2$$

$$= 75.000 \text{ mm.}$$

Length of Inner Rib [L]:

$$= \text{Saddle Height} - \sqrt{ (Ro + Wpdthk)^2 - (Pitch/2)^2 } - Bpthk$$

$$= 600.0 - \sqrt{ (310.5 + 10.0)^2 - (179.267/2)^2 } - 16.0$$

$$= 286.719 \text{ mm.}$$

Radius of Gyration [r]:

$$= \sqrt{ \text{Total Inertia} / \text{Total Area} }$$

$$= \sqrt{ 282.7 / 31.927 }$$

$$= 29.755 \text{ mm.}$$

Slenderness ratio [KL/r]:

$$= KL/r$$

$$= 1 * 286.719 / 29.755$$

$$= 9.636$$

Unit Force [Force,u]:

$$= F1 / (2 * \text{Baseplate Length})$$

$$= 11.488 / (2 * 545.0)$$

$$= 0.011 \text{ kN/mm.}$$

Moment at base of inner Rib [Mbase,c]:

$$= \text{Unit Force} * e * L$$

$$= 0.011 * 179.267 * 286.719$$

$$= 541.946 \text{ N-m}$$

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Bending Stress due to Transverse Force and Weight Load [SigmaB,base,c]:

$$= \text{Bending Moment} / \text{Section Modulus}$$

$$= 541.946/37688.074$$

$$= 14.375 \text{ N./mm}^2$$

Compressive Allowable, $KL/r < Cc$ ($9.6361 < 138.1347$) per AISC E2-1 [Sca]:

$$= (1 - (Klr)^2 / (2 * Cc^2)) Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$= (1 - (9.64)^2 / (2 * 138.13^2)) 207 /$$

$$(5/3 + 3 * (9.64) / (8 * 138.13) - (9.64^3) / (8 * 138.13^3))$$

$$= 121.9 \text{ N./mm}^2$$

AISC Unity Check of Inside Ribs (must be ≤ 1)

$$= Sc/Sca + (Mbase,c * C1/I) / Sba$$

$$= 4.29/121.9 + (541.95 * 75.0/282.66) / 137.9$$

$$= 0.139$$

Input Data for Base Plate Bolting Calculations:

Total Number of Bolts per BasePlate	Nbolts	4	
Total Number of Bolts in Tension/Baseplate	Nbt	2	
Bolt Material Specification		SA-193 B7	
Bolt Allowable Stress	Stba	172.38	N./mm ²
Bolt Corrosion Allowance	Bca	0.0	mm.
Distance from Bolts to Edge	Edgedis	72.0	mm.
Nominal Bolt Diameter	Bnd	24.0000	mm.
Thread Series	Series	TEMA Metric	
BasePlate Allowable Stress	S	108.25	N./mm ²
Area Available in a Single Bolt	BltArea	3.1275	cm ²
Saddle Load QO (Weight)	QO	23.6	kN
Saddle Load QL (Wind/Seismic contribution)	QL	1.7	kN
Maximum Transverse Force	Ft	5.7	kN
Maximum Longitudinal Force	F1	11.5	kN
Saddle Bolted to Steel Foundation		Yes	

Shear Stress in a Single Bolt [taub]:

$$= \text{Shear Force} / (2 * \text{Bolt Area} * \text{Number of Bolts})$$

$$= 11 / (2 * 3.13 * 4)$$

$$= 4.6 \text{ N./mm}^2. \text{ Must be less than } 103.4 \text{ N./mm}^2.$$

Bolt Area Calculation per Dennis R. Moss

Bolt Area Requirement Due to Longitudinal Load [Bltarearl]:

$$= 0.0 \text{ (} QO > QL \text{ --> No Uplift in Longitudinal direction)}$$

Bolt Area due to Shear Load [Bltarears]:

$$= F1 / (Stba * Nbolts)$$

$$= 11.49 / (172.38 * 4.0)$$

$$= 0.1666 \text{ cm}^2$$

Bolt Area due to Transverse Load:

Moment on Baseplate Due to Transverse Load [Rmom]:

$$= B * Ft + \text{Sum of X Moments}$$

$$= 600.0 * 5.74 + 0.0$$

$$= 3447.84 \text{ N-m}$$

Eccentricity (e):

$$= Rmom / QO$$

$$= 3447.84 / 23.61$$

$$= 145.98 \text{ mm.} > Bplen/6 \text{ --> Uplift in Transverse direction}$$

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

$$\begin{aligned} f &= B_{pln} / 2 - E_{gedis} \\ &= 545.0/2 - 72.0 \\ &= 200.50 \text{ mm.} \end{aligned}$$

$$\begin{aligned} K1 &= 3 (e - 0.5 * B_{pln}) \\ &= 3 (145.98 - 0.5*545.0) \\ &= -379.55 \text{ mm.} \end{aligned}$$

$$\begin{aligned} K2 &= 6 * n1 * A_t / B_{pwid} * (f + e) \\ &= 6 * 1.0 * 6.25/180.0 * (200.5 + 145.98) \\ &= 7224.12 \text{ mm.}^2 \end{aligned}$$

$$\begin{aligned} K3 &= -K2 * (0.5 * B_{pln} + f) \\ &= -7224.12 * (0.5 * 545.0 + 200.5) \\ &= -3417006.12 \text{ mm.}^3 \end{aligned}$$

Iteratively Solving for the Effective Bearing Length:

$$\begin{aligned} Y^3 + K1 * Y^2 + K2 * Y + K3 &= 0 \\ Y^3 + -379.55 * Y^2 + 7224.12 * Y + -0.3E+07 &= 0 \\ Y &= 383.92 \text{ mm.} \end{aligned}$$

$$\begin{aligned} Num &= (B_{pln} / 2 - Y / 3 - e) \\ &= (545.0/2 - 383.92/3 - 145.98) \\ &= -1.46 \end{aligned}$$

$$\begin{aligned} Denom &= (B_{pln} / 2 - Y / 3 + f) \\ &= (545.0/2 - 383.92/3 + 200.5) \\ &= 345.03 \end{aligned}$$

Total Bolt Tension Force [Tforce]:

$$\begin{aligned} &= -QO * Num / Denom \\ &= -23.61 * -1.46/345.03 \\ &= 0.10 \text{ kN} \end{aligned}$$

Bolt Area Required due to Transverse Load [Bltareart]:

$$\begin{aligned} &= Tforce / (Stba * Nbt) \\ &= 0.1/(172.38 * 2.0) \\ &= 0.0029 \text{ cm}^2 \end{aligned}$$

Required Area of a Single Bolt [Bltarear]:

$$\begin{aligned} &= \max[Bltarearl, Bltarears, Bltareart] \\ &= \max[0.0, 0.1666, 0.0029] \\ &= 0.1666 \text{ cm}^2 \end{aligned}$$

Baseplate Thickness Calculation per D. Moss:

Bearing Pressure (fc)

$$\begin{aligned} &= 2(QO + Tforce) / (Y * B_{pwid}) \\ &= 2(23.61 + 0.1)/(383.92 * 180.0) \\ &= 6.86 \text{ bars} \end{aligned}$$

Distance from Baseplate Edge to the Web [ADIST]:

$$\begin{aligned} &= (B_{pln} - Weblngth) / 2 \\ &= (545.0 - 494.2)/2 \\ &= 25.4000 \text{ mm.} \end{aligned}$$

Overturning Moment due To Bolt Tension [Mt]:

$$\begin{aligned} &= Tforce * Adist \\ &= 0.1 * 25.4 \\ &= 2.53 \text{ N-m} \end{aligned}$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Ope.): Step: 12 11:46pm Dec 22,2021

Equivalent Bearing Pressure (f1):

$$\begin{aligned}
 &= f_c * (Y - Adist) / Y \\
 &= 6.86 * (383.92 - 25.4) / 383.92 \\
 &= 6.41 \text{ bars}
 \end{aligned}$$

Overturing Moment due to Bearing Pressure [Mc]:

$$\begin{aligned}
 &= (Adist^2 * Bpwid / 6) * (f1 + 2 * f_c) \\
 &= (25.4^2 * 180.0 / 6) * (6.41 + 2 * 6.86) \\
 &= 38.98 \text{ N-m}
 \end{aligned}$$

Baseplate Required Thickness [Treq]:

$$\begin{aligned}
 &= (6 * \max(Mt, Mc) / (Bpwid * Sba))^{1/2} \\
 &= (6 * \max(2.53, 38.98 / (180.0 * 162.38))^{1/2} \\
 &= 2.8282 \text{ mm.}
 \end{aligned}$$

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

ASME Horizontal Vessel Analysis: Stresses for the Left Saddle
 (per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Test Case

Input and Calculated Values:

Vessel Mean Radius	Rm	297.00	mm.
Stiffened Vessel Length per 4.15.6	L	5880.00	mm.
Distance from Saddle to Vessel tangent	a	992.00	mm.
Saddle Width	b	150.00	mm.
Saddle Bearing Angle	theta	120.00	degrees
Wear Plate Width	b1	225.00	mm.
Wear Plate Bearing Angle	thetal	132.00	degrees
Wear Plate Thickness	tr	10.0	mm.
Wear Plate Allowable Stress	Sr	137.90	N./mm^2
Shell Allowable Stress used in Calculation		235.81	N./mm^2
Head Allowable Stress used in Calculation		235.81	N./mm^2
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Test Case, no Ext. Forces		30.12	kN
Horizontal Vessel Analysis Results:	Actual	Allowable	
	N./mm^2	N./mm^2	

Long. Stress at Top of Midspan	55.96	235.81	
Long. Stress at Bottom of Midspan	71.04	235.81	
Long. Stress at Top of Saddles	86.78	235.81	
Long. Stress at Bottom of Saddles	50.59	235.81	

Tangential Shear in Shell	11.24	188.65	
Circ. Stress at Horn of Saddle	10.27	353.71	
Circ. Compressive Stress in Shell	1.48	235.81	

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E$$

$$= 3.0 * (2.0/2 + 0) * 600.0/537.8018$$

$$= 3.3 \text{ kN}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= \max(F_l, \text{Friction Load, Sum of X Forces}) * B / L_s$$

$$= \max(0.2, 0.0, 0) * 600.0/4000.0$$

$$= 0.0 \text{ kN}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st})$$

$$= 27 + \max(0.0, 3, 0, 0)$$

$$= 30.1 \text{ kN}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)		30.71	kN
Transverse Shear Load Saddle	Ft	1.00	kN
Longitudinal Shear Load Saddle		0.20	kN

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Hydrostatic Test Pressure at center of Vessel: 29.929 bars

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

K1 = 0.1066 K2 = 1.1707 K3 = 0.8799 K4 = 0.4011
 K5 = 0.7603 K6 = 0.0529 K7 = 0.0529 K8 = 0.3405
 K9 = 0.2711 K10 = 0.0581 K1* = 0.1923 K6p = 0.0434
 K7p = 0.0434

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right]$$

$$= -30 \cdot 992.0 \left[1 - \left(1 - \frac{992.0}{5880.0} + \frac{(297.0^2 - 0.0^2)}{(2 \cdot 992.0 \cdot 5880.0)} \right) / \left(1 + \frac{(4 \cdot 0.0^2)}{3 \cdot 5880.0} \right) \right]$$

$$= -4816.7 \text{ N-m}$$

Moment per Equation 4.15.4 [M2]:

$$= \frac{Q \cdot L}{4} \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) - 4a/L$$

$$= \frac{30 \cdot 5880.0}{4} \left(1 + 2 \frac{(297.0^2 - 0.0^2)}{(5880.0^2)} \right) / \left(1 + \frac{(4 \cdot 0.0^2)}{3 \cdot 5880.0} \right) - 4 \cdot 992.0 / 5880.0$$

$$= 14628.8 \text{ N-m}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 \cdot t)$$

$$= 29.929 \cdot 297.0 / (2 \cdot 7.0) - 14628.8 / (\pi \cdot 297.0^2 \cdot 7.0)$$

$$= 55.96 \text{ N./mm}^2$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t)$$

$$= 29.929 \cdot 297.0 / (2 \cdot 7.0) + 14628.8 / (\pi \cdot 297.0^2 \cdot 7.0)$$

$$= 71.04 \text{ N./mm}^2$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t)$$

$$= 29.929 \cdot 297.0 / (2 \cdot 7.0) - 4816.7 / (0.1066 \cdot \pi \cdot 297.0^2 \cdot 7.0)$$

$$= 86.78 \text{ N./mm}^2$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t)$$

$$= 29.929 \cdot 297.0 / (2 \cdot 7.0) + 4816.7 / (0.1923 \cdot \pi \cdot 297.0^2 \cdot 7.0)$$

$$= 50.59 \text{ N./mm}^2$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$= \frac{Q(L - 2a)}{(L + (4h^2/3))}$$

$$= \frac{30(5880.0 - 2 \cdot 992.0)}{(5880.0 + (4 \cdot 0.0^2/3))}$$

$$= 20.0 \text{ kN}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$= \frac{K2 \cdot T}{(R_m \cdot t)}$$

$$= \frac{1.1707 \cdot 19.96}{(297.0 \cdot 7.0)}$$

$$= 11.24 \text{ N./mm}^2$$

Decay Length (4.15.22) [x1,x2]:

$$= 0.78 \cdot \sqrt{R_m \cdot t}$$

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

$$= 0.78 * \text{sqrt}(297.0 * 7.0)$$

$$= 35.565 \text{ mm.}$$

Circumferential Stress in shell, no rings (4.15.23) [σ_6]:

$$= -K5 * Q * k / (t * (b + X1 + X2))$$

$$= - 0.7603 * 30 * 0.1 / (7.0 * (150.0 + 35.56 + 35.56))$$

$$= -1.48 \text{ N./mm}^2$$

Effective reinforcing plate width (4.15.1) [B1]:

$$= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a)$$

$$= \min(150.0 + 1.56 * \text{sqrt}(297.0 * 7.0), 2 * 992.0)$$

$$= 221.13 \text{ mm.}$$

Wear Plate/Shell Stress ratio (4.15.29) [η]:

$$= 1.0000 \text{ Materials are the same, test case}$$

Circumferential Stress at Saddle Base with Wear Plate (4.15.26) [$\sigma_{6,r}$]:

$$= -K5 * Q * k / (B1(t + \eta * tr))$$

$$= - 0.7603 * 30 * 0.1 / (221.13(7.0 + 1.0 * 10.0))$$

$$= -0.61 \text{ N./mm}^2$$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8Rm$ (4.15.27) [$\sigma_{7,r}$]:

$$= -Q / (4(t + \eta * tr) b1) - 3 * K7 * Q / (2(t + \eta * tr)^2)$$

$$= -30 / (4(7.0 + 1.0 * 10.0) 221.13) -$$

$$3 * 0.053 * 30 / (2(7.0 + 1.0 * 10.0)^2)$$

$$= -10.27 \text{ N./mm}^2$$

Results for Vessel Ribs, Web and Base:

Baseplate Length	Bplen	545.0000	mm.
Baseplate Thickness	Bpthk	16.0000	mm.
Baseplate Width	Bpwid	180.0000	mm.
Number of Ribs (inc. outside ribs)	Nribs	4	
Rib Thickness	Ribtk	10.0000	mm.
Web Thickness	Webtk	10.0000	mm.
Web Location	Webloc	Center	
Saddle Yield Stress	Sy	206.9	N. /
Height of Web at Center	Hw,c	273.0	mm.
Friction Coefficient	mu	0.000	

Note: In the tables below I_o is I for the rectangle + Area * Centroid Distance²

Moment of Inertia of Saddle - Transverse Direction (90 degrees to long axis)

	B	D	Y	A	AY	I_o
Shell	295.7	7.0	3.5	20.7	7244.9	0.343E+04
Wearplate	225.0	10.0	12.0	22.5	27000.0	0.326E+04
Web	10.0	273.5	153.7	27.3	420506.1	0.183E+04
BasePlate	180.0	16.0	298.5	28.8	859679.9	0.796E+04
Totals	99.3	1314430.9	0.165E+05

Distance to Centroid [C1]:

$$= AY / A$$

$$= 517.493 / 99.35$$

$$= 132.304 \text{ mm.}$$

Angle [beta]:

$$= 180 - \text{Saddle Angle} / 2$$

$$= 180 - 120.0 / 2$$

$$= 120.0$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Saddle Splitting Coefficient [K1]:

$$= (1 + \cos(\beta) - 0.5 \cdot \sin(\beta)^2) / (\pi - \beta + \sin(\beta) \cos(\beta))$$

$$= (1 + \cos(120.0) - 0.5 \cdot \sin(120.0)^2) / (\pi - 2.094 + \sin(120.0) \cos(120.0))$$

$$= 0.2035$$

Saddle Splitting Force [Fh]:

$$= K1 * Q$$

$$= 0.204 * 30.119$$

$$= 6.1299 \text{ kN}$$

$$\text{Tension Stress, } St = (Fh / As) = 0.7795 \text{ N./mm}^2$$

$$\text{Allowed Stress, } Sa = 0.6 * \text{Yield Str} = 124.1100 \text{ N./mm}^2$$

Saddle Splitting Dimension [d]:

$$= B - R * \sin(\theta) / \theta$$

$$= 600.0 - 293.5 * \sin(1.0472) / 1.0472$$

$$= 357.277 \text{ mm.}$$

$$\text{Bending Moment, } M = Fh * d = 2190.9668 \text{ N-m}$$

$$\text{Bending Stress, } Sb = (M * Cl / I) = 1.7578 \text{ N./mm}^2$$

$$\text{Allowed Stress, } Sa = 2/3 * \text{Yield Str} = 137.9000 \text{ N./mm}^2$$

Minimum Thickness of Baseplate per Moss:

$$= (3(Q + \text{Saddle_Wt}) \text{BasePlateWidth} / (4 * \text{BasePlateLength} * \text{AllStress}))^{1/2}$$

$$= (3(30 + 0.6)180.0 / (4 * 545.0 * 137.9))^{1/2}$$

$$= 7.428 \text{ mm.}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress:

Web Length Dimension [Web Length]:

$$= 2 * \cos(90 - \text{Saddle Angle}/2) (\text{Inside Radius} + \text{Shell Thk} + \text{Wear Plate Thk})$$

$$= 2 * \cos(90 - 120.0/2) (290.5 + 10.0 + 10.0)$$

$$= 537.802 \text{ mm.}$$

Distance between Ribs [e]:

$$= \text{Web Length} / (\text{Nr ribs} - 1)$$

$$= 537.8018 / (4 - 1)$$

$$= 179.267 \text{ mm.}$$

Baseplate Pressure Area [Ap]:

$$= e * \text{Bpwid} / 2$$

$$= 179.2673 * 180.0 / 2$$

$$= 161.341 \text{ cm}^2$$

Axial Load [P]:

$$= Ap * Bp$$

$$= 161.3 * 0.03$$

$$= 4.954 \text{ kN}$$

Area of the Rib and Web [Ar]:

$$= \text{Rib Area} + \text{Web Area}$$

$$= 14.0 + 8.963$$

$$= 22.963 \text{ cm}^2$$

Compressive Stress [Sc]:

$$= P / Ar$$

$$= 5.0 / 22.9634$$

$$= 2.157 \text{ N./mm}^2$$

Check of Outside Ribs:

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	B	D	Y	A	AY	Io
Rib+Web	10.0	150.0	...	15.0	...	281.

Rib dimension [D]:

= Saddle Width - Web Thickness
 = 150.0 - 10.0
 = 140.000 mm.

Distance to Centroid from Datum [ytot]:

= AY / A
 = 0.0/22.963
 = 0.000 mm.

Distance to Centroid [C1]:

= Saddle Width / 2
 = 150.0/2
 = 75.000 mm.

Radius of Gyration [r]:

= sqrt(Total Inertia / Total Area)
 = sqrt(281.2/22.963)
 = 34.997 mm.

Length of Outer Rib [L]:

= Saddle Height - cos(theta/2)(radius + shlthk + wpdthk) - bpthk
 = 600.0 - cos(120.0/2)(290.5 + 10.0 + 10.0) - 16.0
 = 428.750 mm.

Intermediate Term [Cc]:

= sqrt(2 * pi² * Elastic Modulus / Yield Stress)
 = sqrt(2 * pi² * 0.19994E+09/206.9)
 = 138.135

Slenderness ratio [KL/r]:

= KL/r
 = 1 * 428.75/34.997
 = 12.251

Bending Moment [Rm]:

= F1 / (2 * Bplen) * e * L / 2
 = 0.2 / (2 * 545.0) * 179.267 * 428.75/2
 = 6.907 N-m

Compressive Allowable, KL/r < Cc (12.2511 < 138.1347) per AISC E2-1 [Sca]:

= (1 - (KL/r)² / (2 * Cc²)) Fy / (5/3 + 3 * (KL/r) / (8 * Cc) - (KL/r)³ / (8 * Cc³))
 = (1 - (12.25)² / (2 * 138.13²)) 207 /
 (5/3 + 3 * (12.25) / (8 * 138.13) - (12.25³) / (8 * 138.13³))
 = 121.2 N./mm²

AISC Unity Check of Outside Ribs (must be <= 1)

= Sc/Sca + (Rm * C1 / I) / Sba
 = 2.16/121.21 + (6.91 * 75.0/2812500) / 137.9
 = 0.019

Check of Inside Ribs:

Inertia of Saddle, Inner Ribs - Axial Direction

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

PV Elite 2019 SP1 Licensee: SPLM Licensed User

FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

	B	D	Y	A	AY	Io
Rib	10.0	140.0	0.0	14.0	0.0	281.
Web	179.3	10.0	0.0	17.9	0.0	1.49
Totals	31.9	...	283.

Distance to Centroid from Datum [ytot]:

$$= AY / A$$

$$= 0.0/31.927$$

$$= 0.000 \text{ mm.}$$

Distance to Centroid [C1]:

$$= \text{Saddle Width} / 2$$

$$= 150.0/2$$

$$= 75.000 \text{ mm.}$$

Length of Inner Rib [L]:

$$= \text{Saddle Height} - \sqrt{(\text{Ro} + \text{Wpdthk})^2 - (\text{Pitch}/2)^2} - \text{Bpthk}$$

$$= 600.0 - \sqrt{(310.5 + 10.0)^2 - (179.267/2)^2} - 16.0$$

$$= 286.719 \text{ mm.}$$

Radius of Gyration [r]:

$$= \sqrt{\text{Total Inertia} / \text{Total Area}}$$

$$= \sqrt{282.7/31.927}$$

$$= 29.755 \text{ mm.}$$

Slenderness ratio [KL/r]:

$$= KL/r$$

$$= 1 * 286.719/29.755$$

$$= 9.636$$

Unit Force [Force,u]:

$$= F1 / (2 * \text{Baseplate Length})$$

$$= 0.196 / (2 * 545.0)$$

$$= 0.000 \text{ kN/mm.}$$

Moment at base of inner Rib [Mbase,c]:

$$= \text{Unit Force} * e * L$$

$$= 0. * 179.267 * 286.719$$

$$= 9.238 \text{ N-m}$$

Bending Stress due to Transverse Force and Weight Load [SigmaB,base,c]:

$$= \text{Bending Moment} / \text{Section Modulus}$$

$$= 9.238/37688.074$$

$$= 0.245 \text{ N./mm}^2$$

Compressive Allowable, $KL/r < Cc$ ($9.6361 < 138.1347$) per AISC E2-1 [Sca]:

$$= (1 - (KL/r)^2 / (2 * Cc^2)) Fy / (5/3 + 3 * (KL/r) / (8 * Cc) - (KL/r^3) / (8 * Cc^3))$$

$$= (1 - (9.64)^2 / (2 * 138.13^2)) 207 /$$

$$(5/3 + 3 * (9.64) / (8 * 138.13) - (9.64^3) / (8 * 138.13^3))$$

$$= 121.9 \text{ N./mm}^2$$

AISC Unity Check of Inside Ribs (must be ≤ 1)

$$= Sc/Sca + (Mbase,c * C1/I) / Sba$$

$$= 3.06/121.9 + (9.24 * 75.0/282.66) / 137.9$$

$$= 0.027$$

Input Data for Base Plate Bolting Calculations:

Total Number of Bolts per BasePlate	Nbolts	4
Total Number of Bolts in Tension/Baseplate	Nbt	2

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Bolt Material Specification		SA-193 B7	
Bolt Allowable Stress	Stba	172.38	N./mm ²
Bolt Corrosion Allowance	Bca	0.0	mm.
Distance from Bolts to Edge	Edgedis	72.0	mm.
Nominal Bolt Diameter	Bnd	24.0000	mm.
Thread Series	Series	TEMA Metric	
BasePlate Allowable Stress	S	108.25	N./mm ²
Area Available in a Single Bolt	BltArea	3.1275	cm ²
Saddle Load QO (Weight)	QO	27.4	kN
Saddle Load QL (Wind/Seismic contribution)	QL	0.0	kN
Maximum Transverse Force	Ft	1.0	kN
Maximum Longitudinal Force	F1	0.2	kN
Saddle Bolted to Steel Foundation		Yes	

Shear Stress in a Single Bolt [taub]:
 = Shear Force / (2 * Bolt Area * Number of Bolts)
 = 1/(2 * 3.13 * 4)
 = 0.4 N./mm². Must be less than 103.4 N./mm².

Bolt Area Calculation per Dennis R. Moss

Bolt Area Requirement Due to Longitudinal Load [Bltarearl]:
 = 0.0 (QO > QL --> No Uplift in Longitudinal direction)

Bolt Area due to Shear Load [Bltarears]:
 = F1 / (Stba * Nbolts)
 = 0.2/(172.38 * 4.0)
 = 0.0028 cm²

Bolt Area due to Transverse Load:

Moment on Baseplate Due to Transverse Load [Rmom]:
 = B * Ft + Sum of X Moments
 = 600.0 * 1.0 + 0.0
 = 600.56 N-m

Eccentricity (e):
 = Rmom / QO
 = 600.56/27.37
 = 21.94 mm. < Bplen/6 --> No Uplift in Transverse direction

Bolt Area due to Transverse Load [Bltareart]:
 = 0 (No Uplift)

Required Area of a Single Bolt [Bltarear]:
 = max[Bltarearl, Bltarears, Bltareart]
 = max[0.0, 0.0028, 0.0]
 = 0.0028 cm²

ASME Horizontal Vessel Analysis: Stresses for the Right Saddle (per ASME Sec. VIII Div. 2 based on the Zick method.)

Input and Calculated Values:

Vessel Mean Radius	Rm	297.00	mm.
Stiffened Vessel Length per 4.15.6	L	5880.00	mm.
Distance from Saddle to Vessel tangent	a	992.00	mm.
Saddle Width	b	150.00	mm.
Saddle Bearing Angle	theta	120.00	degrees

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Tag no:E-PK6101-3 ECONOMIZER

PV Elite 2019 SP1 Licensee: SPLM Licensed User

FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Wear Plate Width	b1	225.00	mm.
Wear Plate Bearing Angle	thetal	132.00	degrees
Wear Plate Thickness	tr	10.0	mm.
Wear Plate Allowable Stress	Sr	137.90	N./mm ²

Shell Allowable Stress used in Calculation	235.81	N./mm ²
Head Allowable Stress used in Calculation	235.81	N./mm ²
Circumferential Efficiency in Plane of Saddle	1.00	
Circumferential Efficiency at Mid-Span	1.00	

Saddle Force Q, Test Case, no Ext. Forces	30.33	kN
---	-------	----

Horizontal Vessel Analysis Results:	Actual N./mm ²	Allowable N./mm ²

Long. Stress at Top of Midspan	55.91	235.81
Long. Stress at Bottom of Midspan	71.09	235.81
Long. Stress at Top of Saddles	86.94	235.81
Long. Stress at Bottom of Saddles	50.50	235.81

Tangential Shear in Shell	11.32	188.65
Circ. Stress at Horn of Saddle	10.34	353.71
Circ. Compressive Stress in Shell	1.49	235.81

Intermediate Results: Saddle Reaction Q due to Wind or Seismic**Saddle Reaction Force due to Wind Ft [Fwt]:**

$$= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E$$

$$= 3.0 * (2.0/2 + 0) * 600.0/537.8018$$

$$= 3.3 \text{ kN}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= \max(F_l, \text{Friction Load, Sum of X Forces}) * B / L_s$$

$$= \max(0.2, 0.0, 0) * 600.0/4000.0$$

$$= 0.0 \text{ kN}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st})$$

$$= 27 + \max(0.0, 3, 0, 0)$$

$$= 30.3 \text{ kN}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	30.92	kN
Transverse Shear Load Saddle	1.00	kN
Longitudinal Shear Load Saddle	0.20	kN

Hydrostatic Test Pressure at center of Vessel: 29.929 bars

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, k = 0.1

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0529	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0434
K7p = 0.0434			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right] \\
 &= -30 \cdot 992.0 \left[1 - \left(1 - \frac{992.0}{5880.0} + \frac{(297.0^2 - 0.0^2)}{(2 \cdot 992.0 \cdot 5880.0)} \right) / \left(1 + \frac{(4 \cdot 0.0)}{(3 \cdot 5880.0)} \right) \right] \\
 &= -4849.7 \text{ N-m}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= \frac{Q \cdot L}{4} \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) - 4a/L \\
 &= \frac{30 \cdot 5880}{4} \left(1 + 2 \frac{(297^2 - 0^2)}{(5880^2)} \right) / \left(1 + \frac{(4 \cdot 0)}{(3 \cdot 5880)} \right) - 4 \cdot 992 / 5880 \\
 &= 14729.0 \text{ N-m}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 t) \\
 &= 29.929 \cdot 297.0 / (2 \cdot 7.0) - 14729.0 / (\pi \cdot 297.0^2 \cdot 7.0) \\
 &= 55.91 \text{ N./mm}^2
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t) \\
 &= 29.929 \cdot 297.0 / (2 \cdot 7.0) + 14729.0 / (\pi \cdot 297.0^2 \cdot 7.0) \\
 &= 71.09 \text{ N./mm}^2
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 t) \\
 &= 29.929 \cdot 297.0 / (2 \cdot 7.0) - 4849.7 / (0.1066 \cdot \pi \cdot 297.0^2 \cdot 7.0) \\
 &= 86.94 \text{ N./mm}^2
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\
 &= 29.929 \cdot 297.0 / (2 \cdot 7.0) + 4849.7 / (0.1923 \cdot \pi \cdot 297.0^2 \cdot 7.0) \\
 &= 50.50 \text{ N./mm}^2
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= \frac{Q(L - 2a)}{L + (4 \cdot h^2 / 3)} \\
 &= \frac{30(5880.0 - 2 \cdot 992.0)}{5880.0 + (4 \cdot 0.0 / 3)} \\
 &= 20.1 \text{ kN}
 \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
 &= \frac{K2 \cdot T}{R_m \cdot t} \\
 &= \frac{1.1707 \cdot 20.09}{297.0 \cdot 7.0} \\
 &= 11.32 \text{ N./mm}^2
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 \cdot \sqrt{R_m \cdot t} \\
 &= 0.78 \cdot \sqrt{297.0 \cdot 7.0} \\
 &= 35.565 \text{ mm.}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 \cdot Q \cdot k / (t \cdot (b + X1 + X2)) \\
 &= -0.7603 \cdot 30 \cdot 0.1 / (7.0 \cdot (150.0 + 35.56 + 35.56)) \\
 &= -1.49 \text{ N./mm}^2
 \end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
 &= \min(b + 1.56 \cdot \sqrt{R_m \cdot t}, 2a) \\
 &= \min(150.0 + 1.56 \cdot \sqrt{297.0 \cdot 7.0}, 2 \cdot 992.0) \\
 &= 221.13 \text{ mm.}
 \end{aligned}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

= 1.0000 Materials are the same, test case

Circumferential Stress at Saddle Base with Wear Plate (4.15.26) [$\sigma_{6,r}$]:

$$= -K5 * Q * k / (B1(t + eta * tr))$$

$$= - 0.7603 * 30 * 0.1 / (221.13(7.0 + 1.0 * 10.0))$$

$$= -0.61 \text{ N./mm}^2$$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8R_m$ (4.15.27) [$\sigma_{7,r}$]:

$$= -Q / (4(t+eta*tr)b1) - 3*K7*Q / (2(t+eta*tr)^2)$$

$$= -30 / (4(7.0 + 1.0 * 10.0)221.13) -$$

$$3 * 0.053 * 30 / (2(7.0 + 1.0 * 10.0)^2)$$

$$= -10.34 \text{ N./mm}^2$$

Results for Vessel Ribs, Web and Base:

Baseplate Length	Bplen	545.0000	mm.
Baseplate Thickness	Bpthk	16.0000	mm.
Baseplate Width	Bpwid	180.0000	mm.
Number of Ribs (inc. outside ribs)	Nribs	4	
Rib Thickness	Ribtk	10.0000	mm.
Web Thickness	Webtk	10.0000	mm.
Web Location	Webloc	Center	
Saddle Yield Stress	Sy	206.9	N./
Height of Web at Center	Hw,c	273.0	mm.
Friction Coefficient	mu	0.400	

Note: In the tables below I_o is I for the rectangle + Area * Centroid Distance²

Moment of Inertia of Saddle - Transverse Direction (90 degrees to long axis)

	B	D	Y	A	AY	I_o
Shell	295.7	7.0	3.5	20.7	7244.9	0.343E+04
Wearplate	225.0	10.0	12.0	22.5	27000.0	0.326E+04
Web	10.0	273.5	153.7	27.3	420506.1	0.183E+04
BasePlate	180.0	16.0	298.5	28.8	859679.9	0.796E+04
Totals	99.3	1314430.9	0.165E+05

Distance to Centroid [C1]:

$$= AY / A$$

$$= 517.493 / 99.35$$

$$= 132.304 \text{ mm.}$$

Angle [beta]:

$$= 180 - \text{Saddle Angle} / 2$$

$$= 180 - 120.0 / 2$$

$$= 120.0$$

Saddle Splitting Coefficient [K1]:

$$= (1 + \cos(\beta) - 0.5 * \sin(\beta)^2) / (\pi - \beta + \sin(\beta) \cos(\beta))$$

$$= (1 + \cos(120.0) - 0.5 * \sin(120.0)^2) / (\pi - 2.094 + \sin(120.0) \cos(120.0))$$

$$= 0.2035$$

Saddle Splitting Force [Fh]:

$$= K1 * Q$$

$$= 0.204 * 30.325$$

$$= 6.1719 \text{ kN}$$

$$\text{Tension Stress, } \sigma_t = (F_h / A_s) = 0.7848 \text{ N./mm}^2$$

$$\text{Allowed Stress, } \sigma_a = 0.6 * \text{Yield Str} = 124.1100 \text{ N./mm}^2$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Saddle Splitting Dimension [d]:

$$= B - R * \sin(\theta) / \theta$$

$$= 600.0 - 293.5 * \sin(1.0472) / 1.0472$$

$$= 357.277 \text{ mm.}$$

$$\text{Bending Moment, } M = F_h * d = 2205.9661 \text{ N-m}$$

$$\text{Bending Stress, } S_b = (M * C_1 / I) = 1.7699 \text{ N./mm}^2$$

$$\text{Allowed Stress, } S_a = 2/3 * \text{Yield Str} = 137.9000 \text{ N./mm}^2$$

Minimum Thickness of Baseplate per Moss:

$$= (3(Q + \text{Saddle_Wt}) \text{BasePlateWidth} / (4 * \text{BasePlateLength} * \text{AllStress}))^{1/2}$$

$$= (3(30 + 0.6)180.0 / (4 * 545.0 * 137.9))^{1/2}$$

$$= 7.453 \text{ mm.}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress:

Web Length Dimension [Web Length]:

$$= 2 * \cos(90 - \text{Saddle Angle}/2) (\text{Inside Radius} + \text{Shell Thk} + \text{Wear Plate Thk})$$

$$= 2 * \cos(90 - 120.0/2) (290.5 + 10.0 + 10.0)$$

$$= 537.802 \text{ mm.}$$

Distance between Ribs [e]:

$$= \text{Web Length} / (\text{Nr ribs} - 1)$$

$$= 537.8018 / (4 - 1)$$

$$= 179.267 \text{ mm.}$$

Baseplate Pressure Area [Ap]:

$$= e * \text{Bpwid} / 2$$

$$= 179.2673 * 180.0 / 2$$

$$= 161.341 \text{ cm}^2$$

Axial Load [P]:

$$= A_p * B_p$$

$$= 161.3 * 0.03$$

$$= 4.987 \text{ kN}$$

Area of the Rib and Web [Ar]:

$$= \text{Rib Area} + \text{Web Area}$$

$$= 14.0 + 8.963$$

$$= 22.963 \text{ cm}^2$$

Compressive Stress [Sc]:

$$= P / A_r$$

$$= 5.0 / 22.9634$$

$$= 2.172 \text{ N./mm}^2$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	B	D	Y	A	AY	Io
Rib+Web	10.0	150.0	...	15.0	...	281.

Rib dimension [D]:

$$= \text{Saddle Width} - \text{Web Thickness}$$

$$= 150.0 - 10.0$$

$$= 140.000 \text{ mm.}$$

Distance to Centroid from Datum [ytot]:

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

$$= AY / A$$

$$= 0.0/22.963$$

$$= 0.000 \text{ mm.}$$

Distance to Centroid [C1]:

$$= \text{Saddle Width} / 2$$

$$= 150.0/2$$

$$= 75.000 \text{ mm.}$$

Radius of Gyration [r]:

$$= \sqrt{\text{Total Inertia} / \text{Total Area}}$$

$$= \sqrt{281.2/22.963}$$

$$= 34.997 \text{ mm.}$$

Length of Outer Rib [L]:

$$= \text{Saddle Height} - \cos(\text{theta}/2) (\text{radius} + \text{shlthk} + \text{wpdthk}) - \text{bpthk}$$

$$= 600.0 - \cos(120.0/2) (290.5 + 10.0 + 10.0) - 16.0$$

$$= 428.750 \text{ mm.}$$

Intermediate Term [Cc]:

$$= \sqrt{2 * \pi^2 * \text{Elastic Modulus} / \text{Yield Stress}}$$

$$= \sqrt{2 * \pi^2 * 0.19994\text{E}+09/206.9}$$

$$= 138.135$$

Slenderness ratio [KL/r]:

$$= KL/r$$

$$= 1 * 428.75/34.997$$

$$= 12.251$$

Bending Moment [Rm]:

$$= F1 / (2 * B\text{plen}) * e * L / 2$$

$$= 0.2 / (2 * 545.0) * 179.267 * 428.75/2$$

$$= 6.907 \text{ N-m}$$

Compressive Allowable, $KL/r < Cc$ ($12.2511 < 138.1347$) per AISC E2-1 [Sca]:

$$= (1 - (KL/r)^2 / (2 * Cc^2)) Fy / (5/3 + 3 * (KL/r) / (8 * Cc) - (KL/r^3) / (8 * Cc^3))$$

$$= (1 - (12.25)^2 / (2 * 138.13^2)) 207 / (5/3 + 3 * (12.25) / (8 * 138.13) - (12.25^3) / (8 * 138.13^3))$$

$$= 121.2 \text{ N./mm}^2$$

AISC Unity Check of Outside Ribs (must be ≤ 1)

$$= Sc/Sca + (Rm * C1 / I) / Sba$$

$$= 2.17/121.21 + (6.91 * 75.0/2812500) / 137.9$$

$$= 0.019$$

Check of Inside Ribs:

Inertia of Saddle, Inner Ribs - Axial Direction

	B	D	Y	A	AY	Io
Rib	10.0	140.0	0.0	14.0	0.0	281.
Web	179.3	10.0	0.0	17.9	0.0	1.49
Totals	31.9	...	283.

Distance to Centroid from Datum [ytot]:

$$= AY / A$$

$$= 0.0/31.927$$

$$= 0.000 \text{ mm.}$$

Distance to Centroid [C1]:

$$= \text{Saddle Width} / 2$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

$$= 150.0/2$$

$$= 75.000 \text{ mm.}$$

Length of Inner Rib [L]:

$$= \text{Saddle Height} - \sqrt{(\text{Ro} + \text{Wpdthk})^2 - (\text{Pitch}/2)^2} - \text{Bpthk}$$

$$= 600.0 - \sqrt{(310.5 + 10.0)^2 - (179.267/2)^2} - 16.0$$

$$= 286.719 \text{ mm.}$$

Radius of Gyration [r]:

$$= \sqrt{\text{Total Inertia} / \text{Total Area}}$$

$$= \sqrt{282.7/31.927}$$

$$= 29.755 \text{ mm.}$$

Slenderness ratio [KL/r]:

$$= \text{KL}/r$$

$$= 1 * 286.719/29.755$$

$$= 9.636$$

Unit Force [Force,u]:

$$= \text{F1} / (2 * \text{Baseplate Length})$$

$$= 0.196 / (2 * 545.0)$$

$$= 0.000 \text{ kN/mm.}$$

Moment at base of inner Rib [Mbase,c]:

$$= \text{Unit Force} * e * L$$

$$= 0. * 179.267 * 286.719$$

$$= 9.238 \text{ N-m}$$

Bending Stress due to Transverse Force and Weight Load [SigmaB,base,c]:

$$= \text{Bending Moment} / \text{Section Modulus}$$

$$= 9.238/37688.074$$

$$= 0.245 \text{ N./mm}^2$$

Compressive Allowable, $\text{KL}/r < \text{Cc}$ ($9.6361 < 138.1347$) per AISC E2-1 [Sca]:

$$= (1 - (\text{KL}/r)^2 / (2 * \text{Cc}^2)) \text{Fy} / (5/3 + 3 * (\text{KL}/r) / (8 * \text{Cc}) - (\text{KL}/r)^3 / (8 * \text{Cc}^3))$$

$$= (1 - (9.64)^2 / (2 * 138.13^2)) 207 /$$

$$(5/3 + 3 * (9.64) / (8 * 138.13) - (9.64^3) / (8 * 138.13^3))$$

$$= 121.9 \text{ N./mm}^2$$

AISC Unity Check of Inside Ribs (must be ≤ 1)

$$= \text{Sc}/\text{Sca} + (\text{Mbase,c} * \text{C1}/\text{I})/\text{Sba}$$

$$= 3.08/121.9 + (9.24 * 75.0/282.66)/137.9$$

$$= 0.027$$

Input Data for Base Plate Bolting Calculations:

Total Number of Bolts per BasePlate	Nbolts	4	
Total Number of Bolts in Tension/Baseplate	Nbt	2	
Bolt Material Specification		SA-193 B7	
Bolt Allowable Stress	Stba	172.38	N./mm ²
Bolt Corrosion Allowance	Bca	0.0	mm.
Distance from Bolts to Edge	Edgedis	72.0	mm.
Nominal Bolt Diameter	Bnd	24.0000	mm.
Thread Series	Series	TEMA Metric	
BasePlate Allowable Stress	S	108.25	N./mm ²
Area Available in a Single Bolt	BltArea	3.1275	cm ²
Saddle Load QO (Weight)	QO	27.6	kN
Saddle Load QL (Wind/Seismic contribution)	QL	0.0	kN
Maximum Transverse Force	Ft	1.0	kN
Maximum Longitudinal Force	F1	0.2	kN
Saddle Bolted to Steel Foundation		Yes	

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 Horizontal Vessel Analysis (Test): Step: 13 11:46pm Dec 22,2021

Shear Stress in a Single Bolt [τ_{aub}]:

$$\begin{aligned}
 &= \text{Shear Force} / (2 * \text{Bolt Area} * \text{Number of Bolts}) \\
 &= 1 / (2 * 3.13 * 4) \\
 &= 0.4 \text{ N./mm}^2. \text{ Must be less than } 103.4 \text{ N./mm}^2.
 \end{aligned}$$

Bolt Area Calculation per Dennis R. Moss

Bolt Area Requirement Due to Longitudinal Load [Bltarearl]:

$$= 0.0 \text{ (} QO > QL \text{ --> No Uplift in Longitudinal direction)}$$

Bolt Area due to Shear Load [Bltarears]:

$$\begin{aligned}
 &= F1 / (Stba * Nbolts) \\
 &= 0.2 / (172.38 * 4.0) \\
 &= 0.0028 \text{ cm}^2
 \end{aligned}$$

Bolt Area due to Transverse Load:

Moment on Baseplate Due to Transverse Load [Rmom]:

$$\begin{aligned}
 &= B * Ft + \text{Sum of X Moments} \\
 &= 600.0 * 1.0 + 0.0 \\
 &= 600.56 \text{ N-m}
 \end{aligned}$$

Eccentricity (e):

$$\begin{aligned}
 &= Rmom / QO \\
 &= 600.56 / 27.57 \\
 &= 21.77 \text{ mm.} < Bplen/6 \text{ --> No Uplift in Transverse direction}
 \end{aligned}$$

Bolt Area due to Transverse Load [Bltareart]:

$$= 0 \text{ (No Uplift)}$$

Required Area of a Single Bolt [Bltarear]:

$$\begin{aligned}
 &= \max[\text{Bltarearl}, \text{Bltarears}, \text{Bltareart}] \\
 &= \max[0.0, 0.0028, 0.0] \\
 &= 0.0028 \text{ cm}^2
 \end{aligned}$$

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 Nozzle Summary: Step: 22 11:46pm Dec 22,2021

Nozzle Calculation Summary:

Description	MAWP bars	Ext	MAPNC bars	UG-45	[tr] mm.	Weld Path	Areas or Stresses
T4	...	OK	...	OK	6.22	OK	Passed
T1	...	OK	...	OK	7.95	OK	Passed
S2	...	OK	...	OK	7.94	OK	Passed
S1	...	OK	...	OK	7.94	OK	Passed
S3	...	OK	...	OK	6.42	OK	Passed
T2	...	OK	...	OK	7.94	OK	Passed
T3	...	OK	...	OK	6.42	OK	Passed

MAWP Summary:

Minimum MAWP Nozzles : 0.000 Nozzle : T3

Note: MAWPs (Internal Case) shown above are at the High Point.

Check the Spatial Relationship between the Nozzles

From Node	Nozzle Description	X Coordinate mm.	Layout Angle deg	Dia. Limit mm.
20	T4	253.000	90.000	61.550
20	T1	253.000	270.000	201.662
40	S2	809.175	90.000	310.145
40	S1	6293.175	90.000	310.145
40	S3	759.175	270.000	102.068
60	T2	6849.351	90.000	310.145
60	T3	6852.351	270.000	68.000

The nozzle spacing is computed by the following:

= Sqrt($l^2 + l_c^2$) where

l - Arc length along the inside vessel surface in the long. direction.

l_c - Arc length along the inside vessel surface in the circ. direction

If any interferences/violations are found, they will be noted below.

No interference violations have been detected !

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T4 Nozl: 8 11:46pm Dec 22,2021

Input, Nozzle Desc: T4 From: 20

Pressure for Reinforcement Calculations	P	23.000	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	505.4167	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		253.00	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

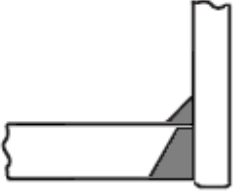
Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-350 LF2	
Material UNS Number		K03011	
Material Specification/Type		Forgings	
Allowable Stress at Temperature	Sn	137.90	N./mm ²
Allowable Stress At Ambient	Sna	137.90	N./mm ²
Diameter Basis (for tr calc only)		ID	
Layout Angle		90.00	deg
Diameter		0.7500	in.
Size and Thickness Basis		Actual	
Actual Thickness	tn	14.2500	mm.
Flange Material		SA-350 LF2	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)

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 Nozzle Calcs.: T4 Noz1: 8 11:46pm Dec 22,2021



Insert/Set-in Nozzle No Pad, no Inside projection

Reinforcement CALCULATION, Description: T4

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Inside Diameter Used in Calculation 0.750 in.
 Actual Thickness Used in Calculation 0.561 in.

Nozzle input data check completed without errors.

Reqd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]

$$= (P \cdot R) / (S_v \cdot E - 0.6 \cdot P) \text{ per UG-27 (c)(1)}$$

$$= (23.0 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.0)$$

$$= 4.9450 \text{ mm.}$$

Reqd thk per App. 1 of Nozzle Wall, Trn [Int. Press]

$$= R \left(\exp\left[\frac{P}{S_n \cdot E}\right] - 1 \right) \text{ per Appendix 1-2 (a)(1)}$$

$$= 12.525 \left(\exp\left[\frac{23.0}{137.9 \cdot 1.0}\right] - 1 \right)$$

$$= 0.2107 \text{ mm.}$$

Required Nozzle thickness under External Pressure per UG-28 : 0.2845 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit)	Dl	61.5500	mm.
Parallel to Vessel Wall	Rn+tn+t	30.7750	mm.
Normal to Vessel Wall (Thickness Limit), no pad	Tlnp	17.5000	mm.

Weld Strength Reduction Factor [fr1]:

$$= \min(1, S_n / S_v)$$

$$= \min(1, 137.9 / 137.9)$$

$$= 1.000$$

Weld Strength Reduction Factor [fr2]:

$$= \min(1, S_n / S_v)$$

$$= \min(1, 137.9 / 137.9)$$

$$= 1.000$$

Weld Strength Reduction Factor [fr3]:

$$= \min(fr2, fr4)$$

$$= \min(1.0, 1.0)$$

$$= 1.000$$

Results of Nozzle Reinforcement Area Calculations: (cm²)

AREA AVAILABLE, A1 to A5		Design	External	Mapnc
Area Required	Ar	1.239	0.230	NA
Area in Shell	A1	0.750	1.886	NA

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Nozzle Calcs.: T4 Nozl: 8 11:46pm Dec 22,2021

Area in Nozzle Wall	A2	3.864	3.838	NA
Area in Inward Nozzle	A3	0.000	0.000	NA
Area in Welds	A41+A42+A43	0.630	0.630	NA
Area in Element	A5	0.000	0.000	NA
TOTAL AREA AVAILABLE	Atot	5.244	6.354	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Sufficient.

Area Required [A]:

$$\begin{aligned}
 &= (d * tr * F + 2 * tn * tr * F * (1 - fr1)) \text{ UG-37(c)} \\
 &= (25.05 * 4.945 * 1.0 + 2 * 11.25 * 4.945 * 1.0 * (1 - 1.0)) \\
 &= 1.239 \text{ cm}^2
 \end{aligned}$$

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

$$\begin{aligned}
 &= d(E1 * t - F * tr) - 2 * tn(E1 * t - F * tr) * (1 - fr1) \\
 &= 36.5(1.0 * 7.0 - 1.0 * 4.945) - 2 * 11.25 \\
 &\quad (1.0 * 7.0 - 1.0 * 4.945) * (1 - 1.0) \\
 &= 0.750 \text{ cm}^2
 \end{aligned}$$

Area Available in Nozzle Projecting Outward [A2]:

$$\begin{aligned}
 &= (2 * tlnp)(tn - trn) fr2 \\
 &= (2 * 17.5)(11.25 - 0.21) 1.0 \\
 &= 3.864 \text{ cm}^2
 \end{aligned}$$

Area Available in Inward Weld + Outward Weld [A41 + A43]:

$$\begin{aligned}
 &= (Wo^2 - \text{Area Lost}) * fr2 + ((Wi - can / 0.707)^2 - \text{Area Lost}) * fr2 \\
 &= (8.0^2 - 0.01) * 1.0 + (0.0^2 - 0.0) * 1.0 \\
 &= 0.630 \text{ cm}^2
 \end{aligned}$$

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures	ta = 3.2845 mm.
Wall Thickness per UG16(b),	tr16b = 4.5000 mm.
Wall Thickness, shell/head, internal pressure	trb1 = 7.9450 mm.
Wall Thickness	tb1 = max(trb1, tr16b) = 7.9450 mm.
Wall Thickness	tb2 = max(trb2, tr16b) = 4.5000 mm.
Wall Thickness per table UG-45	tb3 = 6.2200 mm.

Determine Nozzle Thickness candidate [tb]:

$$\begin{aligned}
 &= \min[tb3, \max(tb1, tb2)] \\
 &= \min[6.22, \max(7.945, 4.5)] \\
 &= 6.2200 \text{ mm.}
 \end{aligned}$$

Minimum Wall Thickness of Nozzle Necks [tUG-45]:

$$\begin{aligned}
 &= \max(ta, tb) \\
 &= \max(3.2845, 6.22) \\
 &= 6.2200 \text{ mm.}
 \end{aligned}$$

Available Nozzle Neck Thickness = 14.2500 mm. --> OK

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

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 Nozzle Calcs.: T4 Nozl: 8 11:46pm Dec 22,2021

Impact Test Temperature provided per Specification -46 °C
 Calculated Minimum Design Metal Temperature -104 °C

Nozzle-Shell/Head Weld (UCS-66(a)1(b)), Curve: D

 Govrn. thk, tg = 10.0, tr = 4.945, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.706$, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Governing MDMT of all the sub-joints of this Junction : -48 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -86 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = $23.00 / 51.10 = 0.450$

Note:
 Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: T4

Intermediate Calc. for nozzle/shell Welds Tmin 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	$4.9000 = 0.7 * t_{min}$	$5.6560 = 0.7 * W_o$ mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:
 $= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv)$
 $= \max(0, (1.2387 - 0.7501 + 2 * 11.25 * 1.0 * (1.0 * 7.0 - 4.945)) 138)$
 $= 13.11$ kN

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:
 $= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv$
 $= (3.8638 + 0.0 + 0.63 - 0.0 * 1.0) * 138$
 $= 61.96$ kN

Weld Load [W2]:
 $= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv$
 $= (3.8638 + 0.0 + 0.63 + (1.575)) * 138$
 $= 83.68$ kN

Weld Load [W3]:
 $= (A2+A3+A4+A5+(2*tn*t*fr1))*S$
 $= (3.8638 + 0.0 + 0.63 + 0.0 + (1.575)) * 138$
 $= 83.68$ kN

Strength of Connection Elements for Failure Path Analysis

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 Nozzle Calcs.: T4 Nozl: 8 11:46pm Dec 22,2021

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * D_{lo} * W_o * 0.49 * S_{nw}$$

$$= (3.1416/2.0) * 47.55 * 8.0 * 0.49 * 138$$

$$= 40. \text{ kN}$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (D_{lr} + D_{lo}) / 4) * (Thk - Can) * 0.7 * S_n$$

$$= (3.1416 * 18.15) * (14.25 - 3.0) * 0.7 * 138$$

$$= 62. \text{ kN}$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * D_{lo} * (W_{gnvi-Cas}) * 0.74 * S_{ng}$$

$$= (3.1416/2.0) * 47.55 * (10.0 - 3.0) * 0.74 * 138$$

$$= 53. \text{ kN}$$

Strength of Failure Paths:

$$\text{PATH11} = (\text{SONW} + \text{SNW}) = (40 + 62) = 102 \text{ kN}$$

$$\text{PATH22} = (\text{Sonw} + \text{Tpgw} + \text{Tngw} + \text{Sinw})$$

$$= (40 + 0 + 53 + 0) = 94 \text{ kN}$$

$$\text{PATH33} = (\text{Sonw} + \text{Tngw} + \text{Sinw})$$

$$= (40 + 53 + 0) = 94 \text{ kN}$$

Summary of Failure Path Calculations:

Path 1-1 = 102 kN , must exceed W = 13 kN or W1 = 61 kN
 Path 2-2 = 93 kN , must exceed W = 13 kN or W2 = 83 kN
 Path 3-3 = 93 kN , must exceed W = 13 kN or W3 = 83 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 0.9745 mm.

The Cut Length for this Nozzle is, Drop + Ho + H + T : 210.9745 mm.

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 Nozzle Calcs.: T1 Nozl: 9 11:46pm Dec 22,2021

Input, Nozzle Desc: T1 From: 20

Pressure for Reinforcement Calculations	P	23.034	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	505.4167	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		253.00	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

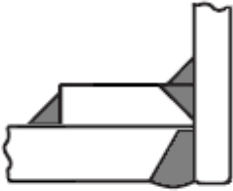
Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-333 6	
Material UNS Number		K03006	
Material Specification/Type	Smls. & wld. pipe		
Allowable Stress at Temperature	Sn	117.90	N./mm ²
Allowable Stress At Ambient	Sna	117.90	N./mm ²
Diameter Basis (for tr calc only)		OD	
Layout Angle		270.00	deg
Diameter		4.0000	in.
Size and Thickness Basis		Minimum	
Nominal Thickness	tn	120	
Flange Material		SA-350 LF2	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Pad Material		SA-516 70	
Pad Allowable Stress at Temperature	Sp	137.90	N./mm ²
Pad Allowable Stress At Ambient	Spa	137.90	N./mm ²
Diameter of Pad along vessel surface	Dp	230.0000	mm.
Thickness of Pad	te	10.0000	mm.
Weld leg size between Pad and Shell	Wp	8.0000	mm.
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm.
Reinforcing Pad Width		57.8500	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T1 Noz1: 9 11:46pm Dec 22,2021

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle With Pad, no Inside projection

Reinforcement CALCULATION, Description: T1

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Outside Diameter Used in Calculation	4.500 in.
Actual Thickness Used in Calculation	0.383 in.

Nozzle input data check completed without errors.

Reqd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]
 $= (P \cdot R) / (S_v \cdot E - 0.6 \cdot P)$ per UG-27 (c)(1)
 $= (23.03 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.03)$
 $= 4.9524 \text{ mm.}$

Reqd thk per UG-37(a) of Nozzle Wall, Trn [Int. Press]
 $= (P \cdot R_o) / (S_n \cdot E + 0.4 \cdot P)$ per Appendix 1-1 (a)(1)
 $= (23.03 \cdot 57.15) / (118 \cdot 1.0 + 0.4 \cdot 23.03)$
 $= 1.1079 \text{ mm.}$

Required Nozzle thickness under External Pressure per UG-28 : 0.4740 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit)	D1	201.6618	mm.
Parallel to Vessel Wall, opening length	d	100.8309	mm.
Normal to Vessel Wall (Thickness Limit), pad side Tlwp		17.5000	mm.

Note: The Pad diameter is greater than the Diameter Limit. The excess will not be considered.

Weld Strength Reduction Factor [fr1]:
 $= \min(1, S_n / S_v)$
 $= \min(1, 117.9 / 137.9)$
 $= 0.855$

Weld Strength Reduction Factor [fr2]:
 $= \min(1, S_n / S_v)$
 $= \min(1, 117.9 / 137.9)$
 $= 0.855$

Weld Strength Reduction Factor [fr4]:
 $= \min(1, S_p / S_v)$
 $= \min(1, 137.9 / 137.9)$
 $= 1.000$

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T1 Nozl: 9 11:46pm Dec 22,2021

Weld Strength Reduction Factor [fr3]:

$$= \min(fr2, fr4)$$

$$= \min(0.855, 1.0)$$

$$= 0.855$$

Results of Nozzle Reinforcement Area Calculations: (cm²)

AREA AVAILABLE, A1 to A5		Design	External	Mapnc
Area Required	Ar	5.090	0.942	NA
Area in Shell	A1	2.025	5.108	NA
Area in Nozzle Wall	A2	1.684	1.873	NA
Area in Inward Nozzle	A3	0.000	0.000	NA
Area in Welds	A41+A42+A43	0.545	0.545	NA
Area in Element	A5	6.552	6.552	NA
TOTAL AREA AVAILABLE	Atot	10.806	14.079	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Insufficient.
 The area available with the given pad is Sufficient.

SELECTION OF POSSIBLE REINFORCING PADS: Diameter Thickness

Based on given Pad Thickness:	125.4577	10.0000 mm.
Based on given Pad Diameter:	230.0000	1.2772 mm.
Based on Shell or Nozzle Thickness:	125.7620	9.7345 mm.

Area Required [A]:

$$= (d * tr * F + 2 * tn * tr * F * (1 - fr1)) \text{ UG-37(c)}$$

$$= (100.8309 * 4.9524 * 1.0 + 2 * 6.7345 * 4.9524 * 1.0 * (1 - 0.86))$$

$$= 5.090 \text{ cm}^2$$

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

$$= d(E1 * t - F * tr) - 2 * tn(E1 * t - F * tr) * (1 - fr1)$$

$$= 100.831(1.0 * 7.0 - 1.0 * 4.952) - 2 * 6.735$$

$$(1.0 * 7.0 - 1.0 * 4.9524) * (1 - 0.855)$$

$$= 2.025 \text{ cm}^2$$

Area Available in Nozzle Wall Projecting Outward [A2]:

$$= (2 * Tlwp) * (tn - trn) * fr2$$

$$= (2 * 17.5) * (6.73 - 1.11) * 0.855$$

$$= 1.684 \text{ cm}^2$$

Area Available in Welds [A41 + A42 + A43]:

$$= (Wo^2 - Ar \text{ Lost}) * Fr3 + ((Wi - can / 0.707)^2 - Ar \text{ Lost}) * fr2 + Wp^2 * fr4$$

$$= (0.6375) * 0.86 + (0.0) * 0.86 + 0.0^2 * 1.0$$

$$= 0.545 \text{ cm}^2$$

Area Available in Element, also see UG-37(h) [A5]:

$$= (\min(Dp, DL) - (\text{Nozzle OD})) (\min(tp, Tlwp, te)) * fr4 * 0.75$$

$$= (201.6618 - 114.3) 10.0 * 1.0 * 0.75$$

$$= 6.552 \text{ cm}^2$$

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures $t_a = 4.1079 \text{ mm.}$
 Wall Thickness per UG16(b), $tr_{16b} = 4.5000 \text{ mm.}$

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Nozzle Calcs.: T1 Noz1: 9 11:46pm Dec 22,2021

Wall Thickness, shell/head, internal pressure trbl = 7.9524 mm.
 Wall Thickness tb1 = max(trbl, tr16b) = 7.9524 mm.
 Wall Thickness tb2 = max(trb2, tr16b) = 4.5000 mm.
 Wall Thickness per table UG-45 tb3 = 8.2578 mm.

Determine Nozzle Thickness candidate [tb]:

= min[tb3, max(tb1,tb2)]
 = min[8.258, max(7.9524, 4.5)]
 = 7.9524 mm.

Minimum Wall Thickness of Nozzle Necks [tUG-45]:

= max(ta, tb)
 = max(4.1079, 7.9524)
 = 7.9524 mm.

Available Nozzle Neck Thickness = 9.7345 mm. --> OK

Stresses on Nozzle due to External and Pressure Loads per the ASME

B31.3 Piping Code (see 319.4.4 and 302.3.5):

Sustained	: 51.4,	Allowable	: 117.9 N./mm ²	Passed
Expansion	: 0.0,	Allowable	: 243.4 N./mm ²	Passed
Occasional	: 8.1,	Allowable	: 156.8 N./mm ²	Passed
Shear	: 24.8,	Allowable	: 82.5 N./mm ²	Passed

Note : The number of cycles on this nozzle was assumed to be 7000 or less for the determination of the expansion stress allowable.

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for the Nozzle (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for Reinforcement pad, Curve: B

Govrn. thk, tg = 9.735, tr = 1.108, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.165, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-104 °C

Shell to Pad Weld Junction at Pad OD, min(Curve:B, Curve:D)

Govrn. thk, tg = 10.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.707, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
--	--------

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T1 Noz1: 9 11:46pm Dec 22,2021

Min Metal Temp. at Required thickness (UCS 66.1) -45 °C

Nozzle-Shell/Head Weld (UCS-66(a)1(b)), Curve: D

Govrn. thk, tg = 9.735, tr = 1.108, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.165, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Min Metal Temp. at Required thickness (UCS 66.1) -104 °C

Governing MDMT of the Nozzle : -104 °C
 Governing MDMT of the Reinforcement Pad : -45 °C
 Governing MDMT of all the sub-joints of this Junction : -45 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -85 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = 23.03/51.10 = 0.451

Note:

Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: T1

Intermediate Calc. for nozzle/shell Welds Tmin 6.7345 mm.
 Intermediate Calc. for pad/shell Welds TminPad 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	4.7142 = 0.7 * tmin.	5.6560 = 0.7 * Wo mm.
Pad Weld	3.5000 = 0.5*TminPad	5.6560 = 0.7 * Wp mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:

$$\begin{aligned}
 &= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv) \\
 &= \max(0, (5.0903 - 2.0246 + 2 * 6.7345 * 0.855 * \\
 &\quad (1.0 * 7.0 - 4.9524))138) \\
 &= 45.52 \text{ kN}
 \end{aligned}$$

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:

$$\begin{aligned}
 &= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv \\
 &= (1.6838 + 6.5521 + 0.5451 - 0.0 * 0.86) * 138 \\
 &= 121.08 \text{ kN}
 \end{aligned}$$

Weld Load [W2]:

$$\begin{aligned}
 &= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv \\
 &= (1.6838 + 0.0 + 0.5472 + (0.8061)) * 138 \\
 &= 41.88 \text{ kN}
 \end{aligned}$$

Weld Load [W3]:

$$\begin{aligned}
 &= (A2+A3+A4+A5+(2*tn*t*fr1))*S \\
 &= (1.6838 + 0.0 + 0.5451 + 6.5521 + (0.8061)) * 138 \\
 &= 132.19 \text{ kN}
 \end{aligned}$$

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T1 Nozl: 9 11:46pm Dec 22,2021

Strength of Connection Elements for Failure Path Analysis

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * D_{lo} * W_o * 0.49 * S_{nw}$$

$$= (3.1416/2.0) * 114.3 * 8.0 * 0.49 * 118$$

$$= 83. \text{ kN}$$

Shear, Pad Element Weld [Spew]:

$$= (\pi/2) * DP * WP * 0.49 * SEW$$

$$= (3.1416/2.0) * 230.0 * 8.0 * 0.49 * 138$$

$$= 195. \text{ kN}$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (D_{lr} + D_{lo}) / 4) * (Thk - Can) * 0.7 * S_n$$

$$= (3.1416 * 53.7827) * (9.7345 - 3.0) * 0.7 * 118$$

$$= 94. \text{ kN}$$

Tension, Pad Groove Weld [Tpgw]:

$$= (\pi/2) * D_{lo} * W_{gpn} * 0.74 * S_{eg}$$

$$= (3.1416/2) * 114.3 * 10.0 * 0.74 * 138$$

$$= 183. \text{ kN}$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * D_{lo} * (W_{gnvi} - Cas) * 0.74 * S_{ng}$$

$$= (3.1416/2.0) * 114.3 * (10.0 - 3.0) * 0.74 * 138$$

$$= 128. \text{ kN}$$

Strength of Failure Paths:

$$PATH11 = (SPEW + SNW) = (195 + 94) = 289 \text{ kN}$$

$$PATH22 = (Sonw + Tpgw + Tngw + Sinw)$$

$$= (83 + 183 + 128 + 0) = 394 \text{ kN}$$

$$PATH33 = (Spew + Tngw + Sinw)$$

$$= (195 + 128 + 0) = 324 \text{ kN}$$

Summary of Failure Path Calculations:

Path 1-1 = 289 kN , must exceed W = 45 kN or W1 = 121 kN
 Path 2-2 = 394 kN , must exceed W = 45 kN or W2 = 41 kN
 Path 3-3 = 323 kN , must exceed W = 45 kN or W3 = 132 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 5.6770 mm.

The Cut Length for this Nozzle is, Drop + Ho + H + T : 215.6770 mm.

Input Echo, WRC107/537 Item 1, Description: T1 :

Diameter Basis for Vessel	Vbasis	ID	
Cylindrical or Spherical Vessel	Cylsph	Cylindrical	
Internal Corrosion Allowance	Cas	3.0000	mm.
Vessel Diameter	Dv	581.000	mm.
Vessel Thickness	Tv	10.000	mm.
Design Temperature	T1	120.0	°C
Vessel Material		SA-516 70	
Vessel UNS Number		K02700	
Vessel Cold S.I. Allowable	Smc	137.90	N./mm ²
Vessel Hot S.I. Allowable	Smh	137.90	N./mm ²

Note:

Using 2 * Yield for Discontinuity Stress Allowable (Div 2, 4.1.6.3), Sps.

DEHDASHT PETROCHEMICAL INDUSTRY COMPANY
 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T1 Nozl: 9 11:46pm Dec 22,2021

Make sure that material properties at this temperature are not time-dependent for Material: SA-516 70

Attachment Type	Type	Round	
Diameter Basis for Nozzle	Nbasis	OD	
Corrosion Allowance for Nozzle	Can	3.0000	mm.
Nozzle Diameter	Dn	114.300	mm.
Nozzle Thickness	Tn	9.735	mm.
Nozzle Material		SA-333 6	
Nozzle UNS Number		K03006	
Nozzle Cold S.I. Allowable	SNmc	117.90	N./mm ²
Nozzle Hot S.I. Allowable	SNmh	117.90	N./mm ²
Thickness of Reinforcing Pad	Tpad	10.000	mm.
Diameter of Reinforcing Pad	Dpad	230.000	mm.
Design Internal Pressure	Dp	23.034	bars
Include Pressure Thrust		No	

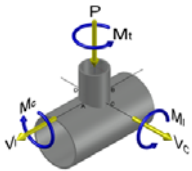
External Forces and Moments in WRC 107/537 Convention:

Radial Load (SUS)	P	4.0	kN
Longitudinal Shear (SUS)	Vl	4.0	kN
Circumferential Shear (SUS)	Vc	4.0	kN
Circumferential Moment (SUS)	Mc	1700.0	N-m
Longitudinal Moment (SUS)	Ml	1700.0	N-m
Torsional Moment (SUS)	Mt	2100.0	N-m

Use Interactive Control		No
WRC107 Version	Version	March 1979
Include Pressure Stress Indices per Div. 2		No
Compute Pressure Stress per WRC-368		No
Local Loads applied at end of Nozzle/Attachment		No

Note:

WRC Bulletin 537 provides equations for the dimensionless curves found in bulletin 107. As noted in the foreword to bulletin 537, "537 is equivalent to WRC 107". Where 107 is printed in the results below, "537" can be interchanged with "107".



Stress Attenuation Diameter (for Insert Plates) per WRC 297:

$$\begin{aligned}
 &= \text{NozzleOD} + 2 * 1.65 * \text{sqrt}(\text{Rmean}(t - ca)) \\
 &= 114.3 + 2 * 1.65 * \text{sqrt}(297.0 (10.0 - 3.0)) \\
 &= 264.767 \text{ mm.}
 \end{aligned}$$

WRC 107 Stress Calculation for SUSTained loads:

Radial Load	P	4.0	kN
Circumferential Shear	VC	4.0	kN
Longitudinal Shear	VL	4.0	kN
Circumferential Moment	MC	1700.0	N-m
Longitudinal Moment	ML	1700.0	N-m

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Calcs.: T1 Nozl: 9 11:46pm Dec 22,2021

Torsional Moment MT 2100.0 N-m

Dimensionless Parameters used : Gamma = 17.76

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.166	4C	3.000	(A,B)
N(PHI) / (P/Rm)	0.166	3C	2.539	(C,D)
M(PHI) / (P)	0.166	2C1	0.077	(A,B)
M(PHI) / (P)	0.166	1C	0.107	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.166	3A	0.597	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.166	1A	0.095	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.166	3B	1.885	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.166	1B	0.044	(A,B,C,D)
N(x) / (P/Rm)	0.166	3C	2.539	(A,B)
N(x) / (P/Rm)	0.166	4C	3.000	(C,D)
M(x) / (P)	0.166	1C1	0.112	(A,B)
M(x) / (P)	0.166	2C	0.077	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.166	4A	0.893	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.166	2A	0.052	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.166	4B	0.568	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.166	2B	0.069	(A,B,C,D)

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction (N./mm^2)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-2.3	-2.3	-2.3	-2.3	-2.0	-2.0	-2.0	-2.0
Circ. Bend. P		-6.4	6.4	-6.4	6.4	-8.9	8.9	-8.9	8.9
Circ. Memb. MC		0.0	0.0	0.0	0.0	-4.0	-4.0	4.0	4.0
Circ. Memb. MC		0.0	0.0	0.0	0.0	-67.0	67.0	67.0	-67.0
Circ. Memb. ML		-12.5	-12.5	12.5	12.5	0.0	0.0	0.0	0.0
Circ. Bend. ML		-31.1	31.1	31.1	-31.1	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-52.3	22.7	34.9	-14.6	-81.8	69.9	60.0	-56.1
Long. Memb. P		-2.0	-2.0	-2.0	-2.0	-2.3	-2.3	-2.3	-2.3
Long. Bend. P		-9.3	9.3	-9.3	9.3	-6.4	6.4	-6.4	6.4
Long. Memb. MC		0.0	0.0	0.0	0.0	-5.9	-5.9	5.9	5.9
Long. Bend. MC		0.0	0.0	0.0	0.0	-36.6	36.6	36.6	-36.6
Long. Memb. ML		-3.8	-3.8	3.8	3.8	0.0	0.0	0.0	0.0
Long. Bend. ML		-48.7	48.7	48.7	-48.7	0.0	0.0	0.0	0.0
Tot. Long. Str.		-63.7	52.3	41.2	-37.7	-51.2	34.7	33.8	-26.7
Shear VC		1.3	1.3	-1.3	-1.3	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.3	-1.3	1.3	1.3
Shear MT		6.0	6.0	6.0	6.0	6.0	6.0	6.0	6.0
Tot. Shear		7.3	7.3	4.7	4.7	4.7	4.7	7.3	7.3
Str. Int.		67.3	54.0	43.7	38.6	82.5	70.6	61.9	57.8

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T1 Noz1: 9 11:46pm Dec 22,2021

Dimensionless Parameters used : Gamma = 42.43

Dimensionless Loads for Cylindrical Shells at Pad edge:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.339	4C	4.173	(A,B)
N(PHI) / (P/Rm)	0.339	3C	1.862	(C,D)
M(PHI) / (P)	0.339	2C1	0.012	(A,B)
M(PHI) / (P)	0.339	1C !	0.063	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.339	3A	1.336	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.339	1A	0.064	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.339	3B	2.607	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.339	1B	0.010	(A,B,C,D)
N(x) / (P/Rm)	0.339	3C	1.862	(A,B)
N(x) / (P/Rm)	0.339	4C	4.173	(C,D)
M(x) / (P)	0.339	1C1	0.030	(A,B)
M(x) / (P)	0.339	2C !	0.032	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.339	4A	4.079	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.339	2A	0.025	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.339	4B	1.423	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.339	2B	0.016	(A,B,C,D)

Note - The ! mark next to the figure name denotes curve value exceeded.

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad (N./mm^2)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-8.0	-8.0	-8.0	-8.0	-3.6	-3.6	-3.6	-3.6
Circ. Bend. P		-5.9	5.9	-5.9	5.9	-30.9	30.9	-30.9	30.9
Circ. Memb. MC		0.0	0.0	0.0	0.0	-10.9	-10.9	10.9	10.9
Circ. Memb. MC		0.0	0.0	0.0	0.0	-131.6	131.6	131.6	-131.6
Circ. Memb. ML		-21.2	-21.2	21.2	21.2	0.0	0.0	0.0	0.0
Circ. Bend. ML		-20.1	20.1	20.1	-20.1	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-55.3	-3.2	27.3	-1.0	-177.0	148.1	107.9	-93.4
Long. Memb. P		-3.6	-3.6	-3.6	-3.6	-8.0	-8.0	-8.0	-8.0
Long. Bend. P		-14.7	14.7	-14.7	14.7	-15.9	15.9	-15.9	15.9
Long. Memb. MC		0.0	0.0	0.0	0.0	-33.1	-33.1	33.1	33.1
Long. Bend. MC		0.0	0.0	0.0	0.0	-52.3	52.3	52.3	-52.3
Long. Memb. ML		-11.6	-11.6	11.6	11.6	0.0	0.0	0.0	0.0
Long. Bend. ML		-32.6	32.6	32.6	-32.6	0.0	0.0	0.0	0.0
Tot. Long. Str.		-62.4	32.1	25.8	-9.9	-109.4	27.1	61.5	-11.3
Shear VC		1.6	1.6	-1.6	-1.6	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.6	-1.6	1.6	1.6
Shear MT		3.6	3.6	3.6	3.6	3.6	3.6	3.6	3.6
Tot. Shear		5.2	5.2	2.0	2.0	2.0	2.0	5.2	5.2
Str. Int.		65.1	36.8	28.7	10.3	177.0	148.1	108.5	93.7

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		38.7	41.0	38.7	41.0	38.7	41.0	38.7	41.0
Circ. Pl (SUS)		-14.8	-14.8	10.1	10.1	-5.9	-5.9	2.0	2.0
Circ. Q (SUS)		-37.5	37.5	24.7	-24.7	-75.9	75.9	58.0	-58.0
Long. Pm (SUS)		19.3	19.3	19.3	19.3	19.3	19.3	19.3	19.3
Long. Pl (SUS)		-5.7	-5.7	1.8	1.8	-8.2	-8.2	3.6	3.6
Long. Q (SUS)		-58.0	58.0	39.4	-39.4	-43.0	43.0	30.2	-30.2
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.3	1.3	-1.3	-1.3	-1.3	-1.3	1.3	1.3
Shear Q (SUS)		6.0	6.0	6.0	6.0	6.0	6.0	6.0	6.0
Pm (SUS)		38.7	41.0	38.7	41.0	38.7	41.0	38.7	41.0
Pm+Pl (SUS)		24.0	26.3	48.9	51.2	32.8	35.1	40.7	43.0
Pm+Pl+Q (Total)		46.1	75.9	75.0	45.7	44.9	111.3	99.8	19.5

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	40.95	137.90	Passed
Pm+Pl (SUS)	51.15	206.85	Passed
Pm+Pl+Q (TOTAL)	111.29	413.70	Passed

*Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.*

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		95.4	97.8	95.4	97.8	95.4	97.8	95.4	97.8
Circ. Pl (SUS)		-29.2	-29.2	13.1	13.1	-14.4	-14.4	7.3	7.3
Circ. Q (SUS)		-26.0	26.0	14.1	-14.1	-162.5	162.5	100.6	-100.6
Long. Pm (SUS)		47.7	47.7	47.7	47.7	47.7	47.7	47.7	47.7
Long. Pl (SUS)		-15.1	-15.1	8.0	8.0	-41.2	-41.2	25.1	25.1
Long. Q (SUS)		-47.2	47.2	17.9	-17.9	-68.2	68.2	36.4	-36.4
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.6	1.6	-1.6	-1.6	-1.6	-1.6	1.6	1.6
Shear Q (SUS)		3.6	3.6	3.6	3.6	3.6	3.6	3.6	3.6
Pm (SUS)		95.4	97.8	95.4	97.8	95.4	97.8	95.4	97.8

Tag no:E-PK6101-3 ECONOMIZER
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 Nozzle Calcs.: T1 Noz1: 9 11:46pm Dec 22,2021

Pm+Pl (SUS)	66.3	68.6	108.6	110.9	81.0	83.3	102.8	105.1
Pm+Pl+Q (Total)	55.8	96.2	122.8	96.8	81.7	245.9	203.6	37.2

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	97.75	137.90	Passed
Pm+Pl (SUS)	110.94	206.85	Passed
Pm+Pl+Q (TOTAL)	245.86	413.70	Passed

*Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.*

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

Input, Nozzle Desc: S2 From: 40

Pressure for Reinforcement Calculations	P	23.000	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	5880.0005	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		809.18	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

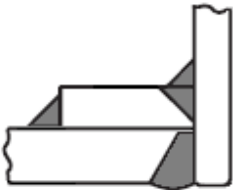
Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-333 6	
Material UNS Number		K03006	
Material Specification/Type	Smls. & wld. pipe		
Allowable Stress at Temperature	Sn	117.90	N./mm ²
Allowable Stress At Ambient	Sna	117.90	N./mm ²
Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		6.0000	in.
Size and Thickness Basis		Minimum	
Nominal Thickness	tn	80	
Flange Material		SA-350 LF2	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Pad Material		SA-516 70	
Pad Allowable Stress at Temperature	Sp	137.90	N./mm ²
Pad Allowable Stress At Ambient	Spa	137.90	N./mm ²
Diameter of Pad along vessel surface	Dp	290.0000	mm.
Thickness of Pad	te	10.0000	mm.
Weld leg size between Pad and Shell	Wp	8.0000	mm.
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm.
Reinforcing Pad Width		60.8625	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle With Pad, no Inside projection

Reinforcement CALCULATION, Description: S2

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Outside Diameter Used in Calculation	6.625 in.
Actual Thickness Used in Calculation	0.378 in.

Nozzle input data check completed without errors.

Req'd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]
 = $(P \cdot R) / (S_v \cdot E - 0.6 \cdot P)$ per UG-27 (c)(1)
 = $(23.0 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.0)$
 = 4.9450 mm.

Req'd thk per UG-37(a) of Nozzle Wall, Trn [Int. Press]
 = $(P \cdot R_o) / (S_n \cdot E + 0.4 \cdot P)$ per Appendix 1-1 (a)(1)
 = $(23.0 \cdot 84.1375) / (118 \cdot 1.0 + 0.4 \cdot 23.0)$
 = 1.6287 mm.

Required Nozzle thickness under External Pressure per UG-28 : 0.5944 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit)	D1	310.1452	mm.
Parallel to Vessel Wall, opening length	d	155.0726	mm.
Normal to Vessel Wall (Thickness Limit), pad side Tlwp		17.5000	mm.

Weld Strength Reduction Factor [fr1]:
 = $\min(1, S_n / S_v)$
 = $\min(1, 117.9 / 137.9)$
 = 0.855

Weld Strength Reduction Factor [fr2]:
 = $\min(1, S_n / S_v)$
 = $\min(1, 117.9 / 137.9)$
 = 0.855

Weld Strength Reduction Factor [fr4]:
 = $\min(1, S_p / S_v)$
 = $\min(1, 137.9 / 137.9)$
 = 1.000

Weld Strength Reduction Factor [fr3]:

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

= min(fr2, fr4)
 = min(0.855, 1.0)
 = 0.855

Results of Nozzle Reinforcement Area Calculations: (cm²)

AREA AVAILABLE, A1 to A5		Design	External	Mapnc
Area Required	Ar	7.763	4.029	NA
Area in Shell	A1	3.147	2.860	NA
Area in Nozzle Wall	A2	1.488	1.798	NA
Area in Inward Nozzle	A3	0.000	0.000	NA
Area in Welds A41+A42+A43		1.185	1.185	NA
Area in Element	A5	12.172	12.172	NA
TOTAL AREA AVAILABLE	Atot	17.993	18.015	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Insufficient.
 The area available with the given pad is Sufficient.

SELECTION OF POSSIBLE REINFORCING PADS: Diameter Thickness
 Based on given Pad Thickness: 187.6997 10.0000 mm.
 Based on given Pad Diameter: 290.0000 1.5958 mm.
 Based on Shell or Nozzle Thickness: 188.5066 9.6012 mm.

Area Required [A]:

= (d * tr*F + 2 * tn * tr*F * (1-fr1)) UG-37(c)
 = (155.0726*4.945*1.0+2*6.6012*4.945*1.0*(1-0.86))
 = 7.763 cm²

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

= d(E1*t - F*tr) - 2 * tn(E1*t - F*tr) * (1 - fr1)
 = 155.073(1.0 * 7.0 - 1.0 * 4.945) - 2 * 6.601
 (1.0 * 7.0 - 1.0 * 4.945) * (1 - 0.855)
 = 3.147 cm²

Area Available in Nozzle Wall Projecting Outward [A2]:

= (2 * Tlwp) * (tn - trn) * fr2
 = (2 * 17.5) * (6.6 - 1.63) * 0.855
 = 1.488 cm²

Area Available in Welds [A41 + A42 + A43]:

= (Wo² - Ar Lost)*Fr3+((Wi-can/0.707)² - Ar Lost)*fr2 + Wp²*fr4
 = (0.6375) * 0.86 + (0.0) * 0.86 + 203.2² * 1.0
 = 1.185 cm²

Area Available in Element [A5]:

= (min(Dp,DL)-(Nozzle OD))*(min(tp,Tlwp,te)) * fr4
 = (290.0 - 168.275) * 10.0 * 1.0
 = 12.172 cm²

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures ta = 4.6287 mm.
 Wall Thickness per UG16(b), tr16b = 4.5000 mm.
 Wall Thickness, shell/head, internal pressure trb1 = 7.9450 mm.
 Wall Thickness tbt1 = max(trb1, tr16b) = 7.9450 mm.

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Noz1: 10 11:46pm Dec 22,2021

Wall Thickness tb2 = max(trb2, tr16b) = 4.5000 mm.
 Wall Thickness per table UG-45 tb3 = 9.2200 mm.

Determine Nozzle Thickness candidate [tb]:
 = min[tb3, max(tb1,tb2)]
 = min[9.22, max(7.945, 4.5)]
 = 7.9450 mm.

Minimum Wall Thickness of Nozzle Necks [tUG-45]:
 = max(ta, tb)
 = max(4.6287, 7.945)
 = 7.9450 mm.

Available Nozzle Neck Thickness = 9.6012 mm. --> OK

**Stresses on Nozzle due to External and Pressure Loads per the ASME
 B31.3 Piping Code (see 319.4.4 and 302.3.5):**

Sustained	:	55.9,	Allowable	:	117.9 N./mm ²	Passed
Expansion	:	0.0,	Allowable	:	238.8 N./mm ²	Passed
Occasional	:	13.0,	Allowable	:	156.8 N./mm ²	Passed
Shear	:	24.1,	Allowable	:	82.5 N./mm ²	Passed

Note : The number of cycles on this nozzle was assumed to be 7000 or less for the determination of the expansion stress allowable.

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for the Nozzle (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for Reinforcement pad, Curve: B

Govrn. thk, tg = 9.601, tr = 1.629, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.247, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-104 °C

Shell to Pad Weld Junction at Pad OD, min(Curve:B, Curve:D)

Govrn. thk, tg = 10.0, tr = 4.945, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.706, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-45 °C

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

Nozzle-Shell/Head Weld (UCS-66(a)(1)(b)), Curve: D

Govrn. thk, tg = 9.601, tr = 1.629, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.247$, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Min Metal Temp. at Required thickness (UCS 66.1) -104 °C

Governing MDMT of the Nozzle : -104 °C
 Governing MDMT of the Reinforcement Pad : -45 °C
 Governing MDMT of all the sub-joints of this Junction : -45 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -86 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = 23.00/51.10 = 0.450

Note:
 Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: S2

Intermediate Calc. for nozzle/shell Welds Tmin 6.6012 mm.
 Intermediate Calc. for pad/shell Welds TminPad 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	4.6208 = 0.7 * tmin.	5.6560 = 0.7 * Wo mm.
Pad Weld	3.5000 = 0.5 * TminPad	5.6560 = 0.7 * Wp mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:
 $= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv)$
 $= \max(0, (7.763 - 3.1474 + 2 * 6.6012 * 0.855 * (1.0 * 7.0 - 4.945)) 138)$
 $= 66.84 \text{ kN}$

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:
 $= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv$
 $= (1.488 + 12.1725 + 1.1851 - 0.0 * 0.86) * 138$
 $= 204.70 \text{ kN}$

Weld Load [W2]:
 $= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv$
 $= (1.488 + 0.0 + 0.5472 + (0.7902)) * 138$
 $= 38.96 \text{ kN}$

Weld Load [W3]:
 $= (A2+A3+A4+A5+(2*tn*t*fr1))*S$
 $= (1.488 + 0.0 + 1.1851 + 12.1725 + (0.7902)) * 138$
 $= 215.60 \text{ kN}$

Strength of Connection Elements for Failure Path Analysis

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Noz1: 10 11:46pm Dec 22,2021

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * D_{lo} * W_o * 0.49 * S_{nw}$$

$$= (3.1416/2.0) * 168.275 * 8.0 * 0.49 * 118$$

$$= 122. \text{ kN}$$

Shear, Pad Element Weld [Spew]:

$$= (\pi/2) * DP * WP * 0.49 * SEW$$

$$= (3.1416/2.0) * 290.0 * 8.0 * 0.49 * 138$$

$$= 246. \text{ kN}$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (D_{lr} + D_{lo}) / 4) * (Thk - Can) * 0.7 * S_n$$

$$= (3.1416 * 80.8369) * (9.6012 - 3.0) * 0.7 * 118$$

$$= 138. \text{ kN}$$

Tension, Pad Groove Weld [Tpgw]:

$$= (\pi/2) * D_{lo} * W_{gpn} * 0.74 * S_{eg}$$

$$= (3.1416/2) * 168.275 * 10.0 * 0.74 * 138$$

$$= 270. \text{ kN}$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * D_{lo} * (W_{gnvi} - Cas) * 0.74 * S_{ng}$$

$$= (3.1416/2.0) * 168.275 * (10.0 - 3.0) * 0.74 * 138$$

$$= 189. \text{ kN}$$

Strength of Failure Paths:

$$PATH11 = (SPEW + SNW) = (246 + 138) = 385 \text{ kN}$$

$$PATH22 = (Sonw + Tpgw + Tngw + Sinw)$$

$$= (122 + 270 + 189 + 0) = 581 \text{ kN}$$

$$PATH33 = (Spew + Tngw + Sinw)$$

$$= (246 + 189 + 0) = 435 \text{ kN}$$

Summary of Failure Path Calculations:

Path 1-1 = 384 kN , must exceed W = 66 kN or W1 = 204 kN
 Path 2-2 = 580 kN , must exceed W = 66 kN or W2 = 38 kN
 Path 3-3 = 435 kN , must exceed W = 66 kN or W3 = 215 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 12.4512 mm.

The Cut Length for this Nozzle is, Drop + Ho + H + T : 222.4512 mm.

Input Echo, WRC107/537 Item 1, Description: S2 :

Diameter Basis for Vessel	Vbasis	ID	
Cylindrical or Spherical Vessel	Cylsph	Cylindrical	
Internal Corrosion Allowance	Cas	3.0000	mm.
Vessel Diameter	Dv	581.000	mm.
Vessel Thickness	Tv	10.000	mm.
Design Temperature	T1	120.0	°C
Vessel Material		SA-516 70	
Vessel UNS Number		K02700	
Vessel Cold S.I. Allowable	Smc	137.90	N./mm ²
Vessel Hot S.I. Allowable	Smh	137.90	N./mm ²

Note:

Using 2 * Yield for Discontinuity Stress Allowable (Div 2, 4.1.6.3), Sps.

Make sure that material properties at this temperature are not time-dependent for Material: SA-516 70

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

Attachment Type	Type	Round	
Diameter Basis for Nozzle	Nbasis	OD	
Corrosion Allowance for Nozzle	Can	3.0000	mm.
Nozzle Diameter	Dn	168.275	mm.
Nozzle Thickness	Tn	9.601	mm.
Nozzle Material		SA-333 6	
Nozzle UNS Number		K03006	
Nozzle Cold S.I. Allowable	SNmc	117.90	N./mm ²
Nozzle Hot S.I. Allowable	SNmh	117.90	N./mm ²
Thickness of Reinforcing Pad	Tpad	10.000	mm.
Diameter of Reinforcing Pad	Dpad	290.000	mm.
Design Internal Pressure	Dp	23.000	bars
Include Pressure Thrust		No	

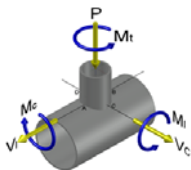
External Forces and Moments in WRC 107/537 Convention:

Radial Load	(SUS)	P	6.0	kN
Longitudinal Shear	(SUS)	Vl	6.0	kN
Circumferential Shear	(SUS)	Vc	6.0	kN
Circumferential Moment	(SUS)	Mc	3800.0	N-m
Longitudinal Moment	(SUS)	Ml	3800.0	N-m
Torsional Moment	(SUS)	Mt	4700.0	N-m

Use Interactive Control		No
WRC107 Version	Version	March 1979
Include Pressure Stress Indices per Div. 2		No
Compute Pressure Stress per WRC-368		No
Local Loads applied at end of Nozzle/Attachment		No

Note:

WRC Bulletin 537 provides equations for the dimensionless curves found in bulletin 107. As noted in the foreword to bulletin 537, "537 is equivalent to WRC 107". Where 107 is printed in the results below, "537" can be interchanged with "107".



Stress Attenuation Diameter (for Insert Plates) per WRC 297:

$$\begin{aligned}
 &= \text{NozzleOD} + 2 * 1.65 * \text{sqrt}(\text{Rmean}(t - ca)) \\
 &= 168.275 + 2 * 1.65 * \text{sqrt}(297.0 (10.0 - 3.0)) \\
 &= 318.742 \text{ mm.}
 \end{aligned}$$

WRC 107 Stress Calculation for SUSTained loads:

Radial Load	P	6.0	kN
Circumferential Shear	VC	6.0	kN
Longitudinal Shear	VL	6.0	kN
Circumferential Moment	MC	3800.0	N-m
Longitudinal Moment	ML	3800.0	N-m
Torsional Moment	MT	4700.0	N-m

Dimensionless Parameters used : Gamma = 17.76

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.244	4C	2.697	(A,B)
N(PHI) / (P/Rm)	0.244	3C	1.952	(C,D)
M(PHI) / (P)	0.244	2C1	0.045	(A,B)
M(PHI) / (P)	0.244	1C	0.074	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.244	3A	0.734	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.244	1A	0.085	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.244	3B	1.941	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.244	1B	0.032	(A,B,C,D)
N(x) / (P/Rm)	0.244	3C	1.952	(A,B)
N(x) / (P/Rm)	0.244	4C	2.697	(C,D)
M(x) / (P)	0.244	1C1	0.077	(A,B)
M(x) / (P)	0.244	2C	0.045	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.244	4A	1.267	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.244	2A	0.044	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.244	4B	0.709	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.244	2B	0.051	(A,B,C,D)

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction (N./mm^2)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-3.2	-3.2	-3.2	-3.2	-2.3	-2.3	-2.3	-2.3
Circ. Bend. P		-5.6	5.6	-5.6	5.6	-9.2	9.2	-9.2	9.2
Circ. Memb. MC		0.0	0.0	0.0	0.0	-7.4	-7.4	7.4	7.4
Circ. Memb. ML		0.0	0.0	0.0	0.0	-91.0	91.0	91.0	-91.0
Circ. Memb. ML		-19.5	-19.5	19.5	19.5	0.0	0.0	0.0	0.0
Circ. Bend. ML		-33.9	33.9	33.9	-33.9	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-62.2	16.8	44.6	-11.9	-109.9	90.5	86.9	-76.7
Long. Memb. P		-2.3	-2.3	-2.3	-2.3	-3.2	-3.2	-3.2	-3.2
Long. Bend. P		-9.6	9.6	-9.6	9.6	-5.6	5.6	-5.6	5.6
Long. Memb. MC		0.0	0.0	0.0	0.0	-12.7	-12.7	12.7	12.7
Long. Bend. MC		0.0	0.0	0.0	0.0	-47.2	47.2	47.2	-47.2
Long. Memb. ML		-7.1	-7.1	7.1	7.1	0.0	0.0	0.0	0.0
Long. Bend. ML		-55.1	55.1	55.1	-55.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-74.1	55.2	50.4	-40.7	-68.7	37.0	51.2	-32.1
Shear VC		1.3	1.3	-1.3	-1.3	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.3	-1.3	1.3	1.3
Shear MT		6.2	6.2	6.2	6.2	6.2	6.2	6.2	6.2
Tot. Shear		7.5	7.5	4.9	4.9	4.9	4.9	7.5	7.5
Str. Int.		77.7	56.7	53.2	41.5	110.4	91.0	88.4	77.9

Dimensionless Parameters used : Gamma = 42.43

Dimensionless Loads for Cylindrical Shells at Pad edge:

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 Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.427	4C	3.340	(A,B)
N(PHI) / (P/Rm)	0.427	3C	1.296	(C,D)
M(PHI) / (P)	0.427	2C1	0.008	(A,B)
M(PHI) / (P)	0.427	1C !	0.063	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.427	3A	1.071	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.427	1A	0.061	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.427	3B	1.851	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.427	1B	0.006	(A,B,C,D)
N(x) / (P/Rm)	0.427	3C	1.296	(A,B)
N(x) / (P/Rm)	0.427	4C	3.340	(C,D)
M(x) / (P)	0.427	1C1	0.019	(A,B)
M(x) / (P)	0.427	2C !	0.032	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.427	4A	4.057	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.427	2A	0.024	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.427	4B	1.123	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.427	2B	0.010	(A,B,C,D)

Note - The ! mark next to the figure name denotes curve value exceeded.

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-9.6	-9.6	-9.6	-9.6	-3.7	-3.7	-3.7	-3.7
Circ. Bend. P		-5.6	5.6	-5.6	5.6	-46.4	46.4	-46.4	46.4
Circ. Memb. MC		0.0	0.0	0.0	0.0	-15.4	-15.4	15.4	15.4
Circ. Memb. ML		0.0	0.0	0.0	0.0	-222.1	222.1	222.1	-222.1
Circ. Bend. ML		-26.7	-26.7	26.7	26.7	0.0	0.0	0.0	0.0
Circ. Bend. ML		-22.7	22.7	22.7	-22.7	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-64.6	-8.0	34.1	-0.1	-287.6	249.3	187.3	-164.0
Long. Memb. P		-3.7	-3.7	-3.7	-3.7	-9.6	-9.6	-9.6	-9.6
Long. Bend. P		-13.7	13.7	-13.7	13.7	-23.9	23.9	-23.9	23.9
Long. Memb. MC		0.0	0.0	0.0	0.0	-58.4	-58.4	58.4	58.4
Long. Bend. MC		0.0	0.0	0.0	0.0	-89.1	89.1	89.1	-89.1
Long. Memb. ML		-16.2	-16.2	16.2	16.2	0.0	0.0	0.0	0.0
Long. Bend. ML		-38.1	38.1	38.1	-38.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-71.7	31.9	36.8	-11.9	-181.1	44.9	114.1	-16.5
Shear VC		1.9	1.9	-1.9	-1.9	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.9	-1.9	1.9	1.9
Shear MT		5.1	5.1	5.1	5.1	5.1	5.1	5.1	5.1
Tot. Shear		7.0	7.0	3.2	3.2	3.2	3.2	7.0	7.0
Str. Int.		76.0	42.3	38.9	13.4	287.7	249.4	188.0	164.3

Tag no:E-PK6101-3 ECONOMIZER

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Nozzle Calcs.: S2 Noz1: 10 11:46pm Dec 22,2021

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		38.6	40.9	38.6	40.9	38.6	40.9	38.6	40.9
Circ. Pl (SUS)		-22.7	-22.7	16.4	16.4	-9.7	-9.7	5.1	5.1
Circ. Q (SUS)		-39.5	39.5	28.2	-28.2	-100.2	100.2	81.8	-81.8
Long. Pm (SUS)		19.3	19.3	19.3	19.3	19.3	19.3	19.3	19.3
Long. Pl (SUS)		-9.4	-9.4	4.8	4.8	-15.9	-15.9	9.6	9.6
Long. Q (SUS)		-64.6	64.6	45.5	-45.5	-52.8	52.8	41.6	-41.6
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.3	1.3	-1.3	-1.3	-1.3	-1.3	1.3	1.3
Shear Q (SUS)		6.2	6.2	6.2	6.2	6.2	6.2	6.2	6.2
Pm (SUS)		38.6	40.9	38.6	40.9	38.6	40.9	38.6	40.9
Pm+Pl (SUS)		16.2	18.4	55.0	57.3	29.0	31.3	43.8	46.1
Pm+Pl+Q (Total)		56.5	77.4	84.8	51.3	72.3	131.8	126.5	38.0

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	40.89	137.90	Passed
Pm+Pl (SUS)	57.30	206.85	Passed
Pm+Pl+Q (TOTAL)	131.75	413.70	Passed

Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		95.3	97.6	95.3	97.6	95.3	97.6	95.3	97.6
Circ. Pl (SUS)		-36.3	-36.3	17.0	17.0	-19.2	-19.2	11.7	11.7
Circ. Q (SUS)		-28.3	28.3	17.1	-17.1	-268.5	268.5	175.7	-175.7
Long. Pm (SUS)		47.7	47.7	47.7	47.7	47.7	47.7	47.7	47.7
Long. Pl (SUS)		-19.9	-19.9	12.4	12.4	-68.1	-68.1	48.8	48.8
Long. Q (SUS)		-51.8	51.8	24.4	-24.4	-113.0	113.0	65.3	-65.3
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.9	1.9	-1.9	-1.9	-1.9	-1.9	1.9	1.9
Shear Q (SUS)		5.1	5.1	5.1	5.1	5.1	5.1	5.1	5.1
Pm (SUS)		95.3	97.6	95.3	97.6	95.3	97.6	95.3	97.6
Pm+Pl (SUS)		59.1	61.4	112.4	114.7	96.6	98.9	107.3	109.6

Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S2 Nozl: 10 11:46pm Dec 22,2021

 Pm+Pl+Q (Total)| 56.5| 93.2| 129.7| 97.7| 192.5| 347.0| 283.0| 98.5|

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	97.61	137.90	Passed
Pm+Pl (SUS)	114.68	206.85	Passed
Pm+Pl+Q (TOTAL)	346.96	413.70	Passed

*Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.*

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

Input, Nozzle Desc: S1 From: 40

Pressure for Reinforcement Calculations	P	23.000	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	5880.0005	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		6293.18	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

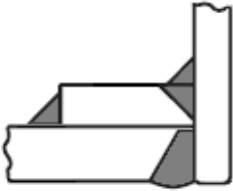
Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-333 6	
Material UNS Number		K03006	
Material Specification/Type	Smls. & wld. pipe		
Allowable Stress at Temperature	Sn	117.90	N./mm ²
Allowable Stress At Ambient	Sna	117.90	N./mm ²
Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		6.0000	in.
Size and Thickness Basis		Minimum	
Nominal Thickness	tn	80	
Flange Material		SA-350 LF2	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Pad Material		SA-516 70	
Pad Allowable Stress at Temperature	Sp	137.90	N./mm ²
Pad Allowable Stress At Ambient	Spa	137.90	N./mm ²
Diameter of Pad along vessel surface	Dp	290.0000	mm.
Thickness of Pad	te	10.0000	mm.
Weld leg size between Pad and Shell	Wp	8.0000	mm.
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm.
Reinforcing Pad Width		60.8625	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S1 Noz1: 11 11:46pm Dec 22,2021

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle With Pad, no Inside projection

Reinforcement CALCULATION, Description: S1

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Outside Diameter Used in Calculation	6.625 in.
Actual Thickness Used in Calculation	0.378 in.

Nozzle input data check completed without errors.

Reqd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]
 = $(P \cdot R) / (S_v \cdot E - 0.6 \cdot P)$ per UG-27 (c)(1)
 = $(23.0 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.0)$
 = 4.9450 mm.

Reqd thk per UG-37(a) of Nozzle Wall, Trn [Int. Press]
 = $(P \cdot R_o) / (S_n \cdot E + 0.4 \cdot P)$ per Appendix 1-1 (a)(1)
 = $(23.0 \cdot 84.1375) / (118 \cdot 1.0 + 0.4 \cdot 23.0)$
 = 1.6287 mm.

Required Nozzle thickness under External Pressure per UG-28 : 0.5944 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit)	D1	310.1452	mm.
Parallel to Vessel Wall, opening length	d	155.0726	mm.
Normal to Vessel Wall (Thickness Limit), pad side Tlwp		17.5000	mm.

Weld Strength Reduction Factor [fr1]:
 = $\min(1, S_n / S_v)$
 = $\min(1, 117.9 / 137.9)$
 = 0.855

Weld Strength Reduction Factor [fr2]:
 = $\min(1, S_n / S_v)$
 = $\min(1, 117.9 / 137.9)$
 = 0.855

Weld Strength Reduction Factor [fr4]:
 = $\min(1, S_p / S_v)$
 = $\min(1, 137.9 / 137.9)$
 = 1.000

Weld Strength Reduction Factor [fr3]:

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 Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

= min(fr2, fr4)
 = min(0.855, 1.0)
 = 0.855

Results of Nozzle Reinforcement Area Calculations: (cm²)

AREA AVAILABLE, A1 to A5	Design	External	Mapnc
Area Required Ar	7.763	4.029	NA
Area in Shell A1	3.147	2.860	NA
Area in Nozzle Wall A2	1.488	1.798	NA
Area in Inward Nozzle A3	0.000	0.000	NA
Area in Welds A41+A42+A43	1.185	1.185	NA
Area in Element A5	12.172	12.172	NA
TOTAL AREA AVAILABLE Atot	17.993	18.015	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Insufficient.
 The area available with the given pad is Sufficient.

SELECTION OF POSSIBLE REINFORCING PADS: Diameter Thickness
 Based on given Pad Thickness: 187.6997 10.0000 mm.
 Based on given Pad Diameter: 290.0000 1.5958 mm.
 Based on Shell or Nozzle Thickness: 188.5066 9.6012 mm.

Area Required [A]:

= (d * tr*F + 2 * tn * tr*F * (1-fr1)) UG-37(c)
 = (155.0726*4.945*1.0+2*6.6012*4.945*1.0*(1-0.86))
 = 7.763 cm²

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

= d(E1*t - F*tr) - 2 * tn(E1*t - F*tr) * (1 - fr1)
 = 155.073(1.0 * 7.0 - 1.0 * 4.945) - 2 * 6.601
 (1.0 * 7.0 - 1.0 * 4.945) * (1 - 0.855)
 = 3.147 cm²

Area Available in Nozzle Wall Projecting Outward [A2]:

= (2 * Tlwp) * (tn - trn) * fr2
 = (2 * 17.5) * (6.6 - 1.63) * 0.855
 = 1.488 cm²

Area Available in Welds [A41 + A42 + A43]:

= (Wo² - Ar Lost)*Fr3+((Wi-can/0.707)² - Ar Lost)*fr2 + Wp²*fr4
 = (0.6375) * 0.86 + (0.0) * 0.86 + 203.2² * 1.0
 = 1.185 cm²

Area Available in Element [A5]:

= (min(Dp,DL)-(Nozzle OD))*(min(tp,Tlwp,te)) * fr4
 = (290.0 - 168.275) * 10.0 * 1.0
 = 12.172 cm²

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures ta = 4.6287 mm.
 Wall Thickness per UG16(b), tr16b = 4.5000 mm.
 Wall Thickness, shell/head, internal pressure trb1 = 7.9450 mm.
 Wall Thickness tbt1 = max(trb1, tr16b) = 7.9450 mm.

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 Nozzle Calcs.: S1 Noz1: 11 11:46pm Dec 22,2021

Wall Thickness tb2 = max(trb2, tr16b) = 4.5000 mm.
 Wall Thickness per table UG-45 tb3 = 9.2200 mm.

Determine Nozzle Thickness candidate [tb]:
 = min[tb3, max(tb1,tb2)]
 = min[9.22, max(7.945, 4.5)]
 = 7.9450 mm.

Minimum Wall Thickness of Nozzle Necks [tUG-45]:
 = max(ta, tb)
 = max(4.6287, 7.945)
 = 7.9450 mm.

Available Nozzle Neck Thickness = 9.6012 mm. --> OK

**Stresses on Nozzle due to External and Pressure Loads per the ASME
 B31.3 Piping Code (see 319.4.4 and 302.3.5):**

Sustained	:	55.9,	Allowable	:	117.9 N./mm ²	Passed
Expansion	:	0.0,	Allowable	:	238.8 N./mm ²	Passed
Occasional	:	13.0,	Allowable	:	156.8 N./mm ²	Passed
Shear	:	24.1,	Allowable	:	82.5 N./mm ²	Passed

Note : The number of cycles on this nozzle was assumed to be 7000 or less for the determination of the expansion stress allowable.

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for the Nozzle (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for Reinforcement pad, Curve: B

Govrn. thk, tg = 9.601, tr = 1.629, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.247, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-104 °C

Shell to Pad Weld Junction at Pad OD, min(Curve:B, Curve:D)

Govrn. thk, tg = 10.0, tr = 4.945, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.706, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-45 °C

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

Nozzle-Shell/Head Weld (UCS-66(a)(b)), Curve: D

Govrn. thk, tg = 9.601, tr = 1.629, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.247$, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Min Metal Temp. at Required thickness (UCS 66.1) -104 °C

Governing MDMT of the Nozzle : -104 °C
 Governing MDMT of the Reinforcement Pad : -45 °C
 Governing MDMT of all the sub-joints of this Junction : -45 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -86 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = 23.00/51.10 = 0.450

Note:
 Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: S1

Intermediate Calc. for nozzle/shell Welds Tmin 6.6012 mm.
 Intermediate Calc. for pad/shell Welds TminPad 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	4.6208 = 0.7 * tmin.	5.6560 = 0.7 * Wo mm.
Pad Weld	3.5000 = 0.5 * TminPad	5.6560 = 0.7 * Wp mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:
 $= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv)$
 $= \max(0, (7.763 - 3.1474 + 2 * 6.6012 * 0.855 * (1.0 * 7.0 - 4.945)) 138)$
 $= 66.84 \text{ kN}$

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:
 $= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv$
 $= (1.488 + 12.1725 + 1.1851 - 0.0 * 0.86) * 138$
 $= 204.70 \text{ kN}$

Weld Load [W2]:
 $= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv$
 $= (1.488 + 0.0 + 0.5472 + (0.7902)) * 138$
 $= 38.96 \text{ kN}$

Weld Load [W3]:
 $= (A2+A3+A4+A5+(2*tn*t*fr1))*S$
 $= (1.488 + 0.0 + 1.1851 + 12.1725 + (0.7902)) * 138$
 $= 215.60 \text{ kN}$

Strength of Connection Elements for Failure Path Analysis

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 Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * D_{lo} * W_o * 0.49 * S_{nw}$$

$$= (3.1416/2.0) * 168.275 * 8.0 * 0.49 * 118$$

$$= 122. \text{ kN}$$

Shear, Pad Element Weld [Spew]:

$$= (\pi/2) * DP * WP * 0.49 * SEW$$

$$= (3.1416/2.0) * 290.0 * 8.0 * 0.49 * 138$$

$$= 246. \text{ kN}$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (D_{lr} + D_{lo}) / 4) * (Thk - Can) * 0.7 * S_n$$

$$= (3.1416 * 80.8369) * (9.6012 - 3.0) * 0.7 * 118$$

$$= 138. \text{ kN}$$

Tension, Pad Groove Weld [Tpgw]:

$$= (\pi/2) * D_{lo} * W_{gpn} * 0.74 * S_{eg}$$

$$= (3.1416/2) * 168.275 * 10.0 * 0.74 * 138$$

$$= 270. \text{ kN}$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * D_{lo} * (W_{gnvi} - Cas) * 0.74 * S_{ng}$$

$$= (3.1416/2.0) * 168.275 * (10.0 - 3.0) * 0.74 * 138$$

$$= 189. \text{ kN}$$

Strength of Failure Paths:

$$PATH11 = (SPEW + SNW) = (246 + 138) = 385 \text{ kN}$$

$$PATH22 = (Sonw + Tpgw + Tngw + Sinw)$$

$$= (122 + 270 + 189 + 0) = 581 \text{ kN}$$

$$PATH33 = (Spew + Tngw + Sinw)$$

$$= (246 + 189 + 0) = 435 \text{ kN}$$

Summary of Failure Path Calculations:

Path 1-1 = 384 kN , must exceed W = 66 kN or W1 = 204 kN
 Path 2-2 = 580 kN , must exceed W = 66 kN or W2 = 38 kN
 Path 3-3 = 435 kN , must exceed W = 66 kN or W3 = 215 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 12.4512 mm.

The Cut Length for this Nozzle is, Drop + Ho + H + T : 222.4512 mm.

Input Echo, WRC107/537 Item 1, Description: S1 :

Diameter Basis for Vessel	Vbasis	ID	
Cylindrical or Spherical Vessel	Cylsph	Cylindrical	
Internal Corrosion Allowance	Cas	3.0000	mm.
Vessel Diameter	Dv	581.000	mm.
Vessel Thickness	Tv	10.000	mm.
Design Temperature	T1	120.0	°C
Vessel Material		SA-516 70	
Vessel UNS Number		K02700	
Vessel Cold S.I. Allowable	Smc	137.90	N./mm ²
Vessel Hot S.I. Allowable	Smh	137.90	N./mm ²

Note:

Using 2 * Yield for Discontinuity Stress Allowable (Div 2, 4.1.6.3), Sps.

Make sure that material properties at this temperature are not time-dependent for Material: SA-516 70

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

Attachment Type	Type	Round	
Diameter Basis for Nozzle	Nbasis	OD	
Corrosion Allowance for Nozzle	Can	3.0000	mm.
Nozzle Diameter	Dn	168.275	mm.
Nozzle Thickness	Tn	9.601	mm.
Nozzle Material		SA-333 6	
Nozzle UNS Number		K03006	
Nozzle Cold S.I. Allowable	SNmc	117.90	N./mm ²
Nozzle Hot S.I. Allowable	SNmh	117.90	N./mm ²
Thickness of Reinforcing Pad	Tpad	10.000	mm.
Diameter of Reinforcing Pad	Dpad	290.000	mm.
Design Internal Pressure	Dp	23.000	bars
Include Pressure Thrust		No	

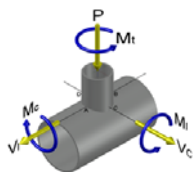
External Forces and Moments in WRC 107/537 Convention:

Radial Load	(SUS)	P	6.0	kN
Longitudinal Shear	(SUS)	Vl	6.0	kN
Circumferential Shear	(SUS)	Vc	6.0	kN
Circumferential Moment	(SUS)	Mc	3800.0	N-m
Longitudinal Moment	(SUS)	Ml	3800.0	N-m
Torsional Moment	(SUS)	Mt	4700.0	N-m

Use Interactive Control		No
WRC107 Version	Version	March 1979
Include Pressure Stress Indices per Div. 2		No
Compute Pressure Stress per WRC-368		No
Local Loads applied at end of Nozzle/Attachment		No

Note:

WRC Bulletin 537 provides equations for the dimensionless curves found in bulletin 107. As noted in the foreword to bulletin 537, "537 is equivalent to WRC 107". Where 107 is printed in the results below, "537" can be interchanged with "107".



Stress Attenuation Diameter (for Insert Plates) per WRC 297:

$$\begin{aligned}
 &= \text{NozzleOD} + 2 * 1.65 * \text{sqrt}(\text{Rmean}(t - ca)) \\
 &= 168.275 + 2 * 1.65 * \text{sqrt}(297.0 (10.0 - 3.0)) \\
 &= 318.742 \text{ mm.}
 \end{aligned}$$

WRC 107 Stress Calculation for SUSTained loads:

Radial Load	P	6.0	kN
Circumferential Shear	VC	6.0	kN
Longitudinal Shear	VL	6.0	kN
Circumferential Moment	MC	3800.0	N-m
Longitudinal Moment	ML	3800.0	N-m
Torsional Moment	MT	4700.0	N-m

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

Dimensionless Parameters used : Gamma = 17.76

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.244	4C	2.697	(A,B)
N(PHI) / (P/Rm)	0.244	3C	1.952	(C,D)
M(PHI) / (P)	0.244	2C1	0.045	(A,B)
M(PHI) / (P)	0.244	1C	0.074	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.244	3A	0.734	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.244	1A	0.085	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.244	3B	1.941	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.244	1B	0.032	(A,B,C,D)
N(x) / (P/Rm)	0.244	3C	1.952	(A,B)
N(x) / (P/Rm)	0.244	4C	2.697	(C,D)
M(x) / (P)	0.244	1C1	0.077	(A,B)
M(x) / (P)	0.244	2C	0.045	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.244	4A	1.267	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.244	2A	0.044	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.244	4B	0.709	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.244	2B	0.051	(A,B,C,D)

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction (N./mm^2)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-3.2	-3.2	-3.2	-3.2	-2.3	-2.3	-2.3	-2.3
Circ. Bend. P		-5.6	5.6	-5.6	5.6	-9.2	9.2	-9.2	9.2
Circ. Memb. MC		0.0	0.0	0.0	0.0	-7.4	-7.4	7.4	7.4
Circ. Memb. ML		0.0	0.0	0.0	0.0	-91.0	91.0	91.0	-91.0
Circ. Memb. ML		-19.5	-19.5	19.5	19.5	0.0	0.0	0.0	0.0
Circ. Bend. ML		-33.9	33.9	33.9	-33.9	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-62.2	16.8	44.6	-11.9	-109.9	90.5	86.9	-76.7
Long. Memb. P		-2.3	-2.3	-2.3	-2.3	-3.2	-3.2	-3.2	-3.2
Long. Bend. P		-9.6	9.6	-9.6	9.6	-5.6	5.6	-5.6	5.6
Long. Memb. MC		0.0	0.0	0.0	0.0	-12.7	-12.7	12.7	12.7
Long. Bend. MC		0.0	0.0	0.0	0.0	-47.2	47.2	47.2	-47.2
Long. Memb. ML		-7.1	-7.1	7.1	7.1	0.0	0.0	0.0	0.0
Long. Bend. ML		-55.1	55.1	55.1	-55.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-74.1	55.2	50.4	-40.7	-68.7	37.0	51.2	-32.1
Shear VC		1.3	1.3	-1.3	-1.3	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.3	-1.3	1.3	1.3
Shear MT		6.2	6.2	6.2	6.2	6.2	6.2	6.2	6.2
Tot. Shear		7.5	7.5	4.9	4.9	4.9	4.9	7.5	7.5
Str. Int.		77.7	56.7	53.2	41.5	110.4	91.0	88.4	77.9

Dimensionless Parameters used : Gamma = 42.43

Dimensionless Loads for Cylindrical Shells at Pad edge:

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 Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.427	4C	3.340	(A,B)
N(PHI) / (P/Rm)	0.427	3C	1.296	(C,D)
M(PHI) / (P)	0.427	2C1	0.008	(A,B)
M(PHI) / (P)	0.427	1C !	0.063	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.427	3A	1.071	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.427	1A	0.061	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.427	3B	1.851	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.427	1B	0.006	(A,B,C,D)
N(x) / (P/Rm)	0.427	3C	1.296	(A,B)
N(x) / (P/Rm)	0.427	4C	3.340	(C,D)
M(x) / (P)	0.427	1C1	0.019	(A,B)
M(x) / (P)	0.427	2C !	0.032	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.427	4A	4.057	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.427	2A	0.024	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.427	4B	1.123	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.427	2B	0.010	(A,B,C,D)

Note - The ! mark next to the figure name denotes curve value exceeded.

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-9.6	-9.6	-9.6	-9.6	-3.7	-3.7	-3.7	-3.7
Circ. Bend. P		-5.6	5.6	-5.6	5.6	-46.4	46.4	-46.4	46.4
Circ. Memb. MC		0.0	0.0	0.0	0.0	-15.4	-15.4	15.4	15.4
Circ. Memb. MC		0.0	0.0	0.0	0.0	-222.1	222.1	222.1	-222.1
Circ. Memb. ML		-26.7	-26.7	26.7	26.7	0.0	0.0	0.0	0.0
Circ. Bend. ML		-22.7	22.7	22.7	-22.7	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-64.6	-8.0	34.1	-0.1	-287.6	249.3	187.3	-164.0
Long. Memb. P		-3.7	-3.7	-3.7	-3.7	-9.6	-9.6	-9.6	-9.6
Long. Bend. P		-13.7	13.7	-13.7	13.7	-23.9	23.9	-23.9	23.9
Long. Memb. MC		0.0	0.0	0.0	0.0	-58.4	-58.4	58.4	58.4
Long. Bend. MC		0.0	0.0	0.0	0.0	-89.1	89.1	89.1	-89.1
Long. Memb. ML		-16.2	-16.2	16.2	16.2	0.0	0.0	0.0	0.0
Long. Bend. ML		-38.1	38.1	38.1	-38.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-71.7	31.9	36.8	-11.9	-181.1	44.9	114.1	-16.5
Shear VC		1.9	1.9	-1.9	-1.9	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.9	-1.9	1.9	1.9
Shear MT		5.1	5.1	5.1	5.1	5.1	5.1	5.1	5.1
Tot. Shear		7.0	7.0	3.2	3.2	3.2	3.2	7.0	7.0
Str. Int.		76.0	42.3	38.9	13.4	287.7	249.4	188.0	164.3

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Calcs.: S1 Noz1: 11 11:46pm Dec 22,2021

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		38.6	40.9	38.6	40.9	38.6	40.9	38.6	40.9
Circ. Pl (SUS)		-22.7	-22.7	16.4	16.4	-9.7	-9.7	5.1	5.1
Circ. Q (SUS)		-39.5	39.5	28.2	-28.2	-100.2	100.2	81.8	-81.8
Long. Pm (SUS)		19.3	19.3	19.3	19.3	19.3	19.3	19.3	19.3
Long. Pl (SUS)		-9.4	-9.4	4.8	4.8	-15.9	-15.9	9.6	9.6
Long. Q (SUS)		-64.6	64.6	45.5	-45.5	-52.8	52.8	41.6	-41.6
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.3	1.3	-1.3	-1.3	-1.3	-1.3	1.3	1.3
Shear Q (SUS)		6.2	6.2	6.2	6.2	6.2	6.2	6.2	6.2
Pm (SUS)		38.6	40.9	38.6	40.9	38.6	40.9	38.6	40.9
Pm+Pl (SUS)		16.2	18.4	55.0	57.3	29.0	31.3	43.8	46.1
Pm+Pl+Q (Total)		56.5	77.4	84.8	51.3	72.3	131.8	126.5	38.0

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	40.89	137.90	Passed
Pm+Pl (SUS)	57.30	206.85	Passed
Pm+Pl+Q (TOTAL)	131.75	413.70	Passed

Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		95.3	97.6	95.3	97.6	95.3	97.6	95.3	97.6
Circ. Pl (SUS)		-36.3	-36.3	17.0	17.0	-19.2	-19.2	11.7	11.7
Circ. Q (SUS)		-28.3	28.3	17.1	-17.1	-268.5	268.5	175.7	-175.7
Long. Pm (SUS)		47.7	47.7	47.7	47.7	47.7	47.7	47.7	47.7
Long. Pl (SUS)		-19.9	-19.9	12.4	12.4	-68.1	-68.1	48.8	48.8
Long. Q (SUS)		-51.8	51.8	24.4	-24.4	-113.0	113.0	65.3	-65.3
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.9	1.9	-1.9	-1.9	-1.9	-1.9	1.9	1.9
Shear Q (SUS)		5.1	5.1	5.1	5.1	5.1	5.1	5.1	5.1
Pm (SUS)		95.3	97.6	95.3	97.6	95.3	97.6	95.3	97.6
Pm+Pl (SUS)		59.1	61.4	112.4	114.7	96.6	98.9	107.3	109.6

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Calcs.: S1 Nozl: 11 11:46pm Dec 22,2021

 Pm+Pl+Q (Total)| 56.5| 93.2| 129.7| 97.7| 192.5| 347.0| 283.0| 98.5|

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	97.61	137.90	Passed
Pm+Pl (SUS)	114.68	206.85	Passed
Pm+Pl+Q (TOTAL)	346.96	413.70	Passed

*Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.*

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S3 Nozl: 12 11:46pm Dec 22,2021

Input, Nozzle Desc: S3 From: 40

Pressure for Reinforcement Calculations	P	23.034	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	5880.0005	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		759.18	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

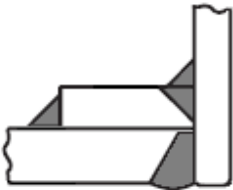
Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-333 6	
Material UNS Number		K03006	
Material Specification/Type	Smls. & wld. pipe		
Allowable Stress at Temperature	Sn	117.90	N./mm ²
Allowable Stress At Ambient	Sna	117.90	N./mm ²
Diameter Basis (for tr calc only)		OD	
Layout Angle		270.00	deg
Diameter		2.0000	in.
Size and Thickness Basis		Minimum	
Nominal Thickness	tn	160	
Flange Material		SA-350 LF2	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Pad Material		SA-516 70	
Pad Allowable Stress at Temperature	Sp	137.90	N./mm ²
Pad Allowable Stress At Ambient	Spa	137.90	N./mm ²
Diameter of Pad along vessel surface	Dp	170.0000	mm.
Thickness of Pad	te	10.0000	mm.
Weld leg size between Pad and Shell	Wp	8.0000	mm.
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm.
Reinforcing Pad Width		54.8375	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S3 Noz1: 12 11:46pm Dec 22,2021

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle With Pad, no Inside projection

Reinforcement CALCULATION, Description: S3

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Outside Diameter Used in Calculation	2.375 in.
Actual Thickness Used in Calculation	0.301 in.

Nozzle input data check completed without errors.

Reqd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]

$$= (P \cdot R) / (S_v \cdot E - 0.6 \cdot P) \text{ per UG-27 (c)(1)}$$

$$= (23.03 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.03)$$

$$= 4.9524 \text{ mm.}$$

Reqd thk per UG-37(a) of Nozzle Wall, Trn [Int. Press]

$$= (P \cdot R_o) / (S_n \cdot E + 0.4 \cdot P) \text{ per Appendix 1-1 (a)(1)}$$

$$= (23.03 \cdot 30.1625) / (118 \cdot 1.0 + 0.4 \cdot 23.03)$$

$$= 0.5847 \text{ mm.}$$

Required Nozzle thickness under External Pressure per UG-28 : 0.3261 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit)	D1	102.0684	mm.
Parallel to Vessel Wall, opening length	d	51.0342	mm.
Normal to Vessel Wall (Thickness Limit), pad side Tlwp		17.5000	mm.

Note: The Pad diameter is greater than the Diameter Limit. The excess will not be considered.

Weld Strength Reduction Factor [fr1]:

$$= \min(1, S_n / S_v)$$

$$= \min(1, 117.9 / 137.9)$$

$$= 0.855$$

Weld Strength Reduction Factor [fr2]:

$$= \min(1, S_n / S_v)$$

$$= \min(1, 117.9 / 137.9)$$

$$= 0.855$$

Weld Strength Reduction Factor [fr4]:

$$= \min(1, S_p / S_v)$$

$$= \min(1, 137.9 / 137.9)$$

$$= 1.000$$

Weld Strength Reduction Factor [fr3]:

$$= \min(fr2, fr4)$$

$$= \min(0.855, 1.0)$$

$$= 0.855$$

Results of Nozzle Reinforcement Area Calculations: (cm^2)

AREA AVAILABLE, A1 to A5		Design	External	Mapnc
Area Required	Ar	2.594	1.344	NA
Area in Shell	A1	1.017	0.928	NA
Area in Nozzle Wall	A2	1.215	1.293	NA
Area in Inward Nozzle	A3	0.000	0.000	NA
Area in Welds	A41+A42+A43	0.545	0.545	NA
Area in Element	A5	4.174	4.174	NA
TOTAL AREA AVAILABLE	Atot	6.952	6.940	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Insufficient.
 The area available with the given pad is Sufficient.

SELECTION OF POSSIBLE REINFORCING PADS:	Diameter	Thickness
Based on given Pad Thickness:	60.3250	10.0000 mm.
Based on given Pad Diameter:	170.0000	0.0000 mm.
Based on Shell or Nozzle Thickness:	60.3250	7.6454 mm.

Area Required [A]:

$$= (d * tr * F + 2 * tn * tr * F * (1 - fr1)) UG-37(c)$$

$$= (51.0342 * 4.9524 * 1.0 + 2 * 4.6454 * 4.9524 * 1.0 * (1 - 0.86))$$

$$= 2.594 \text{ cm}^2$$

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

$$= d(E1 * t - F * tr) - 2 * tn(E1 * t - F * tr) * (1 - fr1)$$

$$= 51.034(1.0 * 7.0 - 1.0 * 4.952) - 2 * 4.645$$

$$(1.0 * 7.0 - 1.0 * 4.9524) * (1 - 0.855)$$

$$= 1.017 \text{ cm}^2$$

Area Available in Nozzle Wall Projecting Outward [A2]:

$$= (2 * Tlwp) * (tn - trn) * fr2$$

$$= (2 * 17.5) * (4.65 - 0.58) * 0.855$$

$$= 1.215 \text{ cm}^2$$

Area Available in Welds [A41 + A42 + A43]:

$$= (Wo^2 - Ar Lost) * Fr3 + ((Wi - can / 0.707)^2 - Ar Lost) * fr2 + Wp^2 * fr4$$

$$= (0.6375) * 0.86 + (0.0) * 0.86 + 0.0^2 * 1.0$$

$$= 0.545 \text{ cm}^2$$

Area Available in Element [A5]:

$$= (\min(Dp, DL) - (Nozzle OD)) * (\min(tp, Tlwp, te)) * fr4$$

$$= (102.0684 - 60.325) * 10.0 * 1.0$$

$$= 4.174 \text{ cm}^2$$

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures ta = 3.5847 mm.
 Wall Thickness per UG16(b), tr16b = 4.5000 mm.

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Nozzle Calcs.: S3 Noz1: 12 11:46pm Dec 22,2021

Wall Thickness, shell/head, internal pressure trb1 = 7.9524 mm.
 Wall Thickness tb1 = max(trb1, tr16b) = 7.9524 mm.
 Wall Thickness tb2 = max(trb2, tr16b) = 4.5000 mm.
 Wall Thickness per table UG-45 tb3 = 6.4200 mm.

Determine Nozzle Thickness candidate [tb]:

= min[tb3, max(tb1,tb2)]
 = min[6.42, max(7.9524, 4.5)]
 = 6.4200 mm.

Minimum Wall Thickness of Nozzle Necks [tUG-45]:

= max(ta, tb)
 = max(3.5847, 6.42)
 = 6.4200 mm.

Available Nozzle Neck Thickness = 7.6454 mm. --> OK

Stresses on Nozzle due to External and Pressure Loads per the ASME

B31.3 Piping Code (see 319.4.4 and 302.3.5):

Sustained	: 69.1,	Allowable	: 117.9 N./mm ²	Passed
Expansion	: 0.0,	Allowable	: 225.6 N./mm ²	Passed
Occasional	: 5.8,	Allowable	: 156.8 N./mm ²	Passed
Shear	: 28.6,	Allowable	: 82.5 N./mm ²	Passed

Note : The number of cycles on this nozzle was assumed to be 7000 or less for the determination of the expansion stress allowable.

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for the Nozzle (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for Reinforcement pad, Curve: B

Govrn. thk, tg = 7.645, tr = 0.585, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.126, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-104 °C

Shell to Pad Weld Junction at Pad OD, min(Curve:B, Curve:D)

Govrn. thk, tg = 10.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.707, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
--	--------

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S3 Noz1: 12 11:46pm Dec 22,2021

Min Metal Temp. at Required thickness (UCS 66.1) -45 °C

Nozzle-Shell/Head Weld (UCS-66(a)1(b)), Curve: D

Govrn. thk, tg = 7.645, tr = 0.585, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.126, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Min Metal Temp. at Required thickness (UCS 66.1) -104 °C

Governing MDMT of the Nozzle : -104 °C
 Governing MDMT of the Reinforcement Pad : -45 °C
 Governing MDMT of all the sub-joints of this Junction : -45 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -85 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = 23.03/51.10 = 0.451

Note:
 Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: S3

Intermediate Calc. for nozzle/shell Welds Tmin 4.6454 mm.
 Intermediate Calc. for pad/shell Welds TminPad 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	3.2518 = 0.7 * tmin.	5.6560 = 0.7 * Wo mm.
Pad Weld	3.5000 = 0.5*TminPad	5.6560 = 0.7 * Wp mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:
 $= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv)$
 $= \max(0, (2.5941 - 1.0174 + 2 * 4.6454 * 0.855 * (1.0 * 7.0 - 4.9524)) 138)$
 $= 23.98 \text{ kN}$

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:
 $= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv$
 $= (1.2152 + 4.1743 + 0.5451 - 0.0 * 0.86) * 138$
 $= 81.83 \text{ kN}$

Weld Load [W2]:
 $= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv$
 $= (1.2152 + 0.0 + 0.5472 + (0.5561)) * 138$
 $= 31.97 \text{ kN}$

Weld Load [W3]:
 $= (A2+A3+A4+A5+(2*tn*t*fr1))*S$
 $= (1.2152 + 0.0 + 0.5451 + 4.1743 + (0.5561)) * 138$
 $= 89.50 \text{ kN}$

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 Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S3 Noz1: 12 11:46pm Dec 22,2021

Strength of Connection Elements for Failure Path Analysis

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * Dlo * Wo * 0.49 * Snw$$

$$= (3.1416/2.0) * 60.325 * 8.0 * 0.49 * 118$$

$$= 44. kN$$

Shear, Pad Element Weld [Spew]:

$$= (\pi/2) * DP * WP * 0.49 * SEW$$

$$= (3.1416/2.0) * 170.0 * 8.0 * 0.49 * 138$$

$$= 144. kN$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (Dlr + Dlo)/4) * (Thk - Can) * 0.7 * Sn$$

$$= (3.1416 * 27.8398) * (7.6454 - 3.0) * 0.7 * 118$$

$$= 34. kN$$

Tension, Pad Groove Weld [Tpgw]:

$$= (\pi/2) * Dlo * Wgpn * 0.74 * Seg$$

$$= (3.1416/2) * 60.325 * 10.0 * 0.74 * 138$$

$$= 97. kN$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * Dlo * (Wgnvi-Cas) * 0.74 * Sng$$

$$= (3.1416/2.0) * 60.325 * (10.0 - 3.0) * 0.74 * 138$$

$$= 68. kN$$

Strength of Failure Paths:

$$PATH11 = (SPEW + SNW) = (144 + 34) = 178 kN$$

$$PATH22 = (Sonw + Tpgw + Tngw + Sinw)$$

$$= (44 + 97 + 68 + 0) = 208 kN$$

$$PATH33 = (Spew + Tngw + Sinw)$$

$$= (144 + 68 + 0) = 212 kN$$

Summary of Failure Path Calculations:

Path 1-1 = 177 kN , must exceed W = 23 kN or W1 = 81 kN
 Path 2-2 = 208 kN , must exceed W = 23 kN or W2 = 31 kN
 Path 3-3 = 212 kN , must exceed W = 23 kN or W3 = 89 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 1.5701 mm.
 The Cut Length for this Nozzle is, Drop + Ho + H + T : 211.5701 mm.

Input Echo, WRC107/537 Item 1, Description: S3 :

Diameter Basis for Vessel	Vbasis	ID	
Cylindrical or Spherical Vessel	Cylsph	Cylindrical	
Internal Corrosion Allowance	Cas	3.0000	mm.
Vessel Diameter	Dv	581.000	mm.
Vessel Thickness	Tv	10.000	mm.
Design Temperature	T1	120.0	°C
Vessel Material		SA-516 70	
Vessel UNS Number		K02700	
Vessel Cold S.I. Allowable	Smc	137.90	N./mm^2
Vessel Hot S.I. Allowable	Smh	137.90	N./mm^2

Note:
 Using 2 * Yield for Discontinuity Stress Allowable (Div 2, 4.1.6.3), Sps.

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: S3 Nozl: 12 11:46pm Dec 22,2021

Make sure that material properties at this temperature are not time-dependent for Material: SA-516 70

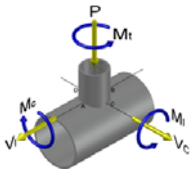
Attachment Type	Type	Round	
Diameter Basis for Nozzle	Nbasis	OD	
Corrosion Allowance for Nozzle	Can	3.0000	mm.
Nozzle Diameter	Dn	60.325	mm.
Nozzle Thickness	Tn	7.645	mm.
Nozzle Material		SA-333 6	
Nozzle UNS Number		K03006	
Nozzle Cold S.I. Allowable	SNmc	117.90	N./mm^2
Nozzle Hot S.I. Allowable	SNmh	117.90	N./mm^2
Thickness of Reinforcing Pad	Tpad	10.000	mm.
Diameter of Reinforcing Pad	Dpad	170.000	mm.
Design Internal Pressure	Dp	23.034	bars
Include Pressure Thrust		No	

External Forces and Moments in WRC 107/537 Convention:

Radial Load (SUS)	P	2.0	kN
Longitudinal Shear (SUS)	Vl	2.0	kN
Circumferential Shear (SUS)	Vc	2.0	kN
Circumferential Moment (SUS)	Mc	500.0	N-m
Longitudinal Moment (SUS)	Ml	400.0	N-m
Torsional Moment (SUS)	Mt	400.0	N-m

Use Interactive Control		No
WRC107 Version	Version	March 1979
Include Pressure Stress Indices per Div. 2		No
Compute Pressure Stress per WRC-368		No
Local Loads applied at end of Nozzle/Attachment		No

Note:
 WRC Bulletin 537 provides equations for the dimensionless curves found in bulletin 107. As noted in the foreword to bulletin 537, "537 is equivalent to WRC 107". Where 107 is printed in the results below, "537" can be interchanged with "107".



Stress Attenuation Diameter (for Insert Plates) per WRC 297:

$$\begin{aligned}
 &= \text{NozzleOD} + 2 * 1.65 * \text{sqrt}(\text{Rmean}(t - ca)) \\
 &= 60.325 + 2 * 1.65 * \text{sqrt}(297.0 (10.0 - 3.0)) \\
 &= 210.792 \text{ mm.}
 \end{aligned}$$

WRC 107 Stress Calculation for SUSTained loads:

Radial Load	P	2.0	kN
Circumferential Shear	VC	2.0	kN
Longitudinal Shear	VL	2.0	kN
Circumferential Moment	MC	500.0	N-m
Longitudinal Moment	ML	400.0	N-m

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Calcs.: S3 Nozl: 12 11:46pm Dec 22,2021

Torsional Moment MT 400.0 N-m

Dimensionless Parameters used : Gamma = 17.76

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.087	4C	3.397	(A,B)
N(PHI) / (P/Rm)	0.087	3C	3.228	(C,D)
M(PHI) / (P)	0.087	2C1	0.138	(A,B)
M(PHI) / (P)	0.087	1C	0.172	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.087	3A	0.320	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.087	1A	0.102	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.087	3B	1.184	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.087	1B	0.056	(A,B,C,D)
N(x) / (P/Rm)	0.087	3C	3.228	(A,B)
N(x) / (P/Rm)	0.087	4C	3.397	(C,D)
M(x) / (P)	0.087	1C1	0.176	(A,B)
M(x) / (P)	0.087	2C	0.138	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.087	4A	0.416	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.087	2A	0.061	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.087	4B	0.306	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.087	2B	0.092	(A,B,C,D)

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction (N./mm^2)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl

Circ. Memb. P		-1.3	-1.3	-1.3	-1.3	-1.3	-1.3	-1.3	-1.3
Circ. Bend. P		-5.7	5.7	-5.7	5.7	-7.2	7.2	-7.2	7.2
Circ. Memb. MC		0.0	0.0	0.0	0.0	-1.2	-1.2	1.2	1.2
Circ. Memb. MC		0.0	0.0	0.0	0.0	-40.3	40.3	40.3	-40.3
Circ. Memb. ML		-3.5	-3.5	3.5	3.5	0.0	0.0	0.0	0.0
Circ. Bend. ML		-17.7	17.7	17.7	-17.7	0.0	0.0	0.0	0.0

Tot. Circ. Str.		-28.2	18.6	14.1	-9.8	-49.9	45.0	33.1	-33.2

Long. Memb. P		-1.3	-1.3	-1.3	-1.3	-1.3	-1.3	-1.3	-1.3
Long. Bend. P		-7.3	7.3	-7.3	7.3	-5.7	5.7	-5.7	5.7
Long. Memb. MC		0.0	0.0	0.0	0.0	-1.5	-1.5	1.5	1.5
Long. Bend. MC		0.0	0.0	0.0	0.0	-23.8	23.8	23.8	-23.8
Long. Memb. ML		-0.9	-0.9	0.9	0.9	0.0	0.0	0.0	0.0
Long. Bend. ML		-28.9	28.9	28.9	-28.9	0.0	0.0	0.0	0.0

Tot. Long. Str.		-38.4	34.1	21.2	-21.9	-32.4	26.7	18.3	-17.9

Shear VC		1.2	1.2	-1.2	-1.2	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.2	-1.2	1.2	1.2
Shear MT		4.1	4.1	4.1	4.1	4.1	4.1	4.1	4.1

Tot. Shear		5.4	5.4	2.9	2.9	2.9	2.9	5.4	5.4

Str. Int.		40.7	35.7	22.2	22.6	50.4	45.5	34.8	34.9

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 Nozzle Calcs.: S3 Noz1: 12 11:46pm Dec 22,2021

Dimensionless Parameters used : Gamma = 42.43

Dimensionless Loads for Cylindrical Shells at Pad edge:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.250	4C	5.214	(A,B)
N(PHI) / (P/Rm)	0.250	3C	2.813	(C,D)
M(PHI) / (P)	0.250	2C1	0.022	(A,B)
M(PHI) / (P)	0.250	1C !	0.063	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.250	3A	1.624	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.250	1A	0.068	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.250	3B	3.634	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.250	1B	0.017	(A,B,C,D)
N(x) / (P/Rm)	0.250	3C	2.813	(A,B)
N(x) / (P/Rm)	0.250	4C	5.214	(C,D)
M(x) / (P)	0.250	1C1	0.049	(A,B)
M(x) / (P)	0.250	2C !	0.032	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.250	4A	3.734	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.250	2A	0.030	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.250	4B	1.677	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.250	2B	0.026	(A,B,C,D)

Note - The ! mark next to the figure name denotes curve value exceeded.

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-5.0	-5.0	-5.0	-5.0	-2.7	-2.7	-2.7	-2.7
Circ. Bend. P		-5.4	5.4	-5.4	5.4	-15.5	15.5	-15.5	15.5
Circ. Memb. MC		0.0	0.0	0.0	0.0	-5.2	-5.2	5.2	5.2
Circ. Memb. MC		0.0	0.0	0.0	0.0	-56.2	56.2	56.2	-56.2
Circ. Memb. ML		-9.4	-9.4	9.4	9.4	0.0	0.0	0.0	0.0
Circ. Bend. ML		-11.0	11.0	11.0	-11.0	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-30.8	2.0	9.9	-1.2	-79.6	63.7	43.2	-38.2
Long. Memb. P		-2.7	-2.7	-2.7	-2.7	-5.0	-5.0	-5.0	-5.0
Long. Bend. P		-11.9	11.9	-11.9	11.9	-8.0	8.0	-8.0	8.0
Long. Memb. MC		0.0	0.0	0.0	0.0	-12.1	-12.1	12.1	12.1
Long. Bend. MC		0.0	0.0	0.0	0.0	-24.5	24.5	24.5	-24.5
Long. Memb. ML		-4.3	-4.3	4.3	4.3	0.0	0.0	0.0	0.0
Long. Bend. ML		-17.1	17.1	17.1	-17.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-36.0	22.0	6.8	-3.5	-49.5	15.3	23.6	-9.5
Shear VC		1.1	1.1	-1.1	-1.1	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.1	-1.1	1.1	1.1
Shear MT		1.3	1.3	1.3	1.3	1.3	1.3	1.3	1.3
Tot. Shear		2.3	2.3	0.2	0.2	0.2	0.2	2.3	2.3
Str. Int.		36.9	22.2	10.0	3.5	79.6	63.7	43.5	38.3

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		38.7	41.0	38.7	41.0	38.7	41.0	38.7	41.0
Circ. Pl (SUS)		-4.8	-4.8	2.2	2.2	-2.4	-2.4	-0.1	-0.1
Circ. Q (SUS)		-23.4	23.4	12.0	-12.0	-47.5	47.5	33.1	-33.1
Long. Pm (SUS)		19.3	19.3	19.3	19.3	19.3	19.3	19.3	19.3
Long. Pl (SUS)		-2.2	-2.2	-0.4	-0.4	-2.9	-2.9	0.2	0.2
Long. Q (SUS)		-36.2	36.2	21.6	-21.6	-29.5	29.5	18.1	-18.1
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.2	1.2	-1.2	-1.2	-1.2	-1.2	1.2	1.2
Shear Q (SUS)		4.1	4.1	4.1	4.1	4.1	4.1	4.1	4.1
Pm (SUS)		38.7	41.0	38.7	41.0	38.7	41.0	38.7	41.0
Pm+Pl (SUS)		33.9	36.2	40.9	43.2	36.3	38.6	38.7	41.0
Pm+Pl+Q (Total)		31.4	62.6	53.4	34.2	15.2	86.2	72.5	12.4

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	40.95	137.90	Passed
Pm+Pl (SUS)	43.19	206.85	Passed
Pm+Pl+Q (TOTAL)	86.17	413.70	Passed

Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		95.4	97.8	95.4	97.8	95.4	97.8	95.4	97.8
Circ. Pl (SUS)		-14.4	-14.4	4.4	4.4	-8.0	-8.0	2.5	2.5
Circ. Q (SUS)		-16.4	16.4	5.6	-5.6	-71.6	71.6	40.7	-40.7
Long. Pm (SUS)		47.7	47.7	47.7	47.7	47.7	47.7	47.7	47.7
Long. Pl (SUS)		-7.0	-7.0	1.6	1.6	-17.1	-17.1	7.1	7.1
Long. Q (SUS)		-29.0	29.0	5.1	-5.1	-32.4	32.4	16.5	-16.5
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.1	1.1	-1.1	-1.1	-1.1	-1.1	1.1	1.1
Shear Q (SUS)		1.3	1.3	1.3	1.3	1.3	1.3	1.3	1.3
Pm (SUS)		95.4	97.8	95.4	97.8	95.4	97.8	95.4	97.8

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Pm+Pl (SUS)	81.1	83.4	99.9	102.2	87.5	89.8	98.0	100.3
Pm+Pl+Q (Total)	64.8	99.9	105.4	96.6	17.7	161.4	138.8	59.8

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	97.75	137.90	Passed
Pm+Pl (SUS)	102.15	206.85	Passed
Pm+Pl+Q (TOTAL)	161.44	413.70	Passed

*Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.*

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 Nozzle Calcs.: T2 Nozl: 13 11:46pm Dec 22,2021

Input, Nozzle Desc: T2 From: 60

Pressure for Reinforcement Calculations	P	23.000	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	505.4167	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		6849.35	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

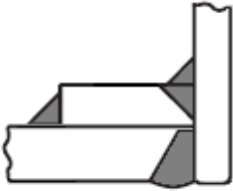
Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-333 6	
Material UNS Number		K03006	
Material Specification/Type	Smls. & wld. pipe		
Allowable Stress at Temperature	Sn	117.90	N./mm ²
Allowable Stress At Ambient	Sna	117.90	N./mm ²
Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		6.0000	in.
Size and Thickness Basis		Minimum	
Nominal Thickness	tn	80	
Flange Material		SA-350 LF2	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Pad Material		SA-516 70	
Pad Allowable Stress at Temperature	Sp	137.90	N./mm ²
Pad Allowable Stress At Ambient	Spa	137.90	N./mm ²
Diameter of Pad along vessel surface	Dp	290.0000	mm.
Thickness of Pad	te	10.0000	mm.
Weld leg size between Pad and Shell	Wp	8.0000	mm.
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm.
Reinforcing Pad Width		60.8625	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

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 Nozzle Calcs.: T2 Noz1: 13 11:46pm Dec 22,2021

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle With Pad, no Inside projection

Reinforcement CALCULATION, Description: T2

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Outside Diameter Used in Calculation	6.625 in.
Actual Thickness Used in Calculation	0.378 in.

Nozzle input data check completed without errors.

Reqd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]
 $= (P \cdot R) / (S_v \cdot E - 0.6 \cdot P)$ per UG-27 (c)(1)
 $= (23.0 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.0)$
 $= 4.9450 \text{ mm.}$

Reqd thk per UG-37(a) of Nozzle Wall, Trn [Int. Press]
 $= (P \cdot R_o) / (S_n \cdot E + 0.4 \cdot P)$ per Appendix 1-1 (a)(1)
 $= (23.0 \cdot 84.1375) / (118 \cdot 1.0 + 0.4 \cdot 23.0)$
 $= 1.6287 \text{ mm.}$

Required Nozzle thickness under External Pressure per UG-28 : 0.5944 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit)	D1	310.1452	mm.
Parallel to Vessel Wall, opening length	d	155.0726	mm.
Normal to Vessel Wall (Thickness Limit), pad side Tlwp		17.5000	mm.

Weld Strength Reduction Factor [fr1]:
 $= \min(1, S_n / S_v)$
 $= \min(1, 117.9 / 137.9)$
 $= 0.855$

Weld Strength Reduction Factor [fr2]:
 $= \min(1, S_n / S_v)$
 $= \min(1, 117.9 / 137.9)$
 $= 0.855$

Weld Strength Reduction Factor [fr4]:
 $= \min(1, S_p / S_v)$
 $= \min(1, 137.9 / 137.9)$
 $= 1.000$

Weld Strength Reduction Factor [fr3]:

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 Nozzle Calcs.: T2 Nozl: 13 11:46pm Dec 22,2021

= min(fr2, fr4)
 = min(0.855, 1.0)
 = 0.855

Results of Nozzle Reinforcement Area Calculations: (cm²)

AREA AVAILABLE, A1 to A5	Design	External	Mapnc
Area Required Ar	7.763	1.439	NA
Area in Shell A1	3.147	7.913	NA
Area in Nozzle Wall A2	1.488	1.798	NA
Area in Inward Nozzle A3	0.000	0.000	NA
Area in Welds A41+A42+A43	1.185	1.185	NA
Area in Element A5	12.172	12.172	NA
TOTAL AREA AVAILABLE Atot	17.993	23.068	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Insufficient.
 The area available with the given pad is Sufficient.

SELECTION OF POSSIBLE REINFORCING PADS: Diameter Thickness
 Based on given Pad Thickness: 187.6997 10.0000 mm.
 Based on given Pad Diameter: 290.0000 1.5958 mm.
 Based on Shell or Nozzle Thickness: 188.5066 9.6012 mm.

Area Required [A]:

= (d * tr*F + 2 * tn * tr*F * (1-fr1)) UG-37(c)
 = (155.0726*4.945*1.0+2*6.6012*4.945*1.0*(1-0.86))
 = 7.763 cm²

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

= d(E1*t - F*tr) - 2 * tn(E1*t - F*tr) * (1 - fr1)
 = 155.073(1.0 * 7.0 - 1.0 * 4.945) - 2 * 6.601
 (1.0 * 7.0 - 1.0 * 4.945) * (1 - 0.855)
 = 3.147 cm²

Area Available in Nozzle Wall Projecting Outward [A2]:

= (2 * Tlwp) * (tn - trn) * fr2
 = (2 * 17.5) * (6.6 - 1.63) * 0.855
 = 1.488 cm²

Area Available in Welds [A41 + A42 + A43]:

= (Wo² - Ar Lost)*Fr3+((Wi-can/0.707)² - Ar Lost)*fr2 + Wp²*fr4
 = (0.6375) * 0.86 + (0.0) * 0.86 + 203.2² * 1.0
 = 1.185 cm²

Area Available in Element [A5]:

= (min(Dp,DL)-(Nozzle OD))*(min(tp,Tlwp,te)) * fr4
 = (290.0 - 168.275) * 10.0 * 1.0
 = 12.172 cm²

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures ta = 4.6287 mm.
 Wall Thickness per UG16(b), tr16b = 4.5000 mm.
 Wall Thickness, shell/head, internal pressure trb1 = 7.9450 mm.
 Wall Thickness tbt = max(trb1, tr16b) = 7.9450 mm.

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 Nozzle Calcs.: T2 Noz1: 13 11:46pm Dec 22,2021

Wall Thickness tb2 = max(trb2, tr16b) = 4.5000 mm.
 Wall Thickness per table UG-45 tb3 = 9.2200 mm.

Determine Nozzle Thickness candidate [tb]:
 = min[tb3, max(tb1,tb2)]
 = min[9.22, max(7.945, 4.5)]
 = 7.9450 mm.

Minimum Wall Thickness of Nozzle Necks [tUG-45]:
 = max(ta, tb)
 = max(4.6287, 7.945)
 = 7.9450 mm.

Available Nozzle Neck Thickness = 9.6012 mm. --> OK

**Stresses on Nozzle due to External and Pressure Loads per the ASME
 B31.3 Piping Code (see 319.4.4 and 302.3.5):**

Sustained	:	55.9,	Allowable	:	117.9 N./mm ²	Passed
Expansion	:	0.0,	Allowable	:	238.8 N./mm ²	Passed
Occasional	:	13.0,	Allowable	:	156.8 N./mm ²	Passed
Shear	:	24.1,	Allowable	:	82.5 N./mm ²	Passed

Note : The number of cycles on this nozzle was assumed to be 7000 or less for the determination of the expansion stress allowable.

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for the Nozzle (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification	-46 °C
Calculated Minimum Design Metal Temperature	-104 °C

Nozzle Neck to Pad Weld for Reinforcement pad, Curve: B

Govrn. thk, tg = 9.601, tr = 1.629, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.247, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-104 °C

Shell to Pad Weld Junction at Pad OD, min(Curve:B, Curve:D)

Govrn. thk, tg = 10.0, tr = 4.945, c = 3.0 mm., E* = 1.0
 Thickness Ratio = tr * (E*)/(tg - c) = 0.706, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve B	-29 °C
Min Metal Temp. at Required thickness (UCS 66.1)	-45 °C

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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Calcs.: T2 Nozl: 13 11:46pm Dec 22,2021

Nozzle-Shell/Head Weld (UCS-66(a)(1)(b)), Curve: D

Govrn. thk, tg = 9.601, tr = 1.629, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.247$, Temp. Reduction = 78 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Min Metal Temp. at Required thickness (UCS 66.1) -104 °C

Governing MDMT of the Nozzle : -104 °C
 Governing MDMT of the Reinforcement Pad : -45 °C
 Governing MDMT of all the sub-joints of this Junction : -45 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -86 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = $23.00 / 51.10 = 0.450$

Note:
 Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: T2

Intermediate Calc. for nozzle/shell Welds Tmin 6.6012 mm.
 Intermediate Calc. for pad/shell Welds TminPad 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	$4.6208 = 0.7 * t_{min}$	$5.6560 = 0.7 * W_o$ mm.
Pad Weld	$3.5000 = 0.5 * T_{minPad}$	$5.6560 = 0.7 * W_p$ mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:
 $= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv)$
 $= \max(0, (7.763 - 3.1474 + 2 * 6.6012 * 0.855 * (1.0 * 7.0 - 4.945)) 138)$
 $= 66.84$ kN

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:
 $= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv$
 $= (1.488 + 12.1725 + 1.1851 - 0.0 * 0.86) * 138$
 $= 204.70$ kN

Weld Load [W2]:
 $= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv$
 $= (1.488 + 0.0 + 0.5472 + (0.7902)) * 138$
 $= 38.96$ kN

Weld Load [W3]:
 $= (A2+A3+A4+A5+(2*tn*t*fr1))*S$
 $= (1.488 + 0.0 + 1.1851 + 12.1725 + (0.7902)) * 138$
 $= 215.60$ kN

Strength of Connection Elements for Failure Path Analysis

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 Nozzle Calcs.: T2 Noz1: 13 11:46pm Dec 22,2021

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * D_{lo} * W_o * 0.49 * S_{nw}$$

$$= (3.1416/2.0) * 168.275 * 8.0 * 0.49 * 118$$

$$= 122. \text{ kN}$$

Shear, Pad Element Weld [Spew]:

$$= (\pi/2) * D_P * W_P * 0.49 * S_{EW}$$

$$= (3.1416/2.0) * 290.0 * 8.0 * 0.49 * 138$$

$$= 246. \text{ kN}$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (D_{lr} + D_{lo}) / 4) * (Thk - Can) * 0.7 * S_n$$

$$= (3.1416 * 80.8369) * (9.6012 - 3.0) * 0.7 * 118$$

$$= 138. \text{ kN}$$

Tension, Pad Groove Weld [Tpgw]:

$$= (\pi/2) * D_{lo} * W_{gpn} * 0.74 * S_{eg}$$

$$= (3.1416/2) * 168.275 * 10.0 * 0.74 * 138$$

$$= 270. \text{ kN}$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * D_{lo} * (W_{gnvi-Cas}) * 0.74 * S_{ng}$$

$$= (3.1416/2.0) * 168.275 * (10.0 - 3.0) * 0.74 * 138$$

$$= 189. \text{ kN}$$

Strength of Failure Paths:

$$\text{PATH11} = (\text{SPEW} + \text{SNW}) = (246 + 138) = 385 \text{ kN}$$

$$\text{PATH22} = (\text{Sonw} + \text{Tpgw} + \text{Tngw} + \text{Sinw})$$

$$= (122 + 270 + 189 + 0) = 581 \text{ kN}$$

$$\text{PATH33} = (\text{Spew} + \text{Tngw} + \text{Sinw})$$

$$= (246 + 189 + 0) = 435 \text{ kN}$$

Summary of Failure Path Calculations:

Path 1-1 = 384 kN , must exceed W = 66 kN or W1 = 204 kN
 Path 2-2 = 580 kN , must exceed W = 66 kN or W2 = 38 kN
 Path 3-3 = 435 kN , must exceed W = 66 kN or W3 = 215 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 12.4512 mm.

The Cut Length for this Nozzle is, Drop + Ho + H + T : 222.4512 mm.

Input Echo, WRC107/537 Item 1, Description: T2 :

Diameter Basis for Vessel	Vbasis	ID
Cylindrical or Spherical Vessel	Cylsph	Cylindrical
Internal Corrosion Allowance	Cas	3.0000 mm.
Vessel Diameter	Dv	581.000 mm.
Vessel Thickness	Tv	10.000 mm.
Design Temperature	T1	120.0 °C
Vessel Material		SA-516 70
Vessel UNS Number		K02700
Vessel Cold S.I. Allowable	Smc	137.90 N./mm ²
Vessel Hot S.I. Allowable	Smh	137.90 N./mm ²

Note:

Using 2 * Yield for Discontinuity Stress Allowable (Div 2, 4.1.6.3), Sps.
 Make sure that material properties at this temperature are not
 time-dependent for Material: SA-516 70

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Nozzle Calcs.: T2 Nozl: 13 11:46pm Dec 22,2021

Attachment Type	Type	Round	
Diameter Basis for Nozzle	Nbasis	OD	
Corrosion Allowance for Nozzle	Can	3.0000	mm.
Nozzle Diameter	Dn	168.275	mm.
Nozzle Thickness	Tn	9.601	mm.
Nozzle Material		SA-333 6	
Nozzle UNS Number		K03006	
Nozzle Cold S.I. Allowable	SNmc	117.90	N./mm ²
Nozzle Hot S.I. Allowable	SNmh	117.90	N./mm ²
Thickness of Reinforcing Pad	Tpad	10.000	mm.
Diameter of Reinforcing Pad	Dpad	290.000	mm.
Design Internal Pressure	Dp	23.000	bars
Include Pressure Thrust		No	

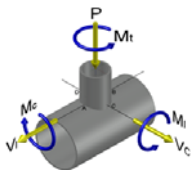
External Forces and Moments in WRC 107/537 Convention:

Radial Load	(SUS)	P	6.0	kN
Longitudinal Shear	(SUS)	Vl	6.0	kN
Circumferential Shear	(SUS)	Vc	6.0	kN
Circumferential Moment	(SUS)	Mc	3800.0	N-m
Longitudinal Moment	(SUS)	Ml	3800.0	N-m
Torsional Moment	(SUS)	Mt	4700.0	N-m

Use Interactive Control		No
WRC107 Version	Version	March 1979
Include Pressure Stress Indices per Div. 2		No
Compute Pressure Stress per WRC-368		No
Local Loads applied at end of Nozzle/Attachment		No

Note:

WRC Bulletin 537 provides equations for the dimensionless curves found in bulletin 107. As noted in the foreword to bulletin 537, "537 is equivalent to WRC 107". Where 107 is printed in the results below, "537" can be interchanged with "107".



Stress Attenuation Diameter (for Insert Plates) per WRC 297:

$$\begin{aligned}
 &= \text{NozzleOD} + 2 * 1.65 * \text{sqrt}(\text{Rmean}(t - ca)) \\
 &= 168.275 + 2 * 1.65 * \text{sqrt}(297.0 (10.0 - 3.0)) \\
 &= 318.742 \text{ mm.}
 \end{aligned}$$

WRC 107 Stress Calculation for SUSTained loads:

Radial Load	P	6.0	kN
Circumferential Shear	VC	6.0	kN
Longitudinal Shear	VL	6.0	kN
Circumferential Moment	MC	3800.0	N-m
Longitudinal Moment	ML	3800.0	N-m
Torsional Moment	MT	4700.0	N-m

Dimensionless Parameters used : Gamma = 17.76

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.244	4C	2.697	(A,B)
N(PHI) / (P/Rm)	0.244	3C	1.952	(C,D)
M(PHI) / (P)	0.244	2C1	0.045	(A,B)
M(PHI) / (P)	0.244	1C	0.074	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.244	3A	0.734	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.244	1A	0.085	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.244	3B	1.941	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.244	1B	0.032	(A,B,C,D)
N(x) / (P/Rm)	0.244	3C	1.952	(A,B)
N(x) / (P/Rm)	0.244	4C	2.697	(C,D)
M(x) / (P)	0.244	1C1	0.077	(A,B)
M(x) / (P)	0.244	2C	0.045	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.244	4A	1.267	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.244	2A	0.044	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.244	4B	0.709	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.244	2B	0.051	(A,B,C,D)

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction (N./mm^2)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		-3.2	-3.2	-3.2	-3.2	-2.3	-2.3	-2.3	-2.3
Circ. Bend. P		-5.6	5.6	-5.6	5.6	-9.2	9.2	-9.2	9.2
Circ. Memb. MC		0.0	0.0	0.0	0.0	-7.4	-7.4	7.4	7.4
Circ. Memb. ML		0.0	0.0	0.0	0.0	-91.0	91.0	91.0	-91.0
Circ. Memb. ML		-19.5	-19.5	19.5	19.5	0.0	0.0	0.0	0.0
Circ. Bend. ML		-33.9	33.9	33.9	-33.9	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-62.2	16.8	44.6	-11.9	-109.9	90.5	86.9	-76.7
Long. Memb. P		-2.3	-2.3	-2.3	-2.3	-3.2	-3.2	-3.2	-3.2
Long. Bend. P		-9.6	9.6	-9.6	9.6	-5.6	5.6	-5.6	5.6
Long. Memb. MC		0.0	0.0	0.0	0.0	-12.7	-12.7	12.7	12.7
Long. Bend. MC		0.0	0.0	0.0	0.0	-47.2	47.2	47.2	-47.2
Long. Memb. ML		-7.1	-7.1	7.1	7.1	0.0	0.0	0.0	0.0
Long. Bend. ML		-55.1	55.1	55.1	-55.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-74.1	55.2	50.4	-40.7	-68.7	37.0	51.2	-32.1
Shear VC		1.3	1.3	-1.3	-1.3	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.3	-1.3	1.3	1.3
Shear MT		6.2	6.2	6.2	6.2	6.2	6.2	6.2	6.2
Tot. Shear		7.5	7.5	4.9	4.9	4.9	4.9	7.5	7.5
Str. Int.		77.7	56.7	53.2	41.5	110.4	91.0	88.4	77.9

Dimensionless Parameters used : Gamma = 42.43

Dimensionless Loads for Cylindrical Shells at Pad edge:

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Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.427	4C	3.340	(A,B)
N(PHI) / (P/Rm)	0.427	3C	1.296	(C,D)
M(PHI) / (P)	0.427	2C1	0.008	(A,B)
M(PHI) / (P)	0.427	1C !	0.063	(C,D)
N(PHI) / (MC/(Rm**2 * Beta))	0.427	3A	1.071	(A,B,C,D)
M(PHI) / (MC/(Rm * Beta))	0.427	1A	0.061	(A,B,C,D)
N(PHI) / (ML/(Rm**2 * Beta))	0.427	3B	1.851	(A,B,C,D)
M(PHI) / (ML/(Rm * Beta))	0.427	1B	0.006	(A,B,C,D)
N(x) / (P/Rm)	0.427	3C	1.296	(A,B)
N(x) / (P/Rm)	0.427	4C	3.340	(C,D)
M(x) / (P)	0.427	1C1	0.019	(A,B)
M(x) / (P)	0.427	2C !	0.032	(C,D)
N(x) / (MC/(Rm**2 * Beta))	0.427	4A	4.057	(A,B,C,D)
M(x) / (MC/(Rm * Beta))	0.427	2A	0.024	(A,B,C,D)
N(x) / (ML/(Rm**2 * Beta))	0.427	4B	1.123	(A,B,C,D)
M(x) / (ML/(Rm * Beta))	0.427	2B	0.010	(A,B,C,D)

Note - The ! mark next to the figure name denotes curve value exceeded.

Stress Concentration Factors: Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl

Circ. Memb. P		-9.6	-9.6	-9.6	-9.6	-3.7	-3.7	-3.7	-3.7
Circ. Bend. P		-5.6	5.6	-5.6	5.6	-46.4	46.4	-46.4	46.4
Circ. Memb. MC		0.0	0.0	0.0	0.0	-15.4	-15.4	15.4	15.4
Circ. Memb. MC		0.0	0.0	0.0	0.0	-222.1	222.1	222.1	-222.1
Circ. Memb. ML		-26.7	-26.7	26.7	26.7	0.0	0.0	0.0	0.0
Circ. Bend. ML		-22.7	22.7	22.7	-22.7	0.0	0.0	0.0	0.0
Tot. Circ. Str.		-64.6	-8.0	34.1	-0.1	-287.6	249.3	187.3	-164.0

Long. Memb. P		-3.7	-3.7	-3.7	-3.7	-9.6	-9.6	-9.6	-9.6
Long. Bend. P		-13.7	13.7	-13.7	13.7	-23.9	23.9	-23.9	23.9
Long. Memb. MC		0.0	0.0	0.0	0.0	-58.4	-58.4	58.4	58.4
Long. Bend. MC		0.0	0.0	0.0	0.0	-89.1	89.1	89.1	-89.1
Long. Memb. ML		-16.2	-16.2	16.2	16.2	0.0	0.0	0.0	0.0
Long. Bend. ML		-38.1	38.1	38.1	-38.1	0.0	0.0	0.0	0.0
Tot. Long. Str.		-71.7	31.9	36.8	-11.9	-181.1	44.9	114.1	-16.5

Shear VC		1.9	1.9	-1.9	-1.9	0.0	0.0	0.0	0.0
Shear VL		0.0	0.0	0.0	0.0	-1.9	-1.9	1.9	1.9
Shear MT		5.1	5.1	5.1	5.1	5.1	5.1	5.1	5.1
Tot. Shear		7.0	7.0	3.2	3.2	3.2	3.2	7.0	7.0

Str. Int.		76.0	42.3	38.9	13.4	287.7	249.4	188.0	164.3

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 Nozzle Calcs.: T2 Nozl: 13 11:46pm Dec 22,2021

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		38.6	40.9	38.6	40.9	38.6	40.9	38.6	40.9
Circ. Pl (SUS)		-22.7	-22.7	16.4	16.4	-9.7	-9.7	5.1	5.1
Circ. Q (SUS)		-39.5	39.5	28.2	-28.2	-100.2	100.2	81.8	-81.8
Long. Pm (SUS)		19.3	19.3	19.3	19.3	19.3	19.3	19.3	19.3
Long. Pl (SUS)		-9.4	-9.4	4.8	4.8	-15.9	-15.9	9.6	9.6
Long. Q (SUS)		-64.6	64.6	45.5	-45.5	-52.8	52.8	41.6	-41.6
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.3	1.3	-1.3	-1.3	-1.3	-1.3	1.3	1.3
Shear Q (SUS)		6.2	6.2	6.2	6.2	6.2	6.2	6.2	6.2
Pm (SUS)		38.6	40.9	38.6	40.9	38.6	40.9	38.6	40.9
Pm+Pl (SUS)		16.2	18.4	55.0	57.3	29.0	31.3	43.8	46.1
Pm+Pl+Q (Total)		56.5	77.4	84.8	51.3	72.3	131.8	126.5	38.0

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	40.89	137.90	Passed
Pm+Pl (SUS)	57.30	206.85	Passed
Pm+Pl+Q (TOTAL)	131.75	413.70	Passed

Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge (N./mm²)

Type of Stress	Load	Stress Intensity Values at							
		Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)		95.3	97.6	95.3	97.6	95.3	97.6	95.3	97.6
Circ. Pl (SUS)		-36.3	-36.3	17.0	17.0	-19.2	-19.2	11.7	11.7
Circ. Q (SUS)		-28.3	28.3	17.1	-17.1	-268.5	268.5	175.7	-175.7
Long. Pm (SUS)		47.7	47.7	47.7	47.7	47.7	47.7	47.7	47.7
Long. Pl (SUS)		-19.9	-19.9	12.4	12.4	-68.1	-68.1	48.8	48.8
Long. Q (SUS)		-51.8	51.8	24.4	-24.4	-113.0	113.0	65.3	-65.3
Shear Pm (SUS)		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Shear Pl (SUS)		1.9	1.9	-1.9	-1.9	-1.9	-1.9	1.9	1.9
Shear Q (SUS)		5.1	5.1	5.1	5.1	5.1	5.1	5.1	5.1
Pm (SUS)		95.3	97.6	95.3	97.6	95.3	97.6	95.3	97.6
Pm+Pl (SUS)		59.1	61.4	112.4	114.7	96.6	98.9	107.3	109.6

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Pm+Pl+Q (Total)| 56.5| 93.2| 129.7| 97.7| 192.5| 347.0| 283.0| 98.5|

Vessel Stress Summation Comparison (N./mm²):

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
Pm (SUS)	97.61	137.90	Passed
Pm+Pl (SUS)	114.68	206.85	Passed
Pm+Pl+Q (TOTAL)	346.96	413.70	Passed

*Because only sustained loads were specified, the Pm+Pl+Q allowable was 3 * Smh.*

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 Nozzle Calcs.: T3 Nozl: 14 11:46pm Dec 22,2021

Input, Nozzle Desc: T3 From: 60

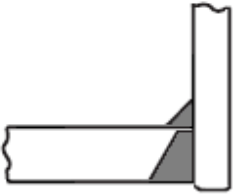
Pressure for Reinforcement Calculations	P	23.034	bars
Temperature for Internal Pressure	Temp	120	°C
Design External Pressure	Pext	1.10	bars
Temperature for External Pressure	Tempex	120	°C
Shell Material [Normalized]		SA-516 70	
Shell Allowable Stress at Temperature	Sv	137.90	N./mm ²
Shell Allowable Stress At Ambient	Sva	137.90	N./mm ²
Inside Diameter of Cylindrical Shell	D	581.00	mm.
Design Length of Section	L	505.4167	mm.
Shell Finished (Minimum) Thickness	t	10.0000	mm.
Shell Internal Corrosion Allowance	c	3.0000	mm.
Shell External Corrosion Allowance	co	0.0000	mm.
Distance from Bottom/Left Tangent		6852.35	mm.
User Entered Minimum Design Metal Temperature		-45.00	°C

Type of Element Connected to the Shell : Nozzle

Material [Impact Tested]		SA-350 LF2	
Material UNS Number		K03011	
Material Specification/Type		Forgings	
Allowable Stress at Temperature	Sn	137.90	N./mm ²
Allowable Stress At Ambient	Sna	137.90	N./mm ²
Diameter Basis (for tr calc only)		ID	
Layout Angle		270.00	deg
Diameter		1.0000	in.
Size and Thickness Basis		Actual	
Actual Thickness	tn	14.3000	mm.
Flange Material		SA-350 LF2	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm.
Joint Efficiency of Shell Seam at Nozzle	E1	1.00	
Joint Efficiency of Nozzle Neck	En	1.00	
Outside Projection	ho	200.0000	mm.
Weld leg size between Nozzle and Pad/Shell	Wo	8.0000	mm.
Groove weld depth between Nozzle and Vessel	Wgnv	10.0000	mm.
Inside Projection	h	0.0000	mm.
Weld leg size, Inside Element to Shell	Wi	0.0000	mm.
Class of attached Flange		300	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle No Pad, no Inside projection

Reinforcement CALCULATION, Description: T3

ASME Code, Section VIII, Div. 1, 2017, UG-37 to UG-45

Actual Inside Diameter Used in Calculation 1.000 in.
 Actual Thickness Used in Calculation 0.563 in.

Nozzle input data check completed without errors.

Reqd thk per UG-37(a) of Cylindrical Shell, Tr [Int. Press]
 $= (P \cdot R) / (S_v \cdot E - 0.6 \cdot P)$ per UG-27 (c)(1)
 $= (23.03 \cdot 293.5) / (138 \cdot 1.0 - 0.6 \cdot 23.03)$
 $= 4.9524$ mm.

Reqd thk per App. 1 of Nozzle Wall, Trn [Int. Press]
 $= R \cdot (\exp([P / (S_n \cdot E)] - 1) - 1)$ per Appendix 1-2 (a)(1)
 $= 15.7 \cdot (\exp([23.03 / (137.9 \cdot 1.0)] - 1) - 1)$
 $= 0.2645$ mm.

Required Nozzle thickness under External Pressure per UG-28 : 0.3056 mm.

UG-40, Limits of Reinforcement : [Internal Pressure]

Parallel to Vessel Wall (Diameter Limit) D1 68.0000 mm.
 Parallel to Vessel Wall Rn+tn+t 34.0000 mm.
 Normal to Vessel Wall (Thickness Limit), no pad Tlnp 17.5000 mm.

Weld Strength Reduction Factor [fr1]:
 $= \min(1, S_n / S_v)$
 $= \min(1, 137.9 / 137.9)$
 $= 1.000$

Weld Strength Reduction Factor [fr2]:
 $= \min(1, S_n / S_v)$
 $= \min(1, 137.9 / 137.9)$
 $= 1.000$

Weld Strength Reduction Factor [fr3]:
 $= \min(fr2, fr4)$
 $= \min(1.0, 1.0)$
 $= 1.000$

Results of Nozzle Reinforcement Area Calculations: (cm^2)

AREA AVAILABLE, A1 to A5		Design	External	Mapnc
Area Required	Ar	1.555	0.288	NA
Area in Shell	A1	0.749	1.891	NA

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Area in Nozzle Wall	A2	3.862	3.848	NA
Area in Inward Nozzle	A3	0.000	0.000	NA
Area in Welds	A41+A42+A43	0.630	0.630	NA
Area in Element	A5	0.000	0.000	NA
TOTAL AREA AVAILABLE	Atot	5.242	6.369	NA

The Internal Pressure Case Governs the Analysis.

Nozzle Angle Used in Area Calculations 90.00 Degs.

The area available without a pad is Sufficient.

Area Required [A]:

$$\begin{aligned}
 &= (d * tr * F + 2 * tn * tr * F * (1 - fr1)) \text{ UG-37(c)} \\
 &= (31.4 * 4.9524 * 1.0 + 2 * 11.3 * 4.9524 * 1.0 * (1 - 1.0)) \\
 &= 1.555 \text{ cm}^2
 \end{aligned}$$

Reinforcement Areas per Figure UG-37.1

Area Available in Shell [A1]:

$$\begin{aligned}
 &= d(E1 * t - F * tr) - 2 * tn(E1 * t - F * tr) * (1 - fr1) \\
 &= 36.6(1.0 * 7.0 - 1.0 * 4.952) - 2 * 11.3 \\
 &\quad (1.0 * 7.0 - 1.0 * 4.9524) * (1 - 1.0) \\
 &= 0.749 \text{ cm}^2
 \end{aligned}$$

Area Available in Nozzle Projecting Outward [A2]:

$$\begin{aligned}
 &= (2 * tlnp)(tn - trn) fr2 \\
 &= (2 * 17.5)(11.3 - 0.26) 1.0 \\
 &= 3.862 \text{ cm}^2
 \end{aligned}$$

Area Available in Inward Weld + Outward Weld [A41 + A43]:

$$\begin{aligned}
 &= (Wo^2 - \text{Area Lost}) * fr2 + ((Wi - can / 0.707)^2 - \text{Area Lost}) * fr2 \\
 &= (8.0^2 - 0.01) * 1.0 + (0.0^2 - 0.0) * 1.0 \\
 &= 0.630 \text{ cm}^2
 \end{aligned}$$

UG-45 Minimum Nozzle Neck Thickness Requirement: [Int. Press.]

Wall Thickness for Internal/External pressures	ta = 3.3056 mm.
Wall Thickness per UG16(b),	tr16b = 4.5000 mm.
Wall Thickness, shell/head, internal pressure	trb1 = 7.9524 mm.
Wall Thickness	tb1 = max(trb1, tr16b) = 7.9524 mm.
Wall Thickness	tb2 = max(trb2, tr16b) = 4.5000 mm.
Wall Thickness per table UG-45	tb3 = 6.4200 mm.

Determine Nozzle Thickness candidate [tb]:

$$\begin{aligned}
 &= \min[tb3, \max(tb1, tb2)] \\
 &= \min[6.42, \max(7.9524, 4.5)] \\
 &= 6.4200 \text{ mm.}
 \end{aligned}$$

Minimum Wall Thickness of Nozzle Necks [tUG-45]:

$$\begin{aligned}
 &= \max(ta, tb) \\
 &= \max(3.3056, 6.42) \\
 &= 6.4200 \text{ mm.}
 \end{aligned}$$

Available Nozzle Neck Thickness = 14.3000 mm. --> OK

Nozzle Junction Minimum Design Metal Temperature (MDMT) Calculations:

Nozzle Neck to Flange Weld (Impact tested) :

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

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 Nozzle Calcs.: T3 Nozl: 14 11:46pm Dec 22,2021

Impact Test Temperature provided per Specification -46 °C
 Calculated Minimum Design Metal Temperature -104 °C

Nozzle-Shell/Head Weld (UCS-66(a)1(b)), Curve: D

 Govrn. thk, tg = 10.0, tr = 4.952, c = 3.0 mm., E* = 1.0
 Thickness Ratio = $tr * (E^*) / (tg - c) = 0.707$, Temp. Reduction = 16 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -48 °C
 Governing MDMT of all the sub-joints of this Junction : -48 °C

ANSI Flange MDMT including Temperature reduction per UCS-66.1:

MDMT of ANSI B16.5/47 flange per Matl. Specification -46 °C
 Flange MDMT with Temp reduction per UCS-66(i)(2) -85 °C
 Flange MDMT with Temp reduction per UCS-66(i)(3) -104 °C

Where the Stress Reduction Ratio per UCS-66(i)(2) is :
 Design Pressure/Ambient Rating = $23.03/51.10 = 0.451$

Note:
 Using the min value from (i)(2) and (i)(3) above as the computed nozzle flange MDMT.

Weld Size Calculations, Description: T3

Intermediate Calc. for nozzle/shell Welds Tmin 7.0000 mm.

Results Per UW-16.1:

	Required Thickness	Actual Thickness
Nozzle Weld	$4.9000 = 0.7 * t_{min}$	$5.6560 = 0.7 * W_o$ mm.

Weld Strength and Weld Loads per UG-41.1, Sketch (a) or (b)

Weld Load [W]:
 $= \max(0, (A-A1+2*tn*fr1*(E1*t-tr))Sv)$
 $= \max(0, (1.5551 - 0.7494 + 2 * 11.3 * 1.0 * (1.0 * 7.0 - 4.9524))138)$
 $= 17.49$ kN

Note: F is always set to 1.0 throughout the calculation.

Weld Load [W1]:
 $= (A2+A5+A4-(Wi-Can/.707)^2*fr2)*Sv$
 $= (3.8624 + 0.0 + 0.63 - 0.0 * 1.0) * 138$
 $= 61.95$ kN

Weld Load [W2]:
 $= (A2 + A3 + A4 + (2 * tn * t * fr1)) * Sv$
 $= (3.8624 + 0.0 + 0.63 + (1.582)) * 138$
 $= 83.76$ kN

Weld Load [W3]:
 $= (A2+A3+A4+A5+(2*tn*t*fr1))*S$
 $= (3.8624 + 0.0 + 0.63 + 0.0 + (1.582)) * 138$
 $= 83.76$ kN

Strength of Connection Elements for Failure Path Analysis

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 Tag no:E-PK6101-3 ECONOMIZER
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 Nozzle Calcs.: T3 Nozl: 14 11:46pm Dec 22,2021

Shear, Outward Nozzle Weld [Sonw]:

$$= (\pi/2) * D_{lo} * W_o * 0.49 * S_{nw}$$

$$= (3.1416/2.0) * 54.0 * 8.0 * 0.49 * 138$$

$$= 46. \text{ kN}$$

Shear, Nozzle Wall [Snw]:

$$= (\pi * (D_{lr} + D_{lo}) / 4) * (Thk - Can) * 0.7 * S_n$$

$$= (3.1416 * 21.35) * (14.3 - 3.0) * 0.7 * 138$$

$$= 73. \text{ kN}$$

Tension, Shell Groove Weld [Tngw]:

$$= (\pi/2) * D_{lo} * (W_{gnvi-Cas}) * 0.74 * S_{ng}$$

$$= (3.1416/2.0) * 54.0 * (10.0 - 3.0) * 0.74 * 138$$

$$= 61. \text{ kN}$$

Strength of Failure Paths:

$$\text{PATH11} = (\text{SONW} + \text{SNW}) = (46 + 73) = 119 \text{ kN}$$

$$\text{PATH22} = (\text{Sonw} + \text{Tpgw} + \text{Tngw} + \text{Sinw})$$

$$= (46 + 0 + 61 + 0) = 106 \text{ kN}$$

$$\text{PATH33} = (\text{Sonw} + \text{Tngw} + \text{Sinw})$$

$$= (46 + 61 + 0) = 106 \text{ kN}$$

Summary of Failure Path Calculations:

Path 1-1 = 119 kN , must exceed W = 17 kN or W1 = 61 kN
 Path 2-2 = 106 kN , must exceed W = 17 kN or W2 = 83 kN
 Path 3-3 = 106 kN , must exceed W = 17 kN or W3 = 83 kN

Nozzle is O.K. for the External Pressure 1.100 bars

The Drop for this Nozzle is : 1.2575 mm.

The Cut Length for this Nozzle is, Drop + Ho + H + T : 211.2574 mm.

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Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Nozzle Schedule: Step: 21 11:46pm Dec 22,2021

Nozzle Schedule:

Flg	Nominal or	Schd	Flg	Nozzle	Wall	Reinforcing Pad	Cut		
Class	Actual	or FVC	Type	O/Dia	Thk	Diameter	Thk		
Description	Size	Type		in	mm.	mm.	mm.		
T4	0.750 in	Actual	SlipOn	1.872	14.250	210.97	300
T3	1.000 in	Actual	SlipOn	2.126	14.300	211.26	300
S3	2.000 in	160	WNF	2.375	8.738	170.00	10.00	211.57	300
T1	4.000 in	120	WNF	4.500	11.125	230.00	10.00	215.68	300
S2	6.000 in	80	WNF	6.625	10.973	290.00	10.00	222.45	300
S1	6.000 in	80	WNF	6.625	10.973	290.00	10.00	222.45	300
T2	6.000 in	80	WNF	6.625	10.973	290.00	10.00	222.45	300

General Notes for the above table:

The Cut Length is the Outside Projection + Inside Projection + Drop + In Plane Shell Thickness. This value does not include weld gaps, nor does it account for shrinkage.

In the case of Oblique Nozzles, the Outside Diameter must be increased. The Re-Pad WIDTH around the nozzle is calculated as follows:
 Width of Pad = (Pad Outside Dia. (per above) - Nozzle Outside Dia.)/2

For hub nozzles, the thickness and diameter shown are those of the smaller and thinner section.

Nozzle Material and Weld Fillet Leg Size Details (mm.):

Description	Material	Shl Grve Weld	Noz Shl/Pad Weld	Pad OD Weld	Pad Grve Weld	Inside Weld
T4	SA-350 LF2	10.000	8.000
T3	SA-350 LF2	10.000	8.000
S3	SA-333 6	10.000	8.000	8.000	10.000	...
T1	SA-333 6	10.000	8.000	8.000	10.000	...
S2	SA-333 6	10.000	8.000	8.000	10.000	...
S1	SA-333 6	10.000	8.000	8.000	10.000	...
T2	SA-333 6	10.000	8.000	8.000	10.000	...

Note: The Outside projections below do not include the flange thickness.

Nozzle Miscellaneous Data:

Description	Elev/Distance From Datum mm.	Layout Angle deg	Proj Outside mm.	Proj Inside mm.	Installed in Component
T4	203.000	90.0	200.00	0.00	CHANNEL 01

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Nozzle Schedule: Step: 21 11:46pm Dec 22,2021

T3	6802.351	270.0	200.00	0.00	CHANNEL 002
S3	709.175	270.0	200.00	0.00	SHELL
T1	203.000	270.0	200.00	0.00	CHANNEL 01
S2	759.175	90.0	200.00	0.00	SHELL
S1	6243.176	90.0	200.00	0.00	SHELL
T2	6799.351	90.0	200.00	0.00	CHANNEL 002

Weld Sizes for Slip On/Socket Weld Nozzle Flanges per UW-21:

Nozzle to Flange Fillet Weld Leg dimension [xmin]:
 = min(1.4 * tn, Hub Thickness)

The Nozzle Wall thicknesses shown below are in the corroded condition. Hubs are considered to be straight.

Description	Nominal or Actual Size	Schd or FVC Type	Flg Type	Noz. O/Dia in	Wall Thk mm.	Hub Thk mm.	Throat Thk mm.	xmin Thk mm.
T4	0.750 in	Actua	SlipOn	1.872	11.250	10.033	7.023	10.033
T3	1.000 in	Actua	SlipOn	2.126	11.300	9.652	6.756	9.652

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Input Echo, Tubesheet Number 1, Description: TUBE SHEET

Shell Data:

Main Shell Description: SHELL

Shell Maximum Design Pressure	Psd,max	23.00	bars
Shell Maximum Operating Pressure	Psox,max	23.00	bars
Shell Minimum Operating Pressure	Psox,min	0.00	bars
Shell Thickness	ts	10.0000	mm.
Shell Internal Corrosion Allowance	cas	3.0000	mm.
Shell External Corrosion Allowance	caext	0.0000	mm.
Inside Diameter of Shell	Ds	581.000	mm.
Shell Circumferential Joint Efficiency	Esw	1.000	
Shell Temperature for Internal Pressure	Ts	120.00	°C
Shell Material		SA-516 70	

Note:

Using 2 * Yield for Discontinuity Stress Allowable (UG-23(e)), Sps.
 Make sure that material properties at this temperature are not
 time-dependent for Material: SA-516 70

Shell Material UNS Number		K02700	
Shell Allowable Stress at Temperature	Ss	137.90	N./mm ²
Shell Allowable Stress at Ambient		137.90	N./mm ²

Channel Description: CHANNEL 01

Channel Type:		Cylinder	
Channel Maximum Design Pressure	Ptd,max	23.00	bars
Channel Maximum Operating Pressure	Ptox,max	23.00	bars
Channel Minimum Operating Pressure	Ptox,min	0.00	bars
Channel Thickness	tc	10.0000	mm.
Channel Corrosion Allowance	cac	3.0000	mm.
Inside Diameter of Channel	Dc	581.000	mm.
Channel Design Temperature	TEMPC	120.00	°C
Channel Material		SA-516 70	

Note:

Using 2 * Yield for Discontinuity Stress Allowable (UG-23(e)), Sps.
 Make sure that material properties at this temperature are not
 time-dependent for Material: SA-516 70

Channel Material UNS Number		K02700	
Channel Allowable Stress at Temperature	Sc	137.90	N./mm ²
Channel Allowable Stress at Ambient		137.90	N./mm ²

Tube Data:

Number of Tube Holes	Nt	241	
Tube Wall Thickness	et	2.1080	mm.
Tube Outside Diameter	D	25.4000	mm.
Total Straight Tube Length	Lt	6000.00	mm.
Straight Tube Length (bet. inner tubsht faces) L		5876.00	mm.
Design Temperature of the Tubes		120.00	°C
Tube Material		SA-334 6	
Tube Material UNS Number		K03006	
Is this a Welded Tube		No	
Tube Material Specification used	Smls. & wld. tube		
Tube Allowable Stress at Temperature		117.90	N./mm ²
Tube Allowable Stress At Ambient		117.90	N./mm ²
Tube Yield Stress At design Temperature	Syt	217.36	N./mm ²
Tube Pitch (Center to Center Spacing)	P	32.0000	mm.
Tube Layout Pattern		Triangular	

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

PV Elite 2019 SP1 Licensee: SPLM Licensed User

FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Fillet Weld Leg	af	1.5000	mm.
Groove Weld Leg	ag	1.5000	mm.
Tube-Tubesheet Joint Weld Type		Full Strength	
Method for Tube-Tubesheet Jt. Allow.		UW-20	
Tube-Tubesheet Joint Classification		b-1	
Radius to Outermost Tube Hole Center	ro	270.770	mm.
Largest Center-to-Center Tube Distance	Ul	40.0000	mm.
Length of Expanded Portion of Tube	ltx	50.0000	mm.
Tube-side pass partition groove depth	hg	5.0000	mm.
Area of Tube Layout	Ap	226.8	cm ²

Tubesheet Data:

Tubesheet TYPE: Fixed Tubesheet Exchanger, Conf B

Tubesheet Design Metal Temperature	T	120.00	°C
Tubesheet Material		SA-350 LF2	

Note:

Using 2 * Yield for Discontinuity Stress Allowable (UG-23(e)), Sps.
Make sure that material properties at this temperature are not
time-dependent for Material: SA-350 LF2

Tubesheet Material UNS Number		K03011	
Tubesheet Allowable Stress at Temperature	S	137.90	N./mm ²
Tubesheet Allowable Stress at Ambient	Tt	137.90	N./mm ²
Thickness of Tubesheet	h	62.0000	mm.
Tubesheet Corr. Allowance (Shell side)	Cats	3.0000	mm.
Tubesheet Corr. Allowance (Channel side)	Catc	3.0000	mm.
Tubesheet Outside Diameter	A	715.000	mm.

Additional Data for Stepped Tubesheets:

Is the Tubesheet Stepped?		YES	
Is the Tubesheet Flat on Tubeside?		NO	
Step 1 Diameter on the Tubeside	dt1	644.00	mm.
Step 1 Depth on the Tubeside	ht1	6.00	mm.
Step 2 Diameter on the Tubeside	dt2	578.00	mm.
Step 2 Depth on the Tubeside	ht2	5.00	mm.
Is the Tubesheet Flat on Shellside?		NO	
Step 1 Diameter on the Shellside	ds1	0.00	mm.
Step 1 Depth on the Shellside	hs1	0.00	mm.
Step 2 Diameter on the Shellside	ds2	578.00	mm.
Step 2 Depth on the Shellside	hs2	5.00	mm.
Calculated Tubesheet Diameter as per UHX-10(b)		644.00	mm.

Note: Tubesheet diameter is now: 644.000 mm. per UHX-10(b).

Area of the Untubed Lanes	AL	226.8	cm ²
---------------------------	----	-------	-----------------

Additional Data for Fixed/Floating Tubesheet Exchangers:

Unsupported Tube Span under consideration	l	712.000	mm.
Tube End condition corresponding to Span (l)	k	0.80	

Ignore Radial Thermal Exp. effects (UHX-13.8/14.6)		YES	
--	--	-----	--

Note: The Metal temperatures at the Rim are set to ambient (21 °C)

Tubesheet Metal Temp. at Rim	T'	21.11	°C
Shell Metal Temp. at Tubesheet	T'S	21.11	°C
Channel Metal Temp. at Tubesheet	T'C	21.11	°C

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Perform Differential Pressure Design			N
Run Multiple Load Cases			YES
Shell Side Min. Design Pressure	Psd,min	1.0342	bars
Channel Side Min. Design Pressure	Ptd,min	1.0314	bars
Mean Shell Metal Temp. along Shell len.	Tsm	28.10	°C
Mean Tube Metal Temp. along Tube length	Ttm	21.09	°C
Junction Stress Reduction option			Perform Plastic Calculation

Additional Data for Gasketed Tubesheets:

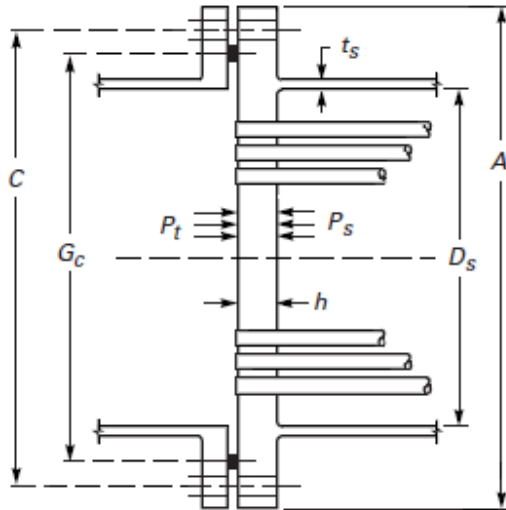
Tubesheet Gasket on which Side		Channel	
Flange Outside Diameter	A	715.000	mm.
Flange Inside Diameter	B	581.000	mm.
Flange Face Outside Diameter	Fod	644.000	mm.
Flange Face Inside Diameter	Fid	581.000	mm.
Gasket Outside Diameter	Go	641.000	mm.
Gasket Inside Diameter	Gi	611.000	mm.
Small end Hub thk.	g0	10.0000	mm.
Large end Hub thk.	g1	17.0000	mm.
Gasket Factor,	m	3.78	
Gasket Design Seating Stress	y	62.05	N./mm ²
Flange Facing Sketch		Code Sketch 1a	
Column for Gasket Seating		Code Column II	
Gasket Thickness	tg	3.0000	mm.
Full face Gasket Flange Option		Program Selects	
Length of Partition Gasket	lp	1078.000	mm.
Width of Partition Gasket	wp	6.0000	mm.
Partition Gasket Factor,	mPart	3.7500	
Partition Gasket Design Seating Stress	yPart	62.05	N./mm ²
Partition Gasket Facing Sketch		Code Sketch 1a	
Partition Gasket Column for Gasket Seating		Code Column II	

Bolting Information:

Diameter of Bolt Circle	C	673.000	mm.
Nominal Bolt Diameter	dB	19.0500	mm.
Type of Thread Series		UNC Thread Series	
Number of Bolts	n	32	

Tag no:E-PK6101-3 ECONOMIZER
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 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Tubesheet Integral With Shell and Gasketed With Channel, Extended as a Flange



Configuration b:

320 L7	Bolt Material		SA-
Bolt Allowable Stress At Temperature	Sb	172.38 N./mm ²	
Bolt Allowable Stress At Ambient	Sa	172.38 N./mm ²	
Weld between Flange and Shell/Channel		0.0000 mm.	
Tubesheet Integral with	Shell		
Tubesheet Extended as Flange	Yes		
Thickness of Extended Portion of Tubesheet	Tf	46.0000 mm.	
Is Bolt Load Transferred to the Tubesheet	Yes		
Is Exchanger in Creep range (skip EP, Use 3S for Sps)	NO		

ASME TubeSheet Results per Part UHX, 2017

Elasticity/Expansion Material Properties:

Shell - TE-1 Carbon & Low Alloy Steels, Group 1
 Shell - TM-1 Carbon Steels with C<= 0.3%

Th. Exp. Coeff. Metal Temp. along Len	28.1 °C	0.0000115945 /°C
Elastic Mod. at Design Temperature	120.0 °C	0.19691E+09 KPa.
Th. Exp. Coeff. Metal Temp. at Tubsht	21.1 °C	0.0000115190 /°C
Elastic Mod. at Metal Temp. along Len	28.1 °C	0.20230E+09 KPa.
Elastic Mod. at Ambient Temperature	21.1 °C	0.20270E+09 KPa.

Channel - TE-1 Carbon & Low Alloy Steels, Group 1
 Channel - TM-1 Carbon Steels with C<= 0.3%

Th. Exp. Coeff. Metal Temp. at Tubsht	21.1 °C	0.0000115190 /°C
Elastic Mod. at Design Temperature	120.0 °C	0.19691E+09 KPa.
Elastic Mod. at Ambient Temperature	21.1 °C	0.20270E+09 KPa.

Tubes - TE-1 Carbon & Low Alloy Steels, Group 1

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Tubes - TM-1 Carbon Steels with C<= 0.3%

Th. Exp. Coeff.	Metal Temp. along Len	21.1 °C	0.0000115187 /°C
Elastic Mod. at Design Temperature	120.0 °C		0.19691E+09 KPa.
Elastic Mod. at Metal Temp. along Len	21.1 °C		0.20270E+09 KPa.
Elastic Mod. at Tubsht. Design Temp.	120.0 °C		0.19691E+09 KPa.
Elastic Mod. at Ambient Temperature	21.1 °C		0.20270E+09 KPa.

TubeSheet - TE-1 Carbon & Low Alloy Steels, Group 1

TubeSheet - TM-1 Carbon Steels with C<= 0.3%

Th. Exp. Coeff.	Metal Temp. at Rim	21.1 °C	0.0000115190 /°C
Elastic Mod. at Design Temperature	120.0 °C		0.19691E+09 KPa.
Elastic Mod. at Metal Temp. at Rim	21.1 °C		0.20270E+09 KPa.
Elastic Mod. at Ambient Temperature	21.1 °C		0.20270E+09 KPa.

Note:

The Elasticity and Alpha values are taken from Tables in ASME II D.

Please insure these properties are consistent with the type of Material for the tubes, shell, channel etc.

Tube Required Thickness under Internal Pressure (Tubeside pressure):

Thickness Due to Internal Pressure:

$$= (P*(D/2-CAE)) / (S*E+0.4*P) \text{ per Appendix 1-1 (a)(1)}$$

$$= (24.03*(25.4/2-0.0))/(117.9*1.0+0.4*24.03)$$

$$= 0.2568 + 0.0000 = 0.2568 \text{ mm.}$$

Tube Required Thickness under External Pressure (Shellside pressure) :

External Pressure Chart	CS-2	at	120.00 °C
Elastic Modulus for Material			199943392.00 KPa.

Results for Max. Allowable External Pressure (Emawp):

TCA	ODCA	SLEN	D/T	L/D	Factor A	B
2.1080	25.40	5876.00	12.05	50.0000	0.0075765	122.73
EMAWP = (4*B)/(3*(D/T)) = (4 *122.731)/(3 *12.0493) = 135.8015 bars						

Results for Req'd Thickness for Ext. Pressure (Tca):

TCA	ODCA	SLEN	D/T	L/D	Factor A	B
0.6452	25.40	5876.00	39.37	50.0000	0.0007098	70.96
EMAWP = (4*B)/(3*(D/T)) = (4 *70.9611)/(3 *39.3675) = 24.0324 bars						

Summary of Tube Required Thickness Results:

Total Required Thickness including Corrosion all.	0.6452 mm.
Allowable Internal Pressure at Corroded thickness	209.61 bars
Required Internal Design Pressure	24.03 bars
Allowable External Pressure at Corroded thickness	135.80 bars
Required External Design Pressure	24.03 bars
Required Thickness due to Shell Side pressure	0.6452 mm.

Detailed Results for load Case D3 un-corr. (Psd,max + Ptd,max)

Intermediate Calculations For Tubesheets Extended As Flanges:

ASME Code, Section VIII Division 1, 2017

Gasket Contact Width,	N = (Goc-Gic) / 2	15.000 mm.
Basic Gasket Width,	b0 = N / 2.0	7.500 mm.

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Effective Gasket Width, $b = \text{SQRT}(b_0) * 2.5$ 6.899 mm.
 Gasket Reaction Diameter, $G = G_0 - 2.0 * b$ 627.203 mm.

Bolting Information for UNC Thread Series (Non Mandatory):

Distance Across Corners for Nuts 35.128 mm.
 Circular Wrench End Diameter a 52.388 mm.

	Minimum	Actual	Maximum
Bolt Area, cm ²	60.575	62.348	
Radial Distance between Hub and Bolts:	20.637	36.000	
Radial Distance between Bolts and Edge:	20.637	21.000	
Circ. Spacing between the Bolts:	44.450	65.966	84.100

Flange Design Bolt Load, Seating Condition W : 1059.36 kN
 Flange Design Bolt Load, Operating Condition Wm1: 1002.73 kN

Results for ASME Fixed Tubesheet Calculations for Configuration b.

Results for Tubesheet Calculations Original Thickness :

UHX-13.5.1 Step 1:

Compute the Tube Expansion Depth Ratio [rho]:
 $= l_{tx} / h$ (modified for corrosion if present)
 $= 50.0 / 62.0 = 0.8065$ (must be $0 \leq \rho \leq 1$)

Compute the Effective Tube Hole Diameter [d*]:
 $= \text{Max}(dt - 2tt * (E_t / E) (S_t T / S) (\rho), dt - 2tt)$
 $= \text{Max}(25.4 - 2 * 2.108 * (.19691E+09 / .19691E+09) * (117 / 137) * (0.806), 25.4 - 2 * 2.108)$
 $= 22.4930$ mm.

Compute the Equivalent Outer Tube Limit Circle Diameter [Do]:
 $= 2 * r_o + dt = 2 * 270.77 + 25.4 = 566.94$ mm.

Determine the Basic Ligament Efficiency for Shear [mu]:
 $= (p - dt) / p = (32.0 - 25.4) / 32.0 = 0.2062$

Compute the Equivalent Outer Tube Limit Radius [ao]:
 $= D_o / 2 = 566.94 / 2 = 283.47$ mm.

Compute the Effective Tube Pitch [p*]:
 $= p / \text{sqrt}(1 - 4 * \text{min}(AL * CNV_factor, 4 * D_o * p) / (\text{Pi} * D_o^2))$
 $= 32.0 / \text{sqrt}(1 - 4 * \text{min}(226.77 * 100.0, 4 * 566.94 * 32.0) / (3.141 * 566.94^2))$
 $= 33.5420$ mm.

Compute the Effective Ligament Efficiency for Bending [mu*]:
 $= (p^* - d^*) / p^* = (33.542 - 22.493) / 33.542 = 0.3294$

Compute the Ratio [Rhos]:
 $= a_s / a_o = 290.5 / 283.47 = 1.0248$

Compute the Ratio [Rhoc]:
 $= a_c / a_o = 313.6014 / 283.47 = 1.106295$

Compute Parameter [xt]:
 $= 1 - N_t * ((dt - 2 * tt) / (2 * a_o))^2$

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 DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

$$= 1 - 241 * ((25.4 - 2 * 2.108) / (2 * 283.47))^2 = 0.6635$$

Determine Parameter [xs]:

$$= 1 - Nt * (dt / (2 * ao))^2$$

$$= 1 - 241 * (25.4 / (2 * 283.47))^2 = 0.5163$$

Determine the Value [h'g]:

$$= \text{Max}((hg - \text{CATC}), 0) \quad (\text{For pressure only cases})$$

$$= \text{Max}((5.0 - 0.0), 0) = 5.0 \text{ mm.}$$

UHX-13.5.2 Step 2:

Determine the Axial Shell Stiffness [Ks]:

$$= \pi * ts * (Ds + ts) Es / L$$

$$= 3.1416 * 10.0 * (581.0 + 10.0) * 1.9691E+09 / 5876.0$$

$$= 622191808.0000 \text{ KPa.} * \text{mm.}$$

Determine the Axial Tube Stiffness [Kt]:

$$= \pi * tt * (Dt - tt) Et / L$$

$$= 3.1416 * 2.108 * (25.4 - 2.108) * 1.9691E+09 / 5876.0$$

$$= 5169092.0000 \text{ KPa.} * \text{mm.}$$

Compute the Stiffness Factor [Ks,t]:

$$= Ks / (Nt * Kt) = 0.62219E+09 / (241 * 5169092) = 0.49945$$

Rigidity Ratio [J]:

$$= 1 / (1 + Ks / Kj)$$

$$= 1 / (1 + 0.62219E+09 / 0.0) = 1. \quad (= 1 \text{ if No Exp. Jt.})$$

Compute Shell Coefficient [betas]:

$$= ((12 * (1 - \nu_s^2))^{0.25}) / ((Ds + ts) * ts)^{0.5}$$

$$= ((12 * (1 - 0.3^2))^{0.25}) / ((581.0 + 10.0) * 10.0)^{0.5}$$

$$= 0.0236 \text{ 1/mm.}$$

Determine Shell Coefficient [ks]:

$$= \text{betas} * Es * ts^3 / (6 * (1 - \nu_s^2))$$

$$= 0.024 * 0.19691E+09 * 10.0^3 / (6 * (1 - 0.3^2))$$

$$= 8527819.0000 \text{ bars*mm.}^2$$

Determine Shell Coefficient [Lambdas]:

$$= (6 * Ds * ks) / (h^3) * (1 + h * \text{betas} + 0.5 * (h * \text{betas})^2)$$

$$= 6 * 581.0 * 8527819 / (62.0^3) * (1 + 62.0 * 0.024 + 1.075)$$

$$= 441656.3750 \text{ bars}$$

Determine Shell Coefficient [deltaS]:

$$= Ds^2 / (4 * Es * Ts) * (1 - \nu_s / 2)$$

$$= 581.0^2 / (4 * 0.19691E+09 * 10.0) * (1 - 0.3 / 2)$$

$$= 0.0364264846 \text{ mm./N./mm}^2$$

Intermediate parameters for Tubesheet Gasketed on the Channel Side:
 betac, kc, deltaC, Lambdac = 0

UHX-13.5.3 Step 3:

E*/E and nu* for Triangular pattern from Fig. UHX-11.3.

$$h/p = 1.937500 ; \mu^* = 0.329408$$

$$E^*/E = 0.322262 ; \nu^* = 0.334182 ; E^* = 63456788. \text{ KPa.}$$

Compute the Tube Bundle Stiffness Factor [Xa]:

$$= ((24 * (1 - \nu^*) * Nt * Et * tt * (dt - tt) * ao^2) / (E^* * L * H^3))^{0.25}$$

Tag no:E-PK6101-3 ECONOMIZER

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$$= ((24 *(1 - 0.334^2)*241 *.19691E+09*2.108 * (25.4 - 2.108)^283.47^2)/(63456788 * 5876.0 * 62.0^3))^{0.25}$$
$$= 2.5888$$

Values from Table UHX-13.1

$$Zd = 0.082771 ; Zv = 0.130480 ; Zm = 0.578202$$
$$Za = 0.138738E+01 ; Zw = 0.130480$$

UHX-13.5.4 Step 4:

Compute the Diameter Ratio [K]:

$$= A/Do = 644.0/566.94 = 1.1359$$

Compute Coefficient [F]:

$$= (1 - nu^*) / (E^*) * (Lambdas + Lambdac + E * ln(K))$$
$$= (1 - 0.33)/(63456788) * (441656.38 + 0.0 + 0.19691E+09 * ln(1.14))$$
$$= 0.7267$$

Compute Parameter [Phi]:

$$= (1 + nu^*) * F = (1 + 0.3342) * 0.7267 = 0.9696$$

Compute Parameter [Q1]:

$$= (Rhos - 1 - Phi * Zv)/(1 + Phi * Zm)$$
$$= (1.0248 - 1 - 0.9696 * 0.1305)/(1 + 0.9696 * 0.5782)$$
$$= -0.065173514$$

Compute Parameter [Qz1]:

$$= (Zd + Q1*Zw)/2 *Xa^4$$
$$= (0.08277 + -0.06517 * 0.13048)/2 *2.58885^4 = 1.668$$

Compute Parameter [Qz2]:

$$= (Zv + Q1*Zm)/2 *Xa^4$$
$$= (0.13048 + -0.06517 * 0.5782)/2 *2.58885^4 = 2.0841$$

Compute Parameter [U]:

$$= (Zw + (Rhos - 1)*Zm)*Xa^4 / (1 + Phi*Zm)$$
$$= (0.1305 + (1.0248 - 1)* 0.5782)^2.58885^4/(1 + 0.9696 * 0.5782)$$
$$= 4.1683$$

UHX-13.5.5 Step 5:

Determine factor [gamab]:

$$= (Gc - C) / Do (config b)$$
$$= (627.2029 - 673.0)/566.94 = -0.08078$$

Compute Parameter [gamma]:

$$= 0.000 \text{ mm. (For Pressure only cases)}$$

Calculate Parameter [OmegaS]:

$$= rhos * ks * Betas * deltaS(1 + h * Betas)$$
$$= 1.0248 * 8527819 * 0.0236 * 0.036426 (1 + 62.0 * 0.0236)$$
$$= 1856.4625 \text{ mm.}^2$$

Calculate Parameter [Omega*S]:

$$= Ao^2 * (Rhos^2 - 1) * (Rhos - 1) / 4 - OmegaS$$
$$= 283.47^2 * (1.025^2 - 1) * (1.025 - 1)/4 - 1856.463$$
$$= -1831.4457 \text{ mm.}^2$$

Calculate Parameter [OmegaC]:

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Tag no:E-PK6101-3 ECONOMIZER

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ASME TS Calc: [Case: 1 11:46p Dec 22,2021

$$\begin{aligned}
 &= \text{rhoc} * \text{kc} * \text{Betac} * \text{deltaC} (1 + \text{h} * \text{Betac}) \\
 &= 1.1063 * 0.0 * 0.0 * 0. (1 + 62.0 * 0.0) \\
 &= 0.0000 \text{ mm.}^2
 \end{aligned}$$

Calculate Parameter [Omega*C]:

$$\begin{aligned}
 &= \text{ao}^2 [(\text{Rhoc}^2 + 1) * (\text{Rhoc} - 1) / 4 - (\text{Rhos} - 1) / 2] - \text{OmegaC} \\
 &= 283.47^2 [(1.10629^2 + 1) * (1.10629 - 1) / 4 - (1.0248 - 1) / 2] - 0. \\
 &= 3752.3599 \text{ mm.}^2
 \end{aligned}$$

Compute the Pressure [P*S]:

= 0 For Pressure only cases or Configurations d,e,f,A,B,C,D

Compute the Pressure [P*C]:

= 0 For Pressure only cases or Configurations b,c,d,B,C,D

UHX-13.5.6 Step 6:

Compute the Pressure [P's]:

$$\begin{aligned}
 &= \text{Ps} * \{ \text{xs} + 2(1 - \text{xs}) \text{nut} + [2 / \text{Kst} (\text{Ds} / \text{Do})^2] \text{nus} - \\
 &\quad [(\text{rhos}^2 - 1) / (\text{J} * \text{Kst})] - [(1 - \text{J}) / (2 \text{J} * \text{Kst})] [(\text{Dj}^2 - (\text{Ds})^2) / \text{Do}^2] \} \\
 &= 23.0 * \{ 0.516 + 2(1 - 0.516) 0.3 + \\
 &\quad [2 / 0.499 (581.0 / 566.94)^2] 0.3 - \\
 &\quad [(1.025^2 - 1) / (1.0 * 0.499)] - \\
 &\quad [(1 - 1.0) / (2 * 1.0 * 0.499)] [(0.0^2 - (581.0)^2) / 566.94^2] \} \\
 &= 45.2550 \text{ bars}
 \end{aligned}$$

Compute the Pressure [P't]:

$$\begin{aligned}
 &= [\text{xt} + 2(1 - \text{xt}) \text{nut} + 1 / (\text{J} * \text{Kst})] * \text{Pt} \\
 &= [0.664 + 2(1 - 0.664) 0.3 + \\
 &\quad 1 / (1. * 0.499)] * 23.0 \\
 &= 65.9549 \text{ bars}
 \end{aligned}$$

Compute the Pressure [Pgama]:

$$\begin{aligned}
 &= \text{Nt} * \text{Kt} * \text{gama} / (\text{pi} * \text{ao}^2) \\
 &= 241 * 5169092 * 0.0 / (3.142 * 283.47^2) = 0.0 \text{ bars}
 \end{aligned}$$

Compute the Pressure [Pw]:

$$\begin{aligned}
 &= -\text{gamab} * \text{U} * \text{W} * / (2 * \text{pi} * \text{ao}^2) \\
 &= --0.081 * 4.168 * 1002.73 / (2 * 3.142 * 283.47^2) \\
 &= 6.6874 \text{ bars}
 \end{aligned}$$

Calculate the Pressure [Prim]:

$$\begin{aligned}
 &= - (\text{U} / \text{ao}^2) (\text{Omega} * \text{S} * \text{Ps} - \text{Omega} * \text{C} * \text{Pt}) \\
 &= - (4.168 / 283.47^2) (-2.839 * 23.0 - 5.816 * 23.0) \\
 &= 6.6619 \text{ bars}
 \end{aligned}$$

Calculate the Pressure [POmega]:

$$\begin{aligned}
 &= \text{U} / \text{ao}^2 (\text{Omega} * \text{S} * \text{P*s} - \text{Omega} * \text{C} * \text{P*c}) \\
 &= 4.168 / 283.47^2 (2.8775 * 0.0 - 0.0 * 0.0) \\
 &= 0.0000 \text{ bars}
 \end{aligned}$$

Determine the Effective Pressure [Pe]:

$$\begin{aligned}
 &= \text{J} * \text{Kst} / (1 + \text{J} * \text{Kst} * (\text{Qz1} + (\text{Rhos} - 1) * \text{Qz2})) * \\
 &\quad (\text{P's} - \text{P't} + \text{Pgama} + \text{Pw} + \text{Prim}) \\
 &= 0.1000\text{E}+01 * 0.499 / (1 + 1.0 * 0.499 * (1.668 + (1.025 - \\
 &\quad 1) * 2.084)) * (45.255 - 65.955 + 0.0 + 6.687 + 6.662) \\
 &= -1.9750 \text{ bars}
 \end{aligned}$$

UHX-13.5.7 Step 7:

Determine Factor [Q2]:

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
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 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

$$\begin{aligned}
 &= [((\Omega * S * P_s - \Omega * C * P_t) - (\Omega * P * s - \Omega * C * P * c)) * CNV_FAC + \\
 &W * \gamma / (2 * \pi)] / (1 + \Phi * Z_m) \\
 &= [((-1831.446 * 23.0 - 3752.36 * 23.0) - \\
 &(1856.463 * 0.0 - 0.0 * 0.0)) * 0. + \\
 &1002.7 * -0.081 / (2 * 3.141)] / (1 + 0.96957 * 0.5782) \\
 &= -16.489673615 \text{ kN}
 \end{aligned}$$

Calculate Factor [Q3]:

$$\begin{aligned}
 &= Q_1 + 2 * Q_2 / (P_e * a_o^2) \\
 &= -0.065 + 2 * -16.49 / (-1.975 * 283.47^2) \\
 &= 2.012980
 \end{aligned}$$

Fm Value from Table UHX-13.1 = 1.035901

The Tubesheet Bending Stress - Original Thickness [Sigma]:

$$\begin{aligned}
 &= (1.5 * F_m / \mu *) * (2 * a_o / (H - h'g))^2 * P_e \\
 &= (1.5 * 1.0359 / 0.3294) * (2 * 283.47 / (62.0 - 5.0))^2 * -1.97 \\
 &= -92.1695 \text{ N./mm}^2
 \end{aligned}$$

The Allowable Tubesheet Bending Stress [Sigma allowed]:

$$= 1.5 * S = 1.5 * 137.9 = 206.85 \text{ N./mm}^2$$

The Tubesheet Bending Stress - Final Thickness [Sigma_f]:

$$\begin{aligned}
 &= (1.5 * F_m / \mu *) * (2 * a_o / (h - h'g))^2 * P_e \\
 &= (1.5 * 1.9125 / 0.3294) * (2 * 283.47 / (40.563 - 5.0))^2 * 0.93 \\
 &= 206.8453 \text{ N./mm}^2
 \end{aligned}$$

Reqd Tubesheet Thickness, for Bending Stress (Including CA) [HReqB]:

$$= h + C_{at_s} + C_{at_c} = 40.5633 + 0.0 + 0.0 = 40.5633 \text{ mm.}$$

UHX-13.5.8 Step 8:

Shear Stress check [Tau_limit]:

$$\begin{aligned}
 &= 1.6 * S * \mu * h / a_o \\
 &= 1.6 * 137.9 * 0.206 * 62.0 / 283.47 \\
 &= 9.9532 \text{ N./mm}^2
 \end{aligned}$$

The Shear Stress is not required to be computed; [Pe] <= Tau_limit

Note: Tubesheet Shear Stress is probably low, use the following req. thk:

$$\begin{aligned}
 \text{Tubesheet thickness (Incl. Corr.)} &= 3.8100 \text{ mm.} \\
 \text{Tubesheet Shear Stress} &= 35.3559 \text{ N./mm}^2
 \end{aligned}$$

Reqd Tubesheet Thickness for Given Loadings (Including CA) [Hreqd]:

$$= \text{Max}(H_{reqB}, H_{reqS}) = \text{Max}(40.5633, 3.81) = 40.5633 \text{ mm.}$$

UHX-13.5.9 Step 9:

The Ftmin and Ftmax Coefficients from Table UHX-13.2:

$$F_{tmin} = -5.0157, F_{tmax} = 7.758$$

First Extreme Tube Axial Stress from among all the tubes [Sigma_t1]:

$$\begin{aligned}
 &= (P_s * x_s - P_t * x_t) - P_e * F_{tmin} / (X_t - X_s) \\
 &= (23.0 * 0.5163 - 23.0 * 0.6635) - (-1.975) * -5.016 / \\
 &(0.6635 - 0.5163) \\
 &= -9.0274 \text{ N./mm}^2
 \end{aligned}$$

Second Extreme value of Tube Axial Stress from among all the tubes [Sigma_t2]:

$$\begin{aligned}
 &= (P_s * x_s - P_t * x_t) - P_e * F_{tmax} / (X_t - X_s) \\
 &= (23.0 * 0.5163 - 23.0 * 0.6635) - (-1.975) * 7.758 /
 \end{aligned}$$

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

$$(0.6635 - 0.5163)) \\ = 8.1052 \text{ N./mm}^2$$

Maximum Tube Axial Stress [Sigmat,max]:
 $= \text{MAX}(\text{abs}(\text{Sigmat1}), \text{abs}(\text{Sigmat2})) = 9.027 \text{ N./mm}^2$

The Allowable Tube Stress, [SigmatA]:
 $= \text{Sot} = 117.9045 \text{ N./mm}^2$

Check for Buckling as some of the Tubes are in Compression

Determine the Factor of Safety [Fs]:
 $= \text{Max}((3.25 - 0.25*(Zd + Q3*Zw)*Xa^4), 1.25)$
 $= \text{Max}((3.25 - 0.25*(0.083 + 2.013 * 0.13) * 2.589^4), 1.25)$
 $= 1.2500 \text{ (Should be } \leq 2 \text{)}$

Determine the Factor [rt]:
 $= ((dt^2 + (dt - 2*tt)^2)^{.5}) / 4$
 $= ((25.4^2 + (25.4 - 2*2.108)^2)^{.5}) / 4 = 8.2686 \text{ mm.}$

Determine the Factor [Ct]:
 $= (2 * \text{PI}^2 * \text{Et}/\text{Syt})^{.5}$
 $= (2 * 3.14^2 * 0.19691\text{E}+09/217)^{.5} = 133.728$

Determine the Factor [Ft]:
 $= k * L/r = 0.8 * 712.0/8.269 = 68.8869$

The Buckling Allowable Stress [Stb]:
 $= \text{Sy,t}/\text{Fs} * (1 - \text{Ft}/(2*\text{Ct}))$
 $= 217/1.25 * (1 - 68.887/(2*133.728))$
 $= 117.905 \text{ N./mm}^2 \text{ (Never greater than Sot)}$

Note: The Axial Compressive stress in Tubes is within limits.

The Largest tube-to-tubesheet Joint Load [Wt]:
 $= \text{Sigmat,max} * \text{Tube Area} = 9.03 * 1.5425 = 1.39 \text{ kN}$

Tube Weld Size Results per UW-20:

Tube Strength [Ft]:
 $= 3.1415 * t * (do - t) * \text{Sa}$
 $= 3.1415 * 2.108 * (25.4 - 2.108) * 117.9 = 18.185 \text{ kN}$

Fillet Weld Strength [Ff]:
 $= 0.55 * 3.1415 * \text{af} * (do + 0.67*\text{af}) * \text{Sw} \text{ (but not } > \text{ Ft)}$
 $= 0.55 * 3.1415 * 1.5 * (25.4 + 0.67*1.5) * 117.9$
 $= 8.0683 \text{ kN}$

Groove Weld Strength [Fg]:
 $= 0.85 * 3.1415 * \text{ag} * (do + 0.67*\text{ag}) * \text{Sw} \text{ (but not } > \text{ Ft)}$
 $= 0.85 * 3.1415 * 1.5 * (25.4 + 0.67*1.5) * 117.9$
 $= 12.4692 \text{ kN}$

Max. Allow. Tube-Tubesheet Joint load, Lmax
 $= \text{Ft} = 18.1853 \text{ kN}$

Design Strength Ratio [fd]:
 $= 1.0000$

Weld Strength Factor [fw]:
 $= \text{Sot} / (\text{Min}(\text{Sot}, \text{S})) = 1.0000$

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Min Weld Length [ar]:

$$= 2 * ((0.75 * do)^2 + 1.07 * t * (do - t) * fw * fd) ^{1/2} - 0.75 * do$$

$$= 2.6646 \text{ mm.}$$

Minimum Required Fillet Weld Leg afr 1.3323 mm.
 Minimum Required Groove Weld Leg agr 1.3323 mm.

Tube-Tubesheet Jt allowable, 18.19 is >= tube strength 18.19 kN

Note: This tube-tubesheet joint is a Full Strength joint

UHX-13.5.10 Step 10:

Shell Axial Membrane Allowable Stress:

$$= S_s * E_{sw} = 137.9 * 1.0 = 137.9 \text{ N./mm}^2$$

Axial Membrane Stress in Shell [Sigmas,m]:

$$= ao^2 / ((Ds+ts)*ts) * [Pe + (Rhos^2-1)(Ps-Pt)] + as^2 * Pt / ((Ds+ts)*ts)$$

$$= 283.47^2 / ((581.0 + 10.0) * 10.0) * [-1.97 + (1.025^2 - 1) (23.0 - 23.0)] + 290.5^2 * 23.0 / ((581.0 + 10.0) * 10.0)$$

$$= 30.1587 \text{ N./mm}^2$$

UHX-13.5.11 Step 11:

Note:

For a given Shell thickness of 10.0 mm., the minimum Shell length adjacent to the tubesheet should be 137.202 mm.

The Shell Membrane Stress due to Joint Interaction [Sigmas,m]:

$$= ao^2 / ((Ds+ts)*ts) [Pe + (Rhos^2-1)(Ps-Pt)] + as^2 * Pt / ((Ds+ts)*ts)$$

$$= 283.47^2 / ((581.0 + 10.0) * 10.0) [-1.97 + (1.025^2 - 1) (23.0 - 23.0)] + 290.5^2 * 23.0 / ((581.0 + 10.0) * 10.0)$$

$$= 30.1587 \text{ N./mm}^2$$

The Shell Bending Stress due to Joint Interaction [Sigmasb]:

$$= 6 * ks / ts^2 \{ betas [delta * Ps + as^2 * PstarS / (Es * ts)] + 6(1 - nu^2) / (E^*) (ao/h)^3 (1 + h * betas / 2) [Pe (Zv + Zm * Q1) + 2 / ao^2 * Zm * Q2] \}$$

$$= 6 * 8527819 / 10.0^2 \{ 0.024 [0.036 * 23.0 + 290.5^2 * 0.0 / (.19691E+1)] + 6(1 - 0.33^2) / (63456788) (283.47 / 62.0)^3 (1 + 62.0 * 0.02 / 2) [-2.0 (0.13 + 0.578 * -0.065) + 2 / 283.47^2 * 0.578 * -16.49] \}$$

$$= -80.6076 \text{ N./mm}^2$$

Shell Stress Summation vs. Allowable

$$abs(Sigmasm) + abs(Sigmasb) \leq 1.5 * S_s$$

$$abs(30.2) + abs(-80.6) \leq 206.85 \text{ N./mm}^2$$

$$110.77 \text{ must be } < \text{ or } = 206.85 \text{ N./mm}^2$$

Computations Completed for ASME Tubesheet Configuration b

Stress/Force Summary for Loadcase D3 un-corr. (Psd,max + Ptd,max):

Stress Description	Actual	Allowable	Pass/Fail
Tubesheet Bend. Stress	92.2 <=	206.9 N./mm ²	Ok
Tubesheet Shear Stress	2.2 <=	110.3 N./mm ²	Ok

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Maximum Tube Stress	9.0	<=	117.9	N./mm ²	Ok
Minimum Tube Stress (Buckling)	-9.0	<=	-117.9	N./mm ²	Ok
Maximum Force on any one Tube	1.4	<=	18.2	kN	Ok
Axial Membrane Stress in Shell	30.2	<=	137.9	N./mm ²	Ok
Shell Stress (jt. inter.)	110.8	<=	206.9	N./mm ²	Ok

Thickness Results for Loadcase D3 un-corr. (Psd,max + Ptd,max):

Thickness (mm.)	Required	Actual	P/F
Tubesheet Thickness :	40.563	62.000	Ok
Tube-Tubesheet Fillet Weld Leg :	1.332	1.500	Ok
Tube-Tubesheet Groove Weld Leg :	1.332	1.500	Ok

Fixed Tubesheet results per ASME UHX-13 2017**Results for 16 Load Cases:**

Case#	--Reqd. Thk. + CA		---- Tubesheet Stresses				Case Type	Pass/Fail
	Tbsht	Extnsn	Bend	Allwd	Shear	Allwd		
D1uc	38.444	23.085	114	207	17	110	Ps+Pt-Th	D1 Ok
D2uc	23.041	...	56	207	15	110	Ps+Pt-Th	D2 Ok
D3uc	40.563	...	92	207	2	110	Ps+Pt-Th	D3 Ok
D4uc	6.585	...	2	207	...	110	Ps+Pt-Th	D4 Ok
O1uc	3.810	...	117	447	23	110	Ps+Pt+Th	O1 Ok
O2uc	16.879	...	64	447	9	110	Ps+Pt+Th	O2 Ok
O3uc	16.735	...	89	447	9	110	Ps+Pt+Th	O3 Ok
O4uc	6.394	...	46	447	5	110	Ps+Pt+Th	O4 Ok
D1c	46.468	23.085	131	207	18	110	Ps+Pt-Th-c	D1 Ok
D2c	21.253	...	56	207	16	110	Ps+Pt-Th-c	D2 Ok
D3c	45.568	...	108	207	3	110	Ps+Pt-Th-c	D3 Ok
D4c	9.810	...	3	207	...	110	Ps+Pt-Th-c	D4 Ok
O1c	28.098	...	139	447	24	110	Ps+Pt+Th-c	O1 Ok
O2c	25.900	...	78	447	11	110	Ps+Pt+Th-c	O2 Ok
O3c	29.435	...	111	447	9	110	Ps+Pt+Th-c	O3 Ok
O4c	22.314	...	60	447	3	110	Ps+Pt+Th-c	O4 Ok
Max:	46.4684	23.085	mm.	0.633		0.213	(Str. Ratio)	

Load Case Definitions:

[Ps & Pt]:

Shell-side and Tube-side Design or Operating Pressures
derived from Psd,min Ptd,max, Psox,min, Ptox,max etc. per the
Load Case Tables

[(+/-)Th]:

With or Without Thermal Expansion, Tt,mx & Ts,mx

[c]:

With or Without Corrosion Allowance

[D1, D2, D3]:

Design Load Cases using the Maximum and Minimum Design Pressures

[D4]:

Design Load Case using the Minimum (Vacuum) Pressures (if specified)

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 Tag no:E-PK6101-3 ECONOMIZER
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 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

[O1, O2, O3, O4]:
 Operating Load Cases using the Maximum and Minimum Operating Pressures and
 Operating Temperatures

Shell Axial Membrane Stress Summary:

Case#	Shell Stresses				:	Shell Band Stress				: Pass Fail
	Ten	Allwd	Cmp	Allwd		Ten	Allwd	Cmp	Allwd	
D1uc	10	137	:	Ok
D2uc	19	137	:	Ok
D3uc	30	137	:	Ok
D4uc	1	137	-1	-117	:	Ok
O1uc	3	471	:	Ok
O2uc	13	471	:	Ok
O3uc	22	471	:	Ok
O4uc	6	471	-6	-117	:	Ok
D1c	15	137	:	Ok
D2c	26	137	:	Ok
D3c	43	137	:	Ok
D4c	2	137	-2	-111	:	Ok
O1c	7	471	:	Ok
O2c	21	471	:	Ok
O3c	34	471	:	Ok
O4c	5	471	-5	-111	:	Ok
Max RATIO		0.312		0.052	:		

Tube, Shell and Channel Stress Summary:

Case#	Tube Stresses				Tube Loads		Shell Stress		Channel Stress		Pass Fail
	Ten	Allwd	Cmp	Allwd	Ld	Allwd	Stress	Allwd	Stress	Allwd	
D1uc	14	117	-14	-80	2	18	256	471	Ok
D2uc	6	117	-6	-80	1	18	184	206	Ok
D3uc	9	117	-9	-117	1	18	111	206	Ok
D4uc	...	117	...	-100	...	18	2	206	Ok
O1uc	20	235	-13	-80	3	36	287	471	Ok
O2uc	4	235	1	36	52	471	Ok
O3uc	15	235	-8	-97	2	36	148	471	Ok
O4uc	9	235	-3	-95	1	36	125	471	Ok
D1c	18	117	-18	-86	3	18	308	471	Ok
D2c	8	117	-7	-80	1	18	255	471	Ok
D3c	11	117	-11	-117	2	18	105	206	Ok
D4c	...	117	...	-107	...	18	3	206	Ok
O1c	24	235	-18	-83	4	36	335	471	Ok
O2c	3	235	1	36	110	471	Ok
O3c	18	235	-11	-117	3	36	140	471	Ok
O4c	10	235	-5	-117	1	36	137	471	Ok
Max RATIO		0.154		0.218		0.154		0.889		...	

Summary of Thickness Comparisons for 16 Load Cases:

Thickness (mm.)	Required	Actual	P/F
Tubesheet Thickness :	46.468	62.000	Ok
Tubesheet Thickness Flanged Extension :	23.085	46.000	Ok
Tube Thickness :	0.645	2.108	Ok
Tube-Tubesheet Fillet Weld Leg :	1.332	1.500	Ok
Tube-Tubesheet Groove Weld Leg :	1.332	1.500	Ok

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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 ASME TS Calc: [] Case: 1 11:46p Dec 22,2021

Min Shell length of thk, (10.000) adj. to tubesheet: 137.202 mm.

Note: This is a full strength Tube to Tubesheet Joint.

Summary of Axial Differential Expansion between Shell and Tubes :

Due to Thermal Expansion Shell Compresses by : -0.478 mm.
 Due to Pressure Shell Compresses by : -0.103 mm.
 Due to Pressure + Thermal Shell Compresses by : -0.580 mm.

Tubesheet MAWP used to Compute Hydrotest Pressure:

Stress / Force Condition	Tubeside MAWP	0 shellside Stress Rat.	Shellside MAWP	0 tubeside Stress Rat.
Tubesheet Bending Stress	37.177	1.000	91.595	1.000
Tubesheet Shear Stress	128.592	1.000	172.522	1.000
Tube Tensile Stress	152.147	1.000	383.662	1.000
Tube Compressive Stress	92.679	0.801	303.414	1.000
Tube-Tubesheet Joint load	152.147	1.000	383.661	1.000
Shell Stress (Axial, Junction)	35.612	1.000	43.964	1.000
Tube Pressure Stress	209.607	1.000	135.801	1.000
Tubesheet Extension Stress	24.648	...	No Calc	No Calc
Minimum MAWP	24.648		43.964	

Tubesheet MAPnc used to Compute Hydrotest Pressure:

Stress / Force Condition	Tubeside MAPnc	0 shellside Stress Rat.	Shellside MAPnc	0 tubeside Stress Rat.
Tubesheet Bending Stress	40.660	0.995	91.595	1.000
Tubesheet Shear Stress	155.495	1.000	179.588	1.000
Tube Tensile Stress	192.723	1.000	537.675	1.000
Tube Compressive Stress	144.474	1.000	403.546	1.000
Tube-Tubesheet Joint load	192.722	1.000	537.674	1.000
Shell Stress (Axial, Junction)	40.660	0.995	68.157	1.000
Tube Pressure Stress	209.607	1.000	135.801	1.000
Tubesheet Extension Stress	24.648	...	No Calc	No Calc
Minimum MAPnc	24.648		68.157	

(*) All load cases were analyzed to compute the MAWP for determining the test pressure.

Tubesheet MDMT Calculations:

Note: The loading conditions from this case will be used to determine the tubesheet MDMT.

Shell Side MDMT calculation:

Governing thickness on the shell side per figure UCS-66.3 (e):
 = max(tubesheet thk/4, min(tubesheet thk, shell thickness))
 = max(62.0/4, min(62.0, 10.0))
 = 15.500 mm.

Thickness Ratio = 0.933, Temperature Reduction per Fig. UCS 66.1 = 4 °C

Min Metal Temp. w/o impact per UCS-66, Curve D -46 °C
 Min Metal Temp. at Required thickness (UCS 66.1) -48 °C

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
Tag no:E-PK6101-3 ECONOMIZER
PV Elite 2019 SP1 Licensee: SPLM Licensed User
FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
ASME TS Calc: ¶ Case: 1 11:46p Dec 22,2021

Channel Side MDMT calculation:

Governing thickness for the channel side:

= tubesheet thickness/4
= 62.0/4
= 15.500 mm.

Note:

This Material was specified as being an Impact Tested (Low Temperature) Material.

Impact Test Temperature provided per Specification -46 °C

where the MDMT reduction ratio per UCS 66 (b)(1)(b) is:

= max(pt/Tubeside MAPnc, ps/Shellside MAPnc), must be <= 1
= max(23.0/24.65, 23.0/68.16)
= 0.933

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT

Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

MDMT Summary: Step: 23 11:46pm Dec 22,2021

Minimum Design Metal Temperature Results Summary :

Description	Notes	Curve	Basic MDMT °C	Reduced MDMT °C	UG-20(f) MDMT °C	Thickness ratio	Gov Thk mm.	E*	PWHT reqd
SHELL	[8]	D	-48	-48	-29	0.707	10.000	1.00	No
S2	[1]	B	-29	-45	-29	0.706	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
S1	[1]	B	-29	-45	-29	0.706	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
S3	[1]	B	-29	-45	-29	0.707	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
Tubesheet: SS[13]		D	-46	-48	-29	0.933	15.500	1.00	No
Warmest MDMT:			-29	-45					
BODY FLANGE 0[11]	!		-46	-46		0.711	10.000	1.00	No
BODY FLANGE 0[11]	!		-46	-46		0.710	10.000	1.00	No
HEAD 1	[10]	D	-48	-48	-29	0.692	10.000	1.00	No
HEAD 1	[7]	D	-48	-48	-29	0.550	12.000	1.00	No
CHANNEL 01	[8]	D	-48	-48	-29	0.707	10.000	1.00	No
CHANNEL 002	[8]	D	-48	-48	-29	0.707	10.000	1.00	No
HEAD 002	[10]	D	-48	-48	-29	0.692	10.000	1.00	No
HEAD 002	[7]	D	-48	-48	-29	0.550	12.000	1.00	No
T4	[1]	D	-48	-48	-29	0.706	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
T1	[1]	B	-29	-45	-29	0.707	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
T2	[1]	B	-29	-45	-29	0.706	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
T3	[1]	D	-48	-48	-29	0.707	10.000	1.00	No
Nozzle Flg	[4]	!	-46	-104					
Tubesheet: CS[14]	!		-46	-46		0.933	15.500	1.00	No
Warmest MDMT:			-29	-45					
Exchanger Side			Computed MDMT °C		Required MDMT °C			Pass/Fail	
Shell			-45.0		-45.0			Pass	
Channel/Tube			-45.0		-45.0			Pass	

Notes:

- [!] - This was an impact tested material.
- [1] - Governing Nozzle Weld.
- [4] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-c).
- [5] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-b).
- [6] - MDMT Calculations at the Shell/Head Joint.
- [7] - MDMT Calculations for the Straight Flange.
- [8] - Cylinder/Cone/Flange Junction MDMT.
- [9] - Calculations in the Spherical Portion of the Head.
- [10] - Calculations in the Knuckle Portion of the Head.
- [11] - Calculated (Body Flange) Flange MDMT.
- [12] - Calculated Flat Head MDMT per UCS-66.3
- [13] - Tubesheet MDMT, shell side, if applicable
- [14] - Tubesheet MDMT, tube side, if applicable
- [15] - Nozzle Material
- [16] - Shell or Head Material

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DEHDASHT HIGH DENSITY POLYETHYLENE PROJECT
Tag no:E-PK6101-3 ECONOMIZER
PV Elite 2019 SP1 Licensee: SPLM Licensed User
FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
MDMT Summary: Step: 23 11:46pm Dec 22,2021

[17] - Impact Testing required
[18] - Impact Testing not required, see UCS-66(b)(3)
[20] - Cylinder/Cone Junction MDMT based on Longitudinal Stress considerations
[21] - Bolting Material

UG-84(b)(2) was not considered.
UCS-66(g) was not considered.
UCS-66(i) was not considered.

Notes:

Impact test temps were not entered in and not considered in the analysis.
UCS-66(i) applies to impact tested materials not by specification and
UCS-66(g) applies to materials impact tested per UG-84.1 General Note (c).
The Basic MDMT includes the (30F) PWHT credit if applicable.

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Tag no:E-PK6101-3 ECONOMIZER

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FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----

Vessel Design Summary: Step: 24 11:46pm Dec 22,2021

ASME Code, Section VIII Division 1, 2017

Diameter Spec : 581.000 mm. ID
 Vessel Design Length, Tangent to Tangent 7106.35 mm.
 Specified Datum Line Distance 50.00 mm.
 Shell Side Design Temperature 120 °C
 Channel Side Design Temperature 120 °C
 Shell Side Design Pressure 23.000 bars
 Channel Side Design Pressure 23.000 bars
 Wind Design Code ASCE-2010
 Earthquake Design Code ASCE 7-2010

Materials of Construction:

Component Type	Material	Class	Thickness	UNS #	Normalized	Impact Tested
Shell	SA-516 70	K02700	Yes	No
Head	SA-516 70	K02700	Yes	No
Flange	SA-350 LF2	1	...	K03011	No	Yes
Nozzle	SA-350 LF2	1	...	K03011	No	Yes
Nozzle	SA-333 6	K03006	No	Yes
Re-Pad	SA-516 70	K02700	No	No
Nozzle Flg	SA-350 LF2	1	...	K03011	No	Yes
Tubes	SA-334 6	K03006	No	Yes
Tubesheet	SA-350 LF2	1	...	K03011	No	Yes
Flg Bolting	SA-320 L7	...	<= 2 1/2	G41400	No	Yes
Hrz Bolting	SA-193 B7	...	2 1/2 < t <= 4	G41400	No	No

Normalized is determined based on the UCS-66 material curve selection and Figure UCS-66.
 Impact Tested is based on material selection and material data properties.

Element Pressures and MAWP (bars & mm.):

Element Description or Type	Design Pressure + Stat. head	Ext. Press.	Element M.A.W.P	Corrosion Allowance	Str. Flg. Gov.	In Creep Range
HEAD 1	23.034	1.10	No Calc	3.0000	No	No
CHANNEL 01	23.034	1.10	No Calc	3.0000	N/A	No
BODY FLANGE 01	23.037	1.10	No Calc	3.0000	N/A	No
SHELL	23.034	1.10	No Calc	3.0000	N/A	No
BODY FLANGE 002	23.034	1.10	No Calc	3.0000	N/A	No
CHANNEL 002	23.034	1.10	No Calc	3.0000	N/A	No
HEAD 002	23.034	1.10	No Calc	3.0000	No	No

Liquid Level: 581.00 mm. Dens.: 0.001 kg./cm³ Sp. Gr.: 0.600

Element Types and Properties:

Element Type	"To" Elev mm.	Element Length mm.	Nominal Thickness mm.	Finished Thickness mm.	Reqd Thk Internal mm.	Reqd Thk External mm.	Long Eff	Circ Eff
Ellipse	0.0	50.0	12.0	10.0	7.8	4.6	1.00	1.00
Cylinder	407.0	407.0	10.0	10.0	8.0	4.8	1.00	1.00

Tag no:E-PK6101-3 ECONOMIZER

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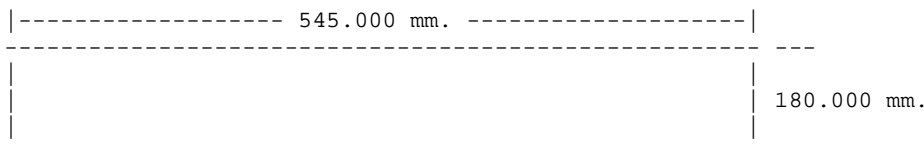
Vessel Design Summary: Step: 24 11:46pm Dec 22,2021

Body Flg	495.0	88.0	79.0	62.0	60.8	50.8	1.00	1.00
Cylinder	6443.2	5880.0	10.0	10.0	8.0	8.1	1.00	1.00
Body Flg	6537.4	88.0	79.0	62.0	60.8	50.8	1.00	1.00
Cylinder	7006.4	407.0	10.0	10.0	8.0	4.8	1.00	1.00
Ellipse	7056.4	50.0	12.0	10.0	7.8	4.6	1.00	1.00

Saddle Parameters:

Saddle Width	150.000	mm.
Saddle Bearing Angle	120.000	deg.
Centerline Dimension	600.000	mm.
Wear Pad Width	225.000	mm.
Wear Pad Thickness	10.000	mm.
Wear Pad Bearing Angle	132.000	deg.
Distance from Saddle to Tangent	992.000	mm.
Baseplate Length	545.000	mm.
Baseplate Thickness	16.000	mm.
Baseplate Width	180.000	mm.
Number of Ribs (including outside ribs)	4	
Rib Thickness	10.000	mm.
Web Thickness	10.000	mm.
Height of Center Web	273.000	mm.
Number of Bolts in Baseplate	4	

Baseplate Sketch



Baseplate Plan View



Baseplate Side View

Maximum Tensile Bolt Load 0. kN

Summary of Maximum Saddle Loads, Operating Case :

Maximum Vertical Saddle Load	42.83	kN
Maximum Transverse Saddle Shear Load	5.74	kN
Maximum Longitudinal Saddle Shear Load	11.49	kN

Summary of Maximum Saddle Loads, Operating Case, Un-Factored :

Maximum Vertical Saddle Load	51.07	kN
Maximum Transverse Saddle Shear Load	19.23	kN
Maximum Longitudinal Saddle Shear Load	16.41	kN

Summary of Maximum Saddle Loads, Hydrotest Case :

Maximum Vertical Saddle Load	30.92	kN
Maximum Transverse Saddle Shear Load	1.00	kN
Maximum Longitudinal Saddle Shear Load	0.20	kN

Local Stress Analysis Results:

	Analysis		Max Stress		Pass	
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 Tag no:E-PK6101-3 ECONOMIZER
 PV Elite 2019 SP1 Licensee: SPLM Licensed User
 FileName : Calculation Book for ECONOMIZER E-PK6101-3 ----
 Vessel Design Summary: Step: 24 11:46pm Dec 22,2021

Description	Type	Ratio	Fail
T1	WRC-107/537	0.709	Passed
S2	WRC-107/537	0.839	Passed
S1	WRC-107/537	0.839	Passed
S3	WRC-107/537	0.709	Passed
T2	WRC-107/537	0.839	Passed

Weights:

Fabricated - Bare W/O Removable Internals	3902.4 kg.
Shop Test - Fabricated + Water (Full)	5602.4 kg.
Shipping - Fab. + Rem. Intls.+ Shipping App.	3902.4 kg.
Erected - Fab. + Rem. Intls.+ Insul. (etc)	3902.4 kg.
Empty - Fab. + Intls. + Details + Wghts.	4058.1 kg.
Operating - Empty + Operating Liquid (No CA)	5131.2 kg.
Field Test - Empty Weight + Water (Full)	5463.0 kg.

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